

Controlling Maneuverability of a Bio-Inspired Swimming Robot Through Morphological Transformation

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Abstract—Biology provides many examples of where body adaption can be used to achieve a change in functionality. The feather star, an underwater crinoid which uses feather arms to locomote and feed, is one such system; it releases its arms to distract prey and vary the maneuverability to help escape prey. Using this crinoid as inspiration, we develop a robotic system that can alter the interaction with the environment by changing its morphology. We propose a robot that can actuate layers of flexible feathers and detach them at will. We first optimize the geometric and control parameters for a flexible feather using a hydrodynamic simulation followed by physical experiments. Secondly, we provide a theoretical framework for understanding how body change affects the controllability. Thirdly we present a novel design of a soft swimming robot with the ability of changing its morphology. Using this optimized feather and theoretical framework, we demonstrate on a robotic setup how the detachment of feathers can be used to change the motion path whilst maintaining the same low level controller.

I. INTRODUCTION

The feather star is a marine crinoid, an invertebrate with multiple soft ‘feathery’ arms which enable swimming and maneuvering to avoid prey, and also feeding on drifting microorganisms [1]. These animals show many fascinating properties, including their deformable feather-like structure and cyclically actuated muscles. One property which is believed to be unique to echinoderms is their *mutable collagenous tissue* [2]. This enables them to drastically alter their body structure within a timescale of seconds under direct control of the nervous system. In the case of feather stars, they use this tissue to detach their feathered arms. It is believed this mechanism is used to distract prey and also change their dynamics to assist its evading predators [3]. This demonstrated ability to drastically alter the body morphology and passive properties to alter maneuverability is of keen interest to the robotics community. It provides inspiration for the development of robots that can utilize or change their body structure to aid their end goal, or indeed their survival [4]. Thus, the goal of this work is to develop a feather star inspired robot which uses an artificial equivalent of this ‘mutable tissue’ to change its body structure. Furthermore, we present a theoretical model to investigate how the detachment of the arms affects the control and maneuverability of the design.

Within the domain of underwater bio-inspired robots, there have been a number of notable examples where limbs (similar to the feathers on the feather star) and their controllers

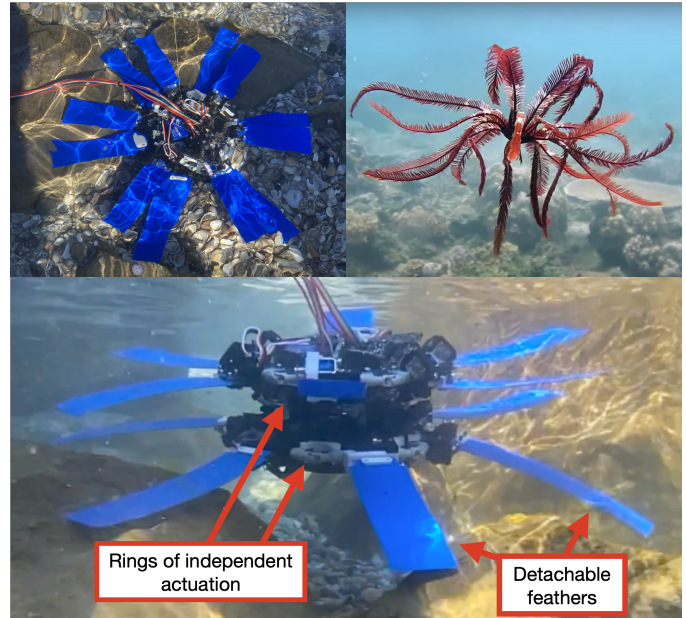


Fig. 1. The developed feather star robot with multiple actuated rings and detachable feathers and its biological inspiration [5].

have been optimized to maximize the generated thrust [6]. This includes an octopus inspired robot [7] and a star fish robot [8]. Whilst these examples consider the optimization of the design of the structure to maximize thrust or behavioral range, there are limited examples of underwater robots that show considerable changes in body structure to aid control. Developing and designing robots that can utilize change in the passive properties or morphology is a key quest for embodied intelligence researchers. The role of morphology driven control has been previously formalized [9] and has been shown to aid in achieving stability in legged underwater vehicles [10] and shaping the behavioral landscape of complex systems [11]. This previous work has highlighted the potential for morphology driven control which could be particularly beneficial in aquatic environments where fluid-structure interactions can be complex and challenging to control and exploit. To explore these capabilities, we must first create robots or structures that show significant variation in their physical structure or passive properties. To date, this has mostly been demonstrated through stiffness change of robotic systems [12], or through modular, reconfigurable robotic systems [13]. Using these new capabilities we must then optimize for the morphology for optimal thrust generation [14], [15], but must also address how we should design the global structure before and after body changes to achieve morphology driven control.

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By developing a feather star inspired robot with detachable feathers we introduce a new approach to achieving significant morphological transformation in a swimming robot, which we then use to explore how body adaption can be used to assist with maneuvering. The novel robotic system is formed from multiple layers of actuated rings of feathers which it uses as a means of thrust generation. All the feathers in a single layer are actuated collectively, and maneuverability is achieved through adaption of the body opposed to control of individual feathers. A mechanical system for rapid detachment has been integrated into the actuated feather rings to allow for detachment of individual feathers. Due to the complex interactions between the deformable feather and water we utilize simulation to perform a wide sweep of the control and design landscape to identify a small range of feather structures and controllers which are likely to maximize the thrust generation. By developing a custom measurement setup, we then validate a small subset of these results to find the optimal feather and controller. To understand how to design the initial configuration of the robot and the choice of feathers to detach we have developed an algorithm that utilizes the state-space representation to evaluate and determine how to change and restore the degree of controllability.

To demonstrate the contributions of this work we experimentally validate the optimized robot structure and controllers on the robot hardware. The maneuverability of the robot is shown to alter with different configuration of the robot's feather, following which the ability to detach the feathers on demand to alter the heading and path is shown. In the remainder of this paper, we first present the methods to systematically address this problem. The novel robotic hardware is then shown, followed by the experimental results. We finish with a discussion and conclusion.

II. PROBLEM STATEMENT

Using the feather star as biological inspiration, we aim to develop a robot that utilizes feather-like structures to swim. By providing the robot with the ability to alter or morph its body, we want to show how changes in maneuverability can be achieved through altering the body structure. In order to achieve this aim, we subdivide the problem into three key goals which all seek to explore how these bio-inspired components can be used to improve the capabilities of robots:

- Explore the role of feather structures and periodic controllers in the generation of thrust through embodied interactions with the water, and optimize their thrust generation through co-design of the morphology and the controller
- Develop a framework for selecting the robot structure before and after detachment of limbs to optimize the performance of the robot. In particular we present methods for the optimization of the thrust in a particular direction and for maximizing the degree of controllability of the robot.
- Develop a robotic hardware that mimics the behavior of the feather star, by considering a mechanism that allows

multiple feathers to be actuated simultaneously and to be detached at will

The following three sections present the methods developed to address these three aspects of the problem.

III. MODELLING & OPTIMIZATION OF BIO-INSPIRED FEATHERS

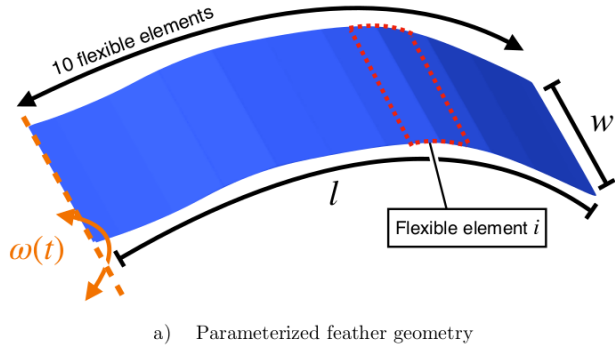
To achieve the best maneuverability, we wish to optimize the feather design parameters to maximize the thrust generated. The thrust is generated through complex interactions with the fluid, and is dependent on the geometry of the feather and the periodic motion at its root. To begin the optimization process, we first define the feather design parameters. Then we use a hydrodynamics simulation model to explore the large design space and we identify a subset of parameters which produces the largest thrust. Finally, the subset identified in simulation is further explored through real work experiments with a custom physical experimental setup.

A. Parametric Feather Design

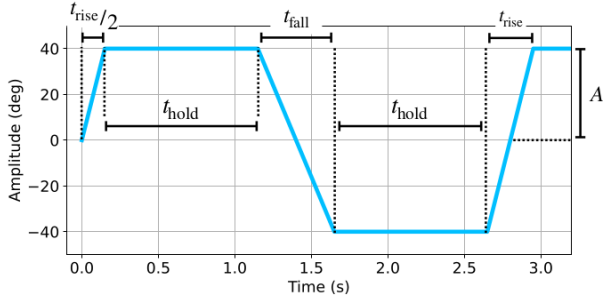
The parameterized feather design and controller is as shown in Fig. 2a and b. The geometry is defined as a rectangle with its width w and length l as parameters. The control motion corresponds to the angular displacement at the root the feather. We evaluate only periodic motions to mimic the movements of the feather star. Consequently, the signal is fully described by the rise time t_{rise} , fall time t_{fall} , and hold time t_{hold} . We keep instead the amplitude constant at $A = 40$ degrees to limit the size of the design search space. The value 40 degrees is the maximum amplitude of the mechanical setup. Hence, the parameters $p_{\text{des}} = [w, l, t_{\text{rise}}, t_{\text{fall}}, t_{\text{hold}}]$ defines the feather design. Polypropylene sheets of 0.4 mm thickness were chosen as the base material for the feathers due to their flexibility and ease of fabrication using a CO₂ laser cutter.

B. Hydrodynamics Modeling

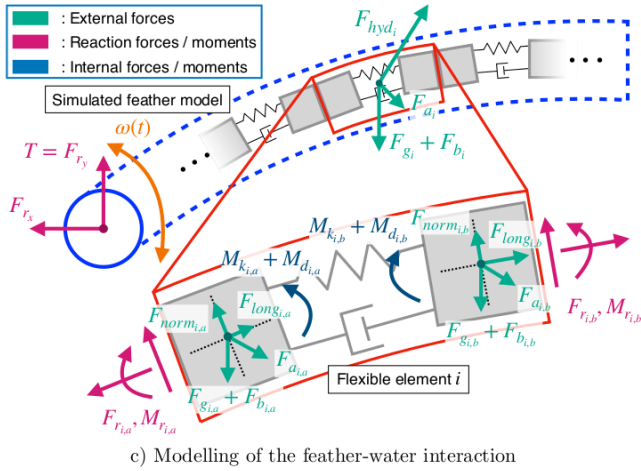
In the following section we present the model of the interaction of a single feather with the water. In particular, we first develop a discretized model for the feather, and then we define the forces exchanged between the structure and the fluid. We model a single feather as a collection of discrete flexible elements using Simscape MultibodyTM, where a single feather is approximated by ten flexible beam elements (Fig. 2a). Each flexible element, i , consists of two masses (a and b) of identical shape and mass, which are joined by internal springs and dampers allowing bending (Fig. 2c). The first mass element, a , of the first beam element ($i = 1$) is rigidly fixed to an angular actuator at the feather's base, while the second mass element, b , of the last beam unit ($i = 10$) is a free end. All flexible elements are connected by rigid rotational joints. The influence of the rest of the feather is imposed on each flexible element at the leading and trailing rigid joint locations as a combination of force, F_r and bending moment, M_r acting on each joint as defined in (Fig. 2c). The bending deformation of the feather is captured by imposing a structural bending stiffness, k_i and damping, d_i between the masses, a and b , which results in moments, M_{k_i} and M_{d_i} . The rotational stiffness, k_i was calculated as EI_i/l_i , where E



a) Parameterized feather geometry



b) Parameterized feather motion



c) Modelling of the feather-water interaction

Fig. 2. a) Simulation of actuated parameterized feather in Simscape Multibody™, b) Plot of the parameterized control signal, c) Underlying multi-body model of a single feather and its interaction with water, with a close-up free-body diagram of a flexible element.

is the elastic modulus, I is the area moment of inertia, and l is length of the section. The elastic modulus was tuned by comparing the simulation's visual output against a physical rectangular polypropylene feather in water. The damping was approximated to be zero for all simulated joints.

Each of the masses are subject to a set of lumped external forces. Although in the simulation, the forces are solved for each mass, for this derivation the formulations will be expressed for the generalized flexible element, i . For each flexible element, i , the total lumped external force, F_{ext_i} , consists of gravitational force, F_{g_i} , buoyancy force, F_{b_i} , hydrodynamic force, F_{hyd_i} , and added mass force, F_{a_i} :

$$F_{\text{ext}_i} = F_{g_i} + F_{b_i} + F_{\text{hyd}_i} + F_{a_i} \quad (1)$$

Since gravity and buoyancy always oppose each other, they can be combined, as $\rho_f V_i \left(1 - \frac{\rho_w}{\rho_f}\right) u_g$, where ρ_f and ρ_w are the density of the polypropylene feather and water respectively, V_i is the volume, g is the gravitational acceleration, and u_g is the unit vector in the direction of gravity.

The hydrodynamic force, F_{hyd} is the total force due to the viscous interaction between the fluid and the structure. In some literature, the hydrodynamic force is decomposed into its lift and drag components, where the drag is in the direction of the relative velocity between the fluid and the body. In our simulation, we decompose the hydrodynamic force into forces in the normal and longitudinal directions of each feather element, F_{norm} and F_{long} , where these individual elements can be approximated as a rectangular prism.

The hydrodynamic opposing forces are approximated using the following equations [16]:

$$F_{\text{norm}_i} = \frac{1}{2} C_{\text{norm}_i} A_i \rho_w \|U_i\|^2 u_i^\perp \quad (2)$$

$$F_{\text{long}_i} = \frac{1}{2} C_{\text{long}_i} A_i \rho_w \|U_i\|^2 u_i^\parallel$$

where the relative velocity, U_i is decomposed into the normal and longitudinal components of the feather's longitudinal axis, A_i is the characteristic area of the feather unit, and C_{norm_i} and C_{long_i} are the hydrodynamic coefficients. These coefficients were found to be affected by the flow incidence angle. The normal component of velocity was used in this model instead of the incidence angle because for our flow regime, the streamlines of flow in the neighborhood of each feather element are bent such that the position of the separation point does not change for small angles of relative flow. The hydrodynamic coefficients for each mass of the flexible element was obtained from steady-state CFD simulations carried out using ANSYS software for Reynolds numbers ranging from 10 to 10^4 , estimated to be fitting for such biological systems [17].

Finally, the added mass force, F_{a_i} must also be accounted for in the total force formulation. The hydrodynamic mass force is the added mass of the movement of fluid around the accelerated body due to the action of pressure [18] and can be defined as $F_{a_i} = \rho_w A_i a_{\text{norm}_i}$, where A_i is the cross-sectional area of each element and ρ_w is the water density. The hydrodynamic added mass force is implemented on the normal component of the acceleration, a_{norm_i} to the feather's longitudinal axis. The added mass force in the longitudinal direction was ignored for this approximation.

The handling of the hydrodynamic forces captures a non-linear feature of the fluid-structure interactions as a dynamic feedback resulting from the motion of the individual elements. The actuation at the base of the feather therefore drives the kinematics and deformation dynamics.

C. Simulation Results

Using the hydrodynamic simulation, the design space spanned by p_{des} is explored. Specifically we explored all

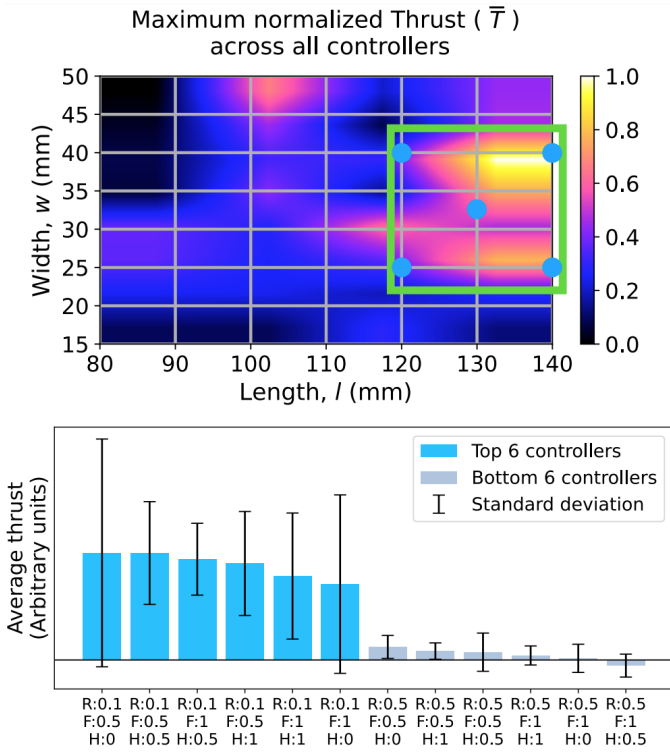


Fig. 3. **Top:** Heat map showing the variation of maximum \bar{T} across all controllers for every feather geometry combination. A subset of l and w is identified (see green box) to be further investigated using physical experiments. **Bottom:** Simulation result of mean and standard deviation of \bar{T} across all feather geometries. The worst and best controllers are identified. R, F, H corresponds to t_{rise} , t_{fall} , and t_{hold} .

combinations of geometries: $w = 15\text{-}50$ mm in 5 mm increments and $l = 80\text{-}140$ mm in 20 mm increments, and all combinations of the motion parameters: $t_{\text{rise}} = 0.1, 0.5$, $t_{\text{fall}} = 0.5, 1$, $t_{\text{hold}} = 0, 0.5, 1$ seconds. The geometries were chosen to match what could be used on the physical robotic setup. Each simulation was run for 10 cycles of the periodic motion, and the average thrust over one period at the base of the feather $\bar{T} = \int_0^{t_{\text{period}}} T(t) dt \approx \frac{1}{n} \sum_n T_t$.

In the remainder of the paper when referring to the ‘thrust’ we consider this to be the average thrust \bar{T} over a period.

Fig. 3 shows the simulation results of the thrust produced by various combinations of the parameters. These results allow us to select a smaller subspace to explore experimentally to find the optimum feather design parameter, represented as p_{des}^* . Although the simulation allows general trends to be captured and the search space to be reduced, there remains a significant *reality-gap* such that a small number of more costly real world experiments must be performed. We explored five feather geometries: $[w, l] = [25, 120], [25, 140], [32.5, 130], [40, 120], [40, 140]$ and seven control motions—the top six performing controllers and one of poorly performing controller.

D. Experimental Exploration

To validate the simulation, we created an experimental setup (Fig. 4) which replicates the simulation. It uses a servo-powered mechanism to actuate the base of the feather, with

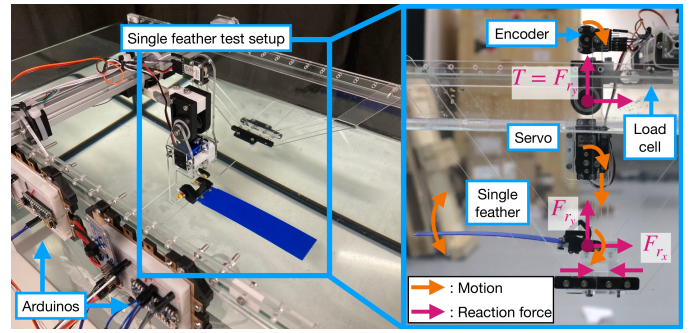


Fig. 4. Experimental setup for isolating and measuring the thrust generated by feathers of different values of p_{des} .

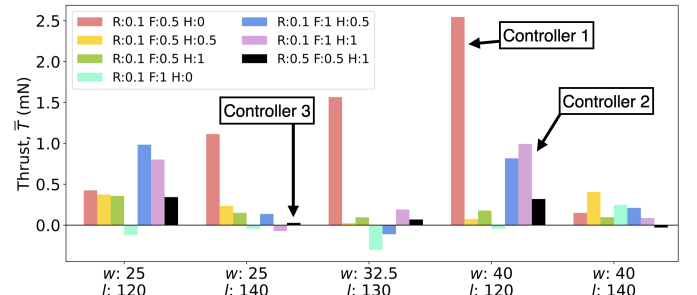


Fig. 5. Experimental thrust values for a subset of feather geometries and controllers identified from simulation.

a load cell used to measure the upwards thrust. The setup ensures there are no moments applied at the load cell such that it truly measures the upwards thrust.

Fig. 5 shows the thrust measurements for the subset of parameters selected from the simulation results. Here we observe the extreme sensitivity of the generated thrust for different design parameters. This highlights the complexity of optimizing the parameterized feather even after reducing the design space to five parameters, and a need to use simulation to systematically explore and reduce the design space while not missing local maximas. From these results, some conclusions can be drawn to identify the optimal parameters p_{des}^* . For the geometry, we identify the parameters $[w, l] = [40, 120]$ as the optimum as it has the highest thrust recorded, and the highest average thrust across different controllers. For the controller, for the optimized feather ($[w, l] = [40, 120]$) we identify the parameters $[t_{\text{rise}}, t_{\text{fall}}, t_{\text{hold}}] = [0.1, 0.5, 0]$ as the optimum. Hence, we have identified $p_{\text{des}}^* = [40, 120, 0.1, 0.5, 0]$. The black bar graph shows the performance of the poorly performing controller in simulation. We observe that this controller is consistently poorly performing for all feather sizes, which corresponds to the simulation results.

Going forwards we will refer to the optimal motion parameters ($[t_{\text{rise}}, t_{\text{fall}}, t_{\text{hold}}] = [0.1, 0.5, 0]$) as controller 1, to the second best motion parameters ($[t_{\text{rise}}, t_{\text{fall}}, t_{\text{hold}}] = [0.1, 1, 1]$) as controller 2, and to the poorly performing parameters as controller 3.

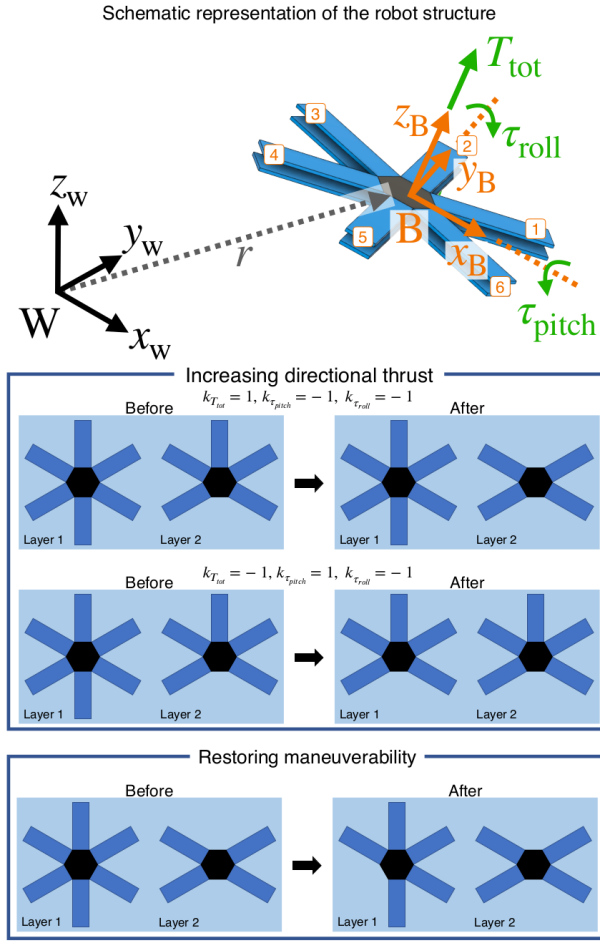


Fig. 6. **Top:** The schematic representation of the abstracted robot. **Bottom:** Diagram to illustrate how the control algorithm allows for altering the level of controllability.

IV. COMPUTATIONAL DESIGN OF THE ROBOT STRUCTURE

For the specific structure of robot we consider a two layered system, where each layer has 6 feathers which are attached to the body radially and symmetrically (Fig. 6). Therefore, in this specific design, we used $n = 6$ and $n_s = 2$. Each layer of feathers can be actuated collectively, with the angle of the base of each feather controlled in the same way as seen in the simulation and experimental validation. Due to the sensitivity of the thrust generation of a feather to slight changes in the controller (Fig. 5), maneuverability through control of individual feathers would be ineffective. Thus, control of the motion must be achieved with a constant controller and utilizing changes in the body. In this section we define how the desired change in behavior can be achieved by varying the morphology of the structure, i.e. by detaching feathers. To this end, by analyzing the controllability of the robot we can then develop algorithms for optimal detachment of the feathers, aimed at increasing the directional acceleration or at restoring the maximal maneuverability of the design.

As the feathers are actuated in synchrony, we can write the position $q \in \mathbb{R}^n$ of the n feathers as a function of the motor position $\sigma \in \mathbb{R}^{n_s}$. We can define the configuration of the robot

with a configuration matrix $C \in \mathbb{B}^{n \times n_s}$ with $c_{i,j} = \{0, 1\}$. A value of 1 represents the presence of a feather in the i -th position for the j -th motor, while a value of 0 represents a detached feather. Consequently it holds that:

$$q_{i,j} = \rho c_{i,j} \sigma_j, \quad (3)$$

where ρ is the transmission ratio between the feather and motor coordinates.

In order to identify designs which are able to maneuver with the minimal number of actuators, we first define the state-space model for a robot with a generic structure C . Then, by analyzing the role of C on the controllability of the robot, we propose the optimal configurations, together with optimal policies for the detachment of the feathers.

A. State-space Representation

In this section we want to define the state-space representation of a robot defined by the configuration matrix C , aiming at controlling a non-holonomic under-actuated system in a 3D space with $n_s < n$ motors.

The equations of motion can be written in the inertial frame, indexed with W or in the body frame, referenced with B , as in Fig. 6. The position of the B frame in W coordinates is defined by the translation vector r with components $[x, y, z]$ and by the Euler angles $[\phi, \theta, \psi]$ defining the roll, pitch, and yaw angles respectively. Using these coordinates we can write the transformation matrix between the reference systems W and B . Finally, the angular velocity of the robot is described as $\omega_{BW} = [p, s, v]$, denoting the angular velocity of frame B in the frame W , with components p , s , and v in the body frame.

Therefore the Newton-Euler equations for the whole robot are:

$$m \ddot{r}_W = \begin{pmatrix} 0 \\ 0 \\ -(m - m_w)g \end{pmatrix}_W + \begin{pmatrix} 0 \\ 0 \\ T_{\text{tot}} \end{pmatrix}_B \quad (4)$$

$$I \begin{pmatrix} \dot{p} \\ \dot{s} \\ \dot{v} \end{pmatrix}_B = \begin{pmatrix} \tau_{\text{pitch}} \\ \tau_{\text{roll}} \\ 0 \end{pmatrix}_B - \begin{pmatrix} p \\ s \\ v \end{pmatrix}_B \times I \begin{pmatrix} p \\ s \\ v \end{pmatrix}_B$$

where I represents the inertia of the body, and m, m_w are the mass of the robot and of the displaced water respectively.

Each feather is actuated following a periodic oscillatory trajectory producing an average thrust \bar{T} directed upward in the plane of motion, as described in Section III. As all feathers on the same layer are moved in synchrony, the thrust produced by a generic feather on the j -th layer can be described as \bar{T}_j . Consequently, the forces produced by the whole robot on the environment can be written as:

$$T_{\text{tot}} = \sum_i^n \sum_j^n c_{i,j} \bar{T}_j$$

$$\tau_{\text{pitch}} = d \sum_j^n \bar{T}_j (c_{2j} - c_{5j} + \frac{1}{2}(-c_{1j} - c_{3j} + c_{4j} + c_{6j})) \quad (5)$$

$$\tau_{\text{roll}} = d \frac{\sqrt{3}}{2} \sum_j^n \bar{T}_j (-c_{1j} + c_{3j} + c_{4j} - c_{6j})$$

where d represent the distance between the center of mass and the location of the resulting force for each feather. In the following, the vector of inputs $u_1 = [T_{\text{tot}}, \tau_{\text{pitch}}, \tau_{\text{roll}}]^\top$ will be used, in order to simplify the physical intuition. However, the vector u_1 can be mapped back to the motor activation input $u_2 = [\bar{T}_1, \dots, \bar{T}_{n_s}]^\top$ through the mapping $M = \frac{\partial u_1}{\partial u_2}$ such that it holds $u_1 = Mu_2$. Thanks to this mapping we can write the resulting forces on the robot, given the configuration C and the activation state of the motors.

We can then rewrite the system in state-space form. The state of the system is given by the position and velocity of the center of mass and the orientation and the angular velocity: $x = [x, y, z, \phi, \theta, \psi, u, v, w, p, s, v]^\top = [x_1, x_2, x_3, x_4, x_5, x_6, x_7, x_8, x_9, x_{10}, x_{11}, x_{12}]^\top$, while the control input is $u = [\bar{T}_1, \dots, \bar{T}_{n_s}]$.

Therefore, the complete nonlinear system becomes:

$$\dot{x} = \begin{pmatrix} x_7 \\ x_8 \\ x_9 \\ x_{10} \\ x_{11} \\ x_{12} \\ x_{12}x_8 - x_{11}x_{10} - g \sin(x_5) \\ x_{10}x_9 - x_{12}x_7 + g \sin(x_4) \cos(x_5) \\ x_{11}x_7 - x_{10}x_8 + g \cos(x_4) \cos(x_5) - \frac{T_{\text{tot}}}{m} \\ \frac{I_y - I_z}{I_x} x_{12}x_{11} + \frac{\tau_{\text{roll}}}{I_x} \\ \frac{I_z - I_x}{I_y} x_{10}x_{12} + \frac{\tau_{\text{pitch}}}{I_y} \\ \frac{I_x - I_y}{I_z} x_{10}x_{11} \end{pmatrix} \quad (6)$$

Using the nonlinear state-space representation, we define a linearization around the hovering configuration, and we study the controllability of the structure as a function of C . In the linearized state around the hovering condition, $\theta = 0$, $\phi = 0$, $\sum T = (m - m_w)g$ holds. Therefore the equations of motion can be written as:

$$\dot{x} = Ax + Bu_1 = Ax + BMu_2. \quad (7)$$

Given A and B , we can study the degrees of controllability of the configuration by evaluating the rank of the controllability Gramian $\text{rank}(\text{gram}(A, BM))$ of the system [19]. A robot with a degree of controllability of n is able to exert forces along n independent directions. This metric is then ultimately linked to the robot's behavior, as it informs us on how many independent directions the robot can accelerate or react to disturbances. Maximizing such metric leads to a robot that can react to any disturbance and that is fully controllable [20].

B. Optimal Detachment

In this section we present how the detachment of feathers can be optimized to increase the performance of the robot in one specific task, such as escaping a predator. In this scenario, we want our robot to move as quickly as possible in one direction. As such, the robot configuration can be optimized to

produce the maximum directional thrust by solving the discrete optimization problem:

$$\begin{aligned} \underset{c_{i,j} \in \{0,1\}}{\text{argmax}} \quad & k_{T_{\text{tot}}} |T_{\text{tot}}| + k_{\tau_{\text{roll}}} |\tau_{\text{roll}}| + k_{\tau_{\text{pitch}}} |\tau_{\text{pitch}}| \quad (8) \\ \text{s.t.} \quad & T_j \in \{-\bar{T}_j, 0, \bar{T}_j\}; \end{aligned}$$

where $[k_{T_{\text{tot}}}, k_{\tau_{\text{roll}}}, k_{\tau_{\text{pitch}}}] \in \mathbb{R}^3$ are weight parameters that describe the desired behavior and $T_{\text{tot}}, \tau_{\text{roll}}, \tau_{\text{pitch}}$ are defined as in Eq. 5. Note that the optimization is solved by evaluating a combinatorial problem on the subset of binary variables $c_{i,j}$ that have value 1 in the current configuration.

On the other hand, a feather can be detached to restore a higher degree of controllability. In this case, the optimal feather to detach is found by solving the optimization problem:

$$\begin{aligned} \underset{c_{i,j} \in \{0,1\}}{\text{argmax}} \quad & \text{rank}(\text{gram}(A, B)) \quad (9) \\ \text{s.t.} \quad & T_j = \bar{T}_j; \end{aligned}$$

Through the evaluation of the controllability Gramian matrix, we can distinguish between designs with the same $\text{rank}(M_C)$, where M_C represent the controllability matrix. In the bottom of Fig. 6 we show how, through the mathematical model of the design, we can optimize the configuration of the robot for a desired behavior. The results from the presented algorithm are used as a framework for the experiments presented in Section VI.

V. ROBOTIC HARDWARE DESIGN

The robot is a multi-layered system. Each layer has six feathers arranged regularly in a hexagonal pattern (shown in Fig. 7a). A servo motor is located in the center of the layer which actuates two feather holders on opposite sides through a linkage mechanism. On either side of the holder directly actuated by the servo motor, an additional holder is located which is connected by a universal joint. This mechanical configuration results in all holders being actuated in synchrony (see Fig. 7c).

A feather shape is fabricated by cutting a polypropylene sheet on a laser cutter. Each feather is bolted to an acrylic connector which can be slotted in to the feather holder. Each holder can be equipped with a solenoid which drives the detachment mechanism (Fig. 7b shows how feathers can be attached or detached to the layer).

The feather detachment mechanism is described in Fig. 7d. When a solenoid attached on the feather holder is energized, a pin holding the feather in place is unlocked, allowing a leaf spring (a 3D printed flexure) to push out the feather from the holder. This allows feathers to be detached rapidly and on demand.

In this experiment, two layers were fabricated and assembled (see in Fig. 1). Electrical wires form the servo motor and solenoids connect the waterproofed robot to the Arduino microcontroller located outside of the water. Solenoids are energized through MOSFETs which can be turned on and off via the Arduino. The Arduino is connected to a PC where the

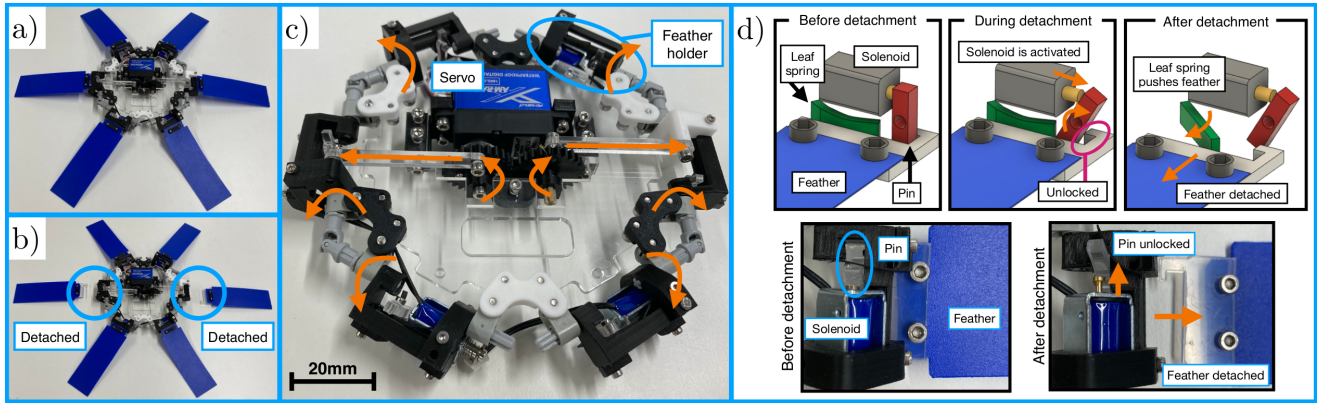


Fig. 7. Mechanical design of the robot focusing on the single layer. a) A single layer with all feathery appendages attached. b) Single layer with two feathery appendages detached. c) Actuation mechanism for moving six feather holders with one servo motor. d) Detachment mechanism using a solenoid.

feather detachment and the servo motion profile is commanded by a human operator.

VI. EXPERIMENTAL RESULTS

A. Feather Optimization

To demonstrate and experimentally validate the optimization of a single feather on the robot, a single ring of the robot was tested with different feathers and controllers. The single ring was tested with the ‘optimized’ feathers (120mm x 40mm) and also feathers with the same area but longer (192mm x 25mm) to provide a baseline. For these two robot configurations three different controllers were tested: the two top controllers from the experimental validation of the optimized feather (Fig. 5), i.e. controllers 1 and 2, and controller 3 which was poorly performing in both simulation and in the single feather experiment. The neutrally buoyant robot was placed in a downwards facing configuration with a thin rod placed through the center of its body to constrain the motion to allow for comparison between experiments. For each combination of feather geometry and controller, the robot was recorded swimming downwards until the bottom of the tank was reached (or the time exceeded 1 minute) for five repetitions, where the average swimming velocity was obtained. The results, in Fig. 5, show that the optimized feathers significantly outperform the baseline feathers, with the speeds generated by the top two controllers offering over 60% increase relative to the baseline feathers. The experiments reinforce that controller 1 shows the greatest thrust generation and hence velocity. However, the relative performance to controller 2 is poorer than expected with controller 2 only showing a velocity on average 10% lower than controller 1. Compared to the single feather case, the servo motor on the robot must exert a higher torque to move all the feathers. Without a sufficiently long hold time, the true position of the servo motor in some cases does not reach the desired position (for both up and down strokes). Hence, we believe that controller 2 with a longer hold time is less susceptible to environmental disturbances and noise which cause the up and down stroke to not meet the desired angles. In addition, it was clearly observed that when using controller 1, the true feather amplitude is less than the servo motor demand position ($A = 40$ degrees). This is because of the lack of hold

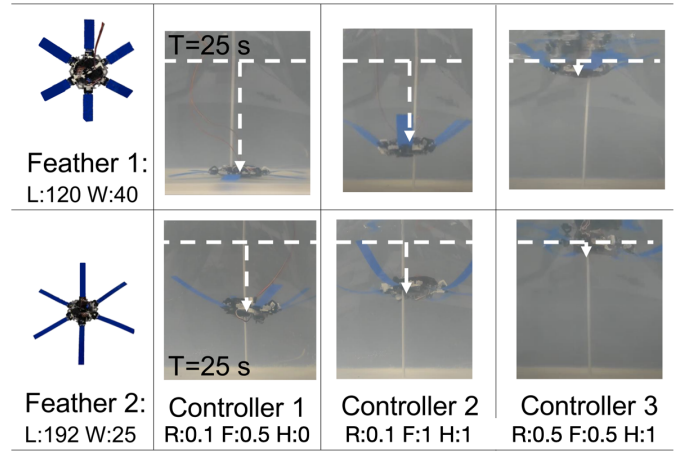
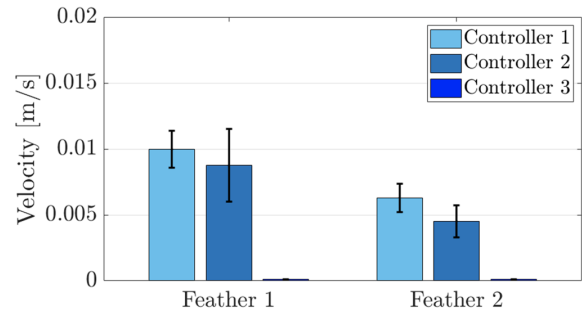


Fig. 8. Comparison between the optimal feathers (120 x 40), and baseline with same area (192 x 25) for three different controllers. Each experiment was repeated five times with the velocity and standard deviation presented.

time in controller 1, which means the servo demand position direction is reversed before the motor has time to reach it. In comparison to the top performing controllers, controller 3 struggled to produce any downwards thrust for both feather geometries.

B. Single Ring Control of Motion

To demonstrate and explore the degree of controllability we first present one single layer of the robot. Whilst the range of motion is lower than for multiple layers, it allows the concepts to be explored on a lower complexity hardware. To demonstrate how varying the body structure leads to different

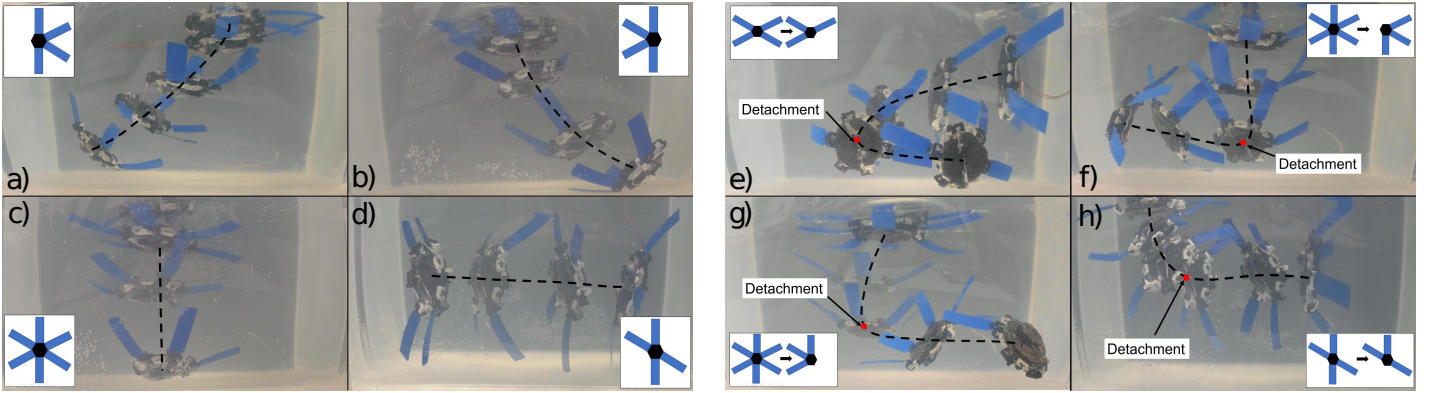


Fig. 9. **Left:** Maneuverability of the robot for different feather configurations. **Right:** Example motions of the robot showcasing the change of heading when feathers are detached. In particular, the detachment of the feathers is computed by solving Eq. 8, with the cost function parameters $[k_{T_{tot}}, k_{\tau_{roll}}, k_{\tau_{pitch}}] = [0, 0, 1], [0, 0.1, -1], [0, -0.1, 1], [0, 1, 0]$ for e, f, g, and h respectively.

behaviors we first show the motion of the robot with different fixed feather configurations (Fig. 9a-d). For these and all following experiments we aimed to perform them in the center of the tank as much as possible, and if the sides of the tank influenced the experiment, the experiment was repeated. These results show how the expected behavior, that different robot configurations lead to different motions and directions of swimming. Fully symmetric designs (Fig. 9c) and partially symmetric designs which produces a net thrust at the centre of the layer (Fig. 9d) leads to the robot swimming in a straight line. Whereas asymmetric designs (Fig. 9a,b) allows for gradual turning arcs.

To demonstrate how we can use this variation in motion with body shape we extend the experiments to include detachment of feathers to transition between different motions. As shown in Fig. 9e-h the detachment of feathers results in the change in heading of the robot. By altering both the starting configuration and the selection of feathers to be detached, the motion before and after detachment can be controlled. Fig. 9e-h shows how the robot transitions from one stable motion to another stable motion, where the change is caused by feather detachment. These experiments demonstrate that even with the same controller, the change in body (detachment) leads to a change in motion. In some cases the robot moves close to the edge of the tank where the fluid behavior may be more complex. Although the dominant effect on the maneuverability is still from the feather detachment, in open water the exact trajectory may differ.

C. Two Layer Robot Demonstration

The full two-layer robot allows for a complete demonstration of the analytical results described by the optimization problems in Eq. 8 and Eq. 9. These optimization problems reflects the need to select the optimal detachment of feathers to increase the thrust in one direction or to maximize the controllability of the structure respectively. In particular, we provide an example of starting in a horizontal configuration and moving in the vertical direction. For this case we consider solving the two different optimization problems and assess how the solutions lead to change in controllability or speed.

In Option 1, we apply the optimization problem described in Eq. 9. We are therefore maximizing the possibility to control the structure, with the resulting structure able to move linearly, as shown in the top image in Fig. 10, and to turn, as shown in the bottom left image. However, this possibility in control comes at the cost of lower acceleration in both motions. On the other hand, in Option 2, we apply Eq. 8 with the parameters of the cost function being $[k_{T_{tot}}, k_{\tau_{roll}}, k_{\tau_{pitch}}] = [1, -1, -1]$. We are therefore maximizing the total thrust produced by the structure while penalizing any turning motion. For this optimization problem, the solution is to detach the asymmetric feather so that, when actuated, both layers will produce thrust in the same direction. These results are reflected in the bar chart in Fig. 10. These results shows that with Option 2 a speed increase of over 60% can be achieved with the optimal feather detachment. Thanks to Eq. 8 and Eq. 9, we are therefore able to intuitively specify the desired behavior of the robot by tuning few parameters in the cost function of the optimization problem, and the optimal feather detachment will be computed, considering the internal dynamics of the system—described in Eq. 7.

VII. DISCUSSION & CONCLUSION

Biology provides many examples of mechanisms by which adaptation of the body can be used to simplify complex functions, or enable otherwise not possible functionality. The feather star provides a number of fascinating examples. Using this as an inspiration we have developed a robot with feathers that utilizes these passive structures for swimming and the detachment of these to change its structure. Utilizing modeling and experimental results we have optimized the morphology and control of these feathers and have developed a theoretical framework for when and how to alter the body to achieve a change in functionality.

The results presented show an initial exploration of using ‘detachment’ inspired by mutable tissues as a means of changing the body. This allows the robot to have a time-varying design space. Expanding this capability to allow not only detachment but also retrieval or growth would further expand this concept and bridge the gap to reality. In addition,

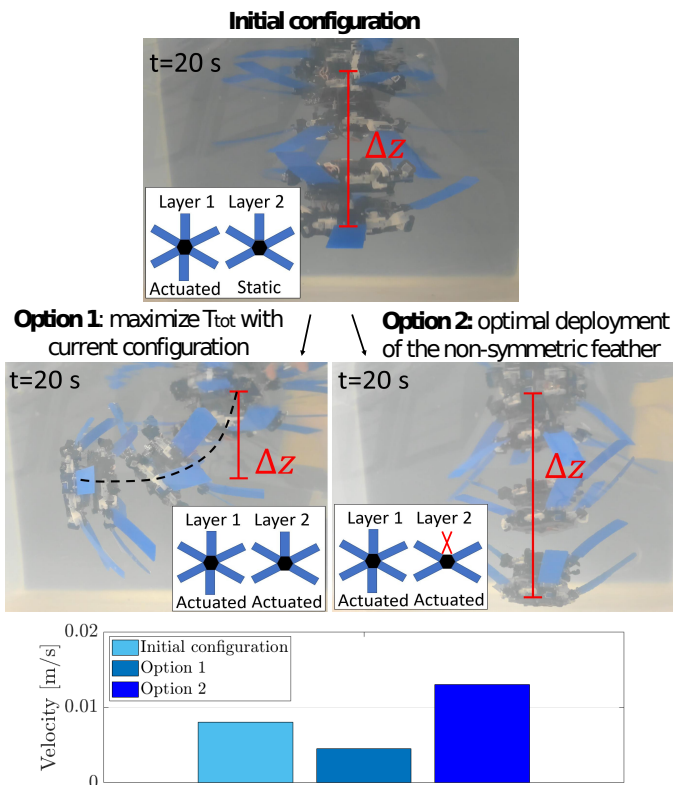


Fig. 10. Demonstration of the increased performance arising from the optimal detachment of the feather. Starting from the initial configuration represented on the top image, we can achieve a higher speed in the z direction by detaching one feather and actuate both layers.

increasing the physical complexity of the robot, for example increasing the number of layers, would allow a larger and more complex control space to be explored. Testing and evaluating the robot in a large body of water would also enable longer trajectories to be studied, and would provide a means of evaluating the maneuverability of the system with fewer edge effects from the tanks walls. Considering the feathers, even using the crude approximation of rectangular structures we have demonstrated that there is a complex relationship between their morphology, control and resultant thrust. Further exploring this study to consider other geometries, materials and also complex structures (for example bristles on the arms) would also extend this exploration.

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