

1     **Full-Scale Testing of Deep Wide-Flange Steel Columns under Multi-Axis Cyclic Loading:**  
2             **Loading Sequence, Boundary Effects and Out-of-Plane Brace Force Demands**

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4     **Abstract:** This paper discusses the findings from ten full-scale steel column tests subjected to  
5     multi-axis cyclic loading. The columns utilize deep wide-flange cross-sections typically seen in  
6     steel moment-resisting frames designed in seismic regions. The effects of boundary conditions,  
7     loading sequence, local web and member slenderness ratios on the column hysteretic behavior are  
8     investigated. The test data underscores the influence of boundary conditions on the damage  
9     progression of steel columns. Local buckling followed by out-of-plane deformations near the  
10    plastified column base are the dominant failure modes in fixed base columns with a realistic flexible  
11    top end. Twisting may occur only at drifts larger than 3% even when the member slenderness is  
12    fairly large. The test data suggest that bidirectional loading amplifies the out-of-plane deformations  
13    but does not significantly affect the overall column performance. The loading sequence strongly  
14    affects the column's plastic deformation capacity but only at story-drifts larger than 2%. After this  
15    drift amplitude, column axial shortening grows exponentially and becomes a controlling failure  
16    mode. Measurements of out-of-plane brace force demands at the column top exceed the lateral  
17    brace design force specified in North American standards.

18    **Keywords:** Deep Steel Columns; Boundary Conditions; Full-Scale Tests; Column Axial  
19    Shortening; Out-of-Plane Brace Force; Loading Sequence; Plastic Hinge Length

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## 20 **Introduction**

21 Due to their high moment of inertia-to-weight ratio, deep and slender wide-flange steel columns  
22 [i.e., depth,  $d \geq 400\text{mm}$  (16inches)] represent an economical solution for the seismic design of  
23 modern steel moment resisting frames (MRFs). The term slender refers to deep cross-sections that  
24 are seismically compact and their web and flange slenderness ratios are within the seismic  
25 compactness limits for highly ductile members ( $\lambda_{hd}$ ) as per AISC (2010a).

26 Past experimental studies on fully restrained beam-to-column moment connections that utilized  
27 deep columns (Chi and Uang 2002; Ricles et al. 2004) demonstrated that such members could be  
28 susceptible to twisting. This is exacerbated by the torsional demand and out-of-plane bending  
29 imposed on the column due to the inelastic buckling of the steel beam protected zones. Surveys  
30 from past full-scale experiments (FEMA 2000; Lignos and Krawinkler 2011, 2013) suggest that  
31 deep and slender wide-flange beams (i.e., absence of compressive axial load) deteriorate in flexural  
32 strength and stiffness at story-drifts in the order of 2.5% on average. This is due to the early onset  
33 of geometric instabilities (i.e., web and/or flange local buckling). Detailed finite element studies  
34 (Elkady and Lignos 2012, 2013, 2015a; Fogarty and El-Tawil 2015) associated with the cyclic  
35 behavior of steel columns of similar size cross-sections indicate that this issue becomes more  
36 critical in the presence of compressive axial loads. Notably, NIST (2010b) developed a research  
37 plan that aimed for a comprehensive understanding of the seismic behavior of deep and slender  
38 wide-flange columns and the development of guidelines for the seismic design of such members.

39 Early experimental studies on steel wide-flange columns mostly utilized relatively small cross-  
40 sections with depths ranging from W4 to W10 (Popov et al. 1975; MacRae et al. 1990; Nakashima  
41 et al. 1990). These specimens were tested either as cantilevers or with fixed-end boundaries (noted  
42 as fixed-fixed from this point on). Therefore, the location of the inflection point was constant

43 throughout the loading sequence. The focus of these tests was primarily on the effects of local  
44 slenderness on the hysteretic behavior of steel columns. These testing programs revealed that: (a)  
45 column axial shortening is a critical failure mode that influences the steel column stability  
46 (MacRae et al. 1990; MacRae et al. 2009); and (b) cyclic deterioration in the column's flexural  
47 strength becomes severe when subjected to compressive axial load levels larger than 50% of the  
48 column's axial yield strength,  $P_y$  (Popov et al. 1975). More recently, Newell and Uang (2008)  
49 tested at full-scale steel columns that utilized stocky W14 cross-sections in a fixed-fixed  
50 configuration. These members were able to sustain story-drift-ratios of 7% prior to 10% reduction  
51 in their flexural strength, even at high axial load demands. Notably, Ozkula et al. (2017) conducted  
52 full-scale tests on steel columns with deep and slender wide-flange cross-sections and fixed-fixed  
53 boundary conditions. These tests revealed that the observed failure modes might vary between  
54 local and lateral torsional buckling depending on the local and member slenderness ratios.

55 The above experimental studies share the following limiting features: (a) they were primarily  
56 conducted with simplified boundary conditions (i.e., cantilever or fixed-fixed); in this case, the  
57 torsional rigidity at the member ends was simultaneously lost after the formation of flexural  
58 yielding and the onset of local buckling. This strongly influences global failure modes associated  
59 with plastic lateral torsional buckling (Galambos and Surovek 2008); (b) the effects of  
60 bidirectional loading due to 3-dimensional ground motion shaking were not evaluated; (c) the  
61 influence of the loading history on the column hysteretic behavior was not assessed; and (d) the  
62 out-of-plane force demands at the columns' top boundary was never quantified such that the lateral  
63 bracing requirements in such members can be evaluated.

64 To address these issues, this paper presents a comprehensive full-scale testing program that  
65 investigated the hysteretic behavior of ten deep and slender wide-flange steel columns subjected

66 to multi-axis cyclic loading. The focus is on first-story interior columns in multi-story steel MRFs  
67 designed in highly seismic regions. More specifically, the scope and objectives of this paper are  
68 as follows:

- 69 1. To assess the effects of the cross-section slenderness and its interaction with the member  
70 slenderness on the steel column stability. Emphasis is placed on the plastic hinge length  
71 formation and the local and member instabilities observed during the damage progression.
- 72 2. To examine the effects of column end boundary conditions as well as the employed lateral  
73 loading histories on the cyclic behavior of steel columns.
- 74 3. To quantify the influence of bidirectional loading histories on the steel column stability in  
75 comparison with unidirectional loading histories.
- 76 4. To quantify the out-of-plane forces developed at the top end of steel columns and to assess  
77 the current North American design requirements for lateral bracing of steel columns.

78 Specific performance indicators of interest are the steel column axial shortening, the member out-  
79 of-plane deformations, flexural strength and stiffness deterioration at story-drift-ratios of interest  
80 to the engineering profession.

### 81 **Description of the Test Setup**

82 The test program was conducted at the structures laboratory of École Polytechnique de Montréal  
83 with a 6-degree-of-freedom (6-DOF) test setup shown in Fig. 1. This setup comprises of a steel  
84 base plate anchored to the laboratory's strong floor and a steel top platen connected to four vertical  
85 actuators. A pair of horizontal actuators per loading direction is connected to the top platen.  
86 Referring to Fig. 1, these actuators provide full control of the 6-DOFs ( $\delta_x, \delta_y, \delta_z, \theta_x, \theta_y, \theta_z$ ) at the  
87 top platen with mixed displacement/force control. The test setup allows for the realistic  
88 representation of the boundary conditions seen in first-story steel MRF columns due to the

89 flexibility of the beam-to-column connections intersecting a column. In this case, the inflection  
90 point location is not fixed but moves while the plastification progresses in the column. To the best  
91 of the author's knowledge, this has never been investigated in prior studies. In order to facilitate  
92 the discussion in the subsequent sections, the reference coordinate system X-Y-Z, shown in Fig.  
93 1, is employed.

## 94 **Description of the Test Matrix**

### 95 **Employed cross-sections**

96 Table 1 provides an overview of the test matrix in terms of the selected cross-sections, the  
97 applied compressive axial load ratios, the employed lateral loading histories, and boundary  
98 conditions. The test matrix comprises of ten column specimens in total (labeled C1 to C10). This  
99 includes six and four nominally identical column specimens that utilize a W24x146 and a W24x84  
100 cross-section, respectively. The former is commonly found in first-story columns in modern low-  
101 and mid-rise steel MRFs designed in North America (NIST 2010a; Elkady and Lignos 2014,  
102 2015b). The latter is representative of columns in low-rise steel special moment frames (SMFs),  
103 ordinary steel MRFs, and/or multi-tiered braced frames (Stoakes and Fahnestock 2016). Table 1  
104 summarizes the measured geometric properties of the two selected cross-sections. Both cross-  
105 sections have the same flange slenderness ratio ( $b_f/2t_f=6.0$ ; in which,  $b_f$  is the flange width and  $t_f$   
106 is the flange thickness of the cross-section) but different web slenderness ratios [ $h/t_w=33$  and  $47.3$   
107 for the W24x146 and W24x84 cross-sections, respectively;  $t_w$  is the web thickness and  $h$  is the  
108 clear web height as defined in AISC (2010a)]. In this way, we can assess the influence of  $h/t_w$  on  
109 the column hysteretic behavior. Detailed finite element studies (Elkady and Lignos 2013) prior to  
110 the testing program indicated that the web slenderness controls the column response over the  
111 flange slenderness. The web and flange slenderness ratios of the selected cross-sections comply

112 with the compactness limits for highly ductile members ( $\lambda_{hd}$ ) as per AISC (2010a). The W24x84  
113 column has a member slenderness ratio  $L_b/r_y=79$  ( $L_b$  is the laterally unsupported length;  $r_y$  is the  
114 radius of gyration about the cross section's weak-axis). The W24x146 column has a  $L_b/r_y=51$ .  
115 This ratio influences the column hysteretic response when member geometric instabilities are  
116 triggered. The member slenderness ratios of the selected column cross-sections allow for the  
117 assessment of the  $L_b/r_y \sim 60$  limit specified by the Canadian seismic provisions (CSA 2009).  
118 Referring to Table 1, the plastic moment ( $M_p$ )-to-elastic critical moment ( $M_{cr}$ ) ratio (i.e., torsional  
119 slenderness ratio,  $\lambda_{LTB}$ ). This parameter indicates how susceptible a column may be to lateral  
120 torsional buckling.

121 The column specimens have a clear length,  $L=3900\text{mm}$  ( $\approx 13$  feet). Each cross-section is welded  
122 into two, 75mm thick steel plates with complete joint penetration (CJP) J-groove welds. Weld  
123 access holes are prepared according to Section J1.6 of the AISC steel specifications (AISC 2010b).  
124 The column specimens are fabricated from A992 Grade-50 steel (i.e., nominal yield stress,  
125  $F_{yn}=345\text{MPa}$ ) as per ASTM (2015). Rectangular tensile coupon specimens are cut from the cross-  
126 section web and flanges to obtain their material properties in accordance with ASTM (2014). Table  
127 1 summarizes the web and flange average material properties based on three tensile coupon tests  
128 per location. Specimens C1 to C4, C5 and C6, and C7 to C10 are fabricated by three different steel  
129 batches. The three steel materials have a similar carbon equivalent value of 0.35%, which complies  
130 with the maximum permissible level of 0.45% specified by ASTM (2015).

### 131 **Employed loading protocols**

132 The focus of this paper is on interior columns because their hysteretic behavior is deemed more  
133 critical than that of end columns (Suzuki and Lignos 2014, 2015). In particular, end columns  
134 experience large axial load demand fluctuations due to the dynamic overturning effects; hence, the

135 neutral axis within the column cross-section considerably shifts while the axial load varies from  
136 compressive to tensile axial load demands coupled with lateral drift deformations. Therefore, the  
137 column axial shortening does not accumulate compared to interior columns in which the  
138 compressive axial load remains more or less constant (Suzuki and Lignos 2014, 2015).

139 To this end, nine out of ten specimens are subjected to a constant compressive axial load,  
140  $P=20\% P_y$ . Although this depends on the building plan view and lateral load resisting frame  
141 configuration, a  $P/P_y = 20\%$  is very representative in modern steel-frame buildings with SMFs  
142 (NIST 2010a; Elkady and Lignos 2014, 2015b). The same axial load ratio complies with the upper  
143 limit of  $30\% P_y$  according to the Canadian seismic provisions (CSA 2009) for “Type-D Ductile”  
144 steel MRFs. The AISC (2010a) seismic provisions do not consider such limit for steel SMFs. In  
145 order to examine the influence of high compressive axial load demands on the steel column  
146 hysteretic behavior, one specimen (i.e., Specimen C2) is subjected to  $50\% P_y$  (i.e.,  $P/P_{cr} > 50\%$ ;  
147  $P_{cr}$  is the column’s critical buckling load). This is representative of interior steel columns in 1970s  
148 tall steel MRF buildings (Bech et al. 2015). Therefore, Specimen C2 offers the opportunity to  
149 examine if steel columns that exhibit high compressive axial load coupled with lateral drift  
150 demands should be treated as force-controlled elements as per ASCE 41-13 (ASCE 2014).

151 Referring to Table 1, two types of unidirectional (UD) lateral loading protocols are employed.  
152 The first is the standard symmetric cyclic (noted here as “SYM”) lateral loading protocol (Clark  
153 et al. 1997). This protocol is shown in Fig. 2a and has been routinely used in past experimental  
154 studies (e.g., FEMA 2000). The second one is a collapse-consistent lateral loading protocol (noted  
155 here as “CPS”) developed by Suzuki and Lignos (2014, 2015). Referring to Fig. 2b, this protocol  
156 involves few inelastic lateral-loading cycles followed by large monotonic pushes in one direction  
157 (“ratcheting”; Ibarra and Krawinkler 2005). This is representative of what a first-story column

158 would experience when a building is subjected to a low probability of occurrence seismic event  
159 (Lignos et al. 2011; Lignos et al. 2013).

160 Bidirectional (BD) cyclic symmetric and collapse-consistent loading protocols (noted as SYM-  
161 BD and CPS-BD, respectively) are also employed. These involve elliptical drift cycles in the XZ  
162 plan view as shown in Figs. 2c and 2d for the SYM-BD and CPS-BD protocols, respectively. These  
163 protocols were developed based on concepts discussed in Krawinkler (1996, 2009). In brief, the  
164 SYM-BD lateral protocol reaches a maximum drift-ratio of 2% in the column's weak-axis bending  
165 direction during the 3% drift amplitude cycle in the column's strong-axis bending direction.  
166 Similarly, the CPS-BD lateral loading protocol reaches a maximum drift-ratio of 3% in the X-  
167 loading direction during the first excursion of the 5% drift amplitude in the Y-loading direction.  
168 Due to brevity, further details regarding the development of these protocols can be found in Elkady  
169 (2016).

#### 170 **Employed boundary conditions**

171 Specimens C1 and C2 are tested with fixed-end boundaries in the strong-axis bending direction.  
172 The rest of the specimens are tested with a fixed base and a flexible top end boundary (noted as  
173 fixed-flexible). To simulate the flexible boundary conditions, a pre-defined rotation  $R_x$  is applied  
174 about the X-axis at the specimen top end. This rotation history is synchronized with the lateral drift  
175 in the Y-axis direction. The pre-defined rotation is such that the inflection point within the column  
176 is set at  $0.75 L$  from the column base prior to column plastification. The inflection point location  
177 is chosen based on surveys from numerous studies on the seismic behavior of typical steel MRFs  
178 ranging from 1 to 20 stories and 1 to 5 bays, conducted by the authors as well as others (Gupta and  
179 Krawinkler 1999; Lignos et al. 2010; NIST 2010a; Suzuki and Lignos 2014; Elkady and Lignos  
180 2015b). All specimens are assumed to be fixed in their weak-axis bending direction including the



181 torsional degrees of freedom. This assumption may not be necessarily true for the column top end  
182 once local buckling initiates at the adjoining steel beams. Depending on the beam-to-column  
183 connection type, an appreciable amount of torsional force may be applied to the steel column (Chi  
184 and Uang 2002; Zhang and Ricles 2006). This issue deserves more attention in future research  
185 studies.

### 186 **Qualitative Summary of Typical Steel Column Damage Progression**

187 The typical damage progression sequence leading to loss of the flexural and/or axial load  
188 carrying capacity of a steel column is shown in Figs. 3(a-i) and 4(a-i). Referring to Fig. 3, the end-  
189 moment is normalized with respect to the measured full plastic flexural strength  $M_p$ , without any  
190 reduction due to the applied compressive axial load. The deduced end moment, at any load  
191 increment, is computed as the summation of the actuator force components transformed to the  
192 global coordinate system (see Fig. 1) multiplied by the corresponding distance from the actuator  
193 swivel to the column base/top. In Fig. 3, the true chord-rotation is calculated over the test  
194 specimen's length, after subtracting the measured column axial shortening. This represents the  
195 story-drift-ratio demands that a column experiences under reversed cyclic loading. Results for  
196 Specimen C4 are disregarded due to a control error in the loading rate application of the rotational  
197 DOF ( $R_x$ ).

198 Figure 5 shows the applied lateral loading protocol for Specimen C7 in the strong-axis bending  
199 direction including key damage states. The initial elastic cycles did not induce any notable  
200 deformation in the specimen. Flexural yielding occurred in the web and flanges prior to the 1.5%  
201 drift amplitude. From Fig. 6a, the inflection point was located near  $0.75 L$  from the column base  
202 as intended. Referring to Fig. 3f, prior to the onset of local buckling, Specimen C7 reached a  
203 maximum flexural strength,  $M_{max}$ , which was 10% higher than its expected unreduced plastic

204 flexural capacity (i.e.,  $M_{max}/M_p = 1.1$ ). Referring to Fig. 3, the same amount of cyclic hardening  
205 was observed in all the test specimens subjected to a  $P/P_y = 20\%$ .

206 Referring to Fig. 7a, flange and web local buckling near the column base became evident at the  
207 first cycle of the 2% drift amplitude and progressed during larger amplitude loading cycles. The  
208 center of the flange local buckling wave was located at  $0.6d$  from the bottom end plate. From Fig.  
209 7b, the local buckling formation was fairly symmetric due to the employed symmetric loading  
210 history. This was not the case for specimens subjected to a collapse-consistent loading history in  
211 which local buckling was only evident on the compressive flange due to ratcheting. Local buckling  
212 triggered flexural and axial strength deterioration near the column base (see Figs. 3f and 4f). This  
213 caused the inflection point to move towards the column base as shown in Fig. 6a. This was due to  
214 the force redistribution within the column once flexural strength deterioration initiated at the  
215 column base. Referring to Fig. 6b, this was also observed in Specimen C8 that was subjected to a  
216 collapse-consistent loading protocol. Notably, the force redistribution was not evident in fixed-end  
217 test specimens; thus, the inflection point remained at the column mid-height due to the  
218 simultaneous plastification of its ends (see Fig. 6c for Specimen C1).

219 Web local buckling caused column axial shortening, which in turn triggered considerable out-  
220 of-plane global deformations in specimens with fixed-flexible boundary conditions as shown in  
221 Fig. 8. The same figure shows the magnitude and progression of these out-of-plane deformations  
222 as monitored by a wireless displacement tracking system. Such deformations caused appreciable  
223 weak-axis moment demands due to member P-delta forces (i.e.,  $M_{y,P-Delta} = P \delta_x$ ). For instance, for  
224 Specimen C7 (see Fig. 8a), at the 4% drift amplitude with respect to the strong-axis bending, the  
225 weak-axis moment demands were equal to about 60% of the column's weak-axis plastic flexural  
226 strength (i.e.,  $M_{y,P-Delta} = 0.6 M_{p,y}$ ). This observation holds true for all the specimens tested with

227 fixed-flexible boundary conditions, regardless of the employed lateral loading protocol. Notably,  
228 this is not traced when columns are tested with fixed-fixed boundary conditions (see Fig. 8c).

229 Finally, the out-of-plane deformations were followed by column twisting near the specimen's  
230 base. The cross-section twisting angle ( $\theta_z$ ) was quantified using six string potentiometers attached  
231 to both flanges at three cross-sectional levels ( $\frac{1}{4} L$ ,  $\frac{1}{2} L$ , and  $\frac{3}{4} L$ ) as well as the wireless  
232 displacement tracking system. Figure 9 shows the cross-section twist angle versus chord-rotation  
233 for Specimen C7. Although in this case,  $L_b/r_y = 79$ , the column twisting became evident near the  
234 column base (i.e., at  $\frac{1}{4} L$ ) only after the 3% drift amplitude. By the end of the test, the maximum  
235 twisting angle was about 3.5 degrees near the column base plastic hinge region but less than 1  
236 degree near the column top end. This was because the torsional restraint at the column top was not  
237 lost simultaneously with that of the column base after the onset of local buckling. This indicates  
238 that characterizing the hysteretic behavior of steel columns with simplified fixed-fixed boundary  
239 conditions may be misleading. This is further elaborated in the subsequent sections.

240 Referring to Fig. 3f, due to the observed out-of-plane deformations and the associated twisting,  
241 Specimen C7 lost more than 70% of its initial lateral stiffness ( $K_e$ ) near its base. Furthermore,  
242 referring to Fig. 4f, the same specimen shortened by 145mm (i.e., 3.7%  $L$ ) at the end of the test.  
243 At this point, its flexural capacity was reduced by more than 70%  $M_{max}$  (see Fig. 3f).

## 244 **Synthesis of Experimental Results and Discussion**

245 This section provides a synthesis of the experimental data to assess several aspects related to  
246 the steel column stability due to reversed cyclic loading.

### 247 **Effect of cross-section and member slenderness**

248 Referring to Figs. 3 and 4, steel column flexural and axial strength deterioration, unloading  
249 stiffness deterioration, as well as column axial shortening were primarily induced by the interactive

250 effects of local buckling and out-of-plane deformations. To assess the influence of the cross-  
251 section web and member slenderness on the hysteretic behavior of steel columns, three pairs of  
252 specimens are compared: Specimens C3 and C7; Specimens C5 and C8; and Specimens C6 and  
253 C9. Each pair consists of two different cross-sections but it was subjected to the same loading  
254 history and boundary conditions. Referring to Fig. 3, although all the test specimens developed  
255 their plastic flexural strength, the ones with the less compact cross-sections (i.e., W24x84)  
256 experienced rapid strength deterioration. For instance, Specimen C3 lost 80% of its flexural  
257 capacity at 5% rads (see Fig. 3c) while Specimen C7 lost the same amount at 4% rads (see Fig.  
258 3f). Referring to Figs. 4e and 4h, at a reference drift of 4%, the W24x84 columns experienced  
259 about 20% more axial shortening, due to severe web local buckling, compared to the W24x146  
260 columns, regardless of the lateral loading protocol or the loading direction. These results suggest  
261 that the current AISC (2010a) compactness limits for highly ductile members warrant further  
262 review such that the column flexural strength deterioration and axial shortening meets certain  
263 acceptance criteria at a reference lateral drift amplitude.

264 Referring to Figs. 3e and 3h, at a reference story-drift-ratio of 4%, the unloading stiffness of  
265 Specimens C6 and C9 was reduced by more than 40% and 70%, respectively, with respect to their  
266 initial elastic stiffness. Referring to Table 1, the W24x84 columns have a relatively large member  
267 slenderness ( $L_b/r_y=79$ ) as well as torsional slenderness ratio ( $\lambda_{LTB}=0.42$ ) compared to the  
268 W24x146 specimens. This makes them more susceptible to out-of-plane and torsional  
269 deformations. For instance, at a story-drift-ratio of 4%, Specimen C9 experienced about double  
270 the out-of-plane deformations near its column base compared to Specimen C6. The above  
271 observation holds true for the other two pairs of specimens. These results suggest that an upper  
272 limit on the member and torsional slenderness ratios should be employed in future versions of the

273 AISC (2010a) provisions for the collapse prevention of steel SMFs subjected to low-probability  
274 of occurrence earthquakes.

### 275 **Effect of column end boundary conditions**

276 Referring to Figs. 3a and 3c, although the differences in the deduced moment-rotation relations  
277 of nominally identical specimens with fixed-fixed (Specimen C1, see Fig. 3a) and fixed-flexible  
278 (Specimen C3, see Fig. 3c) boundary conditions are practically negligible, the column axial  
279 shortening of the former (see Fig. 4a) is nearly double than that of the latter (see Fig. 4c). This is  
280 attributed to the simultaneous formation of local buckling at both ends of a fixed-end column. In  
281 this case, the member loses its torsional ( $J$ ) and warping resistance ( $C_w$ ) simultaneously at both  
282 ends. This is not representative of typical first-story steel columns in capacity-designed steel  
283 MRFs. Figure 10a shows a comparison of the deduced moment-plastic rotation relation at the  
284 column top for Specimens C1 and C3. To facilitate the comparison, both moment-rotation relations  
285 are plotted up to the second cycle of the 4% drift amplitude (see Fig. 3). Due to the flexible top  
286 end, Specimen C3 experienced a maximum plastic rotation of 1% rads because flexural yielding  
287 occurred at its top end only after the 3% drift amplitude of the employed lateral loading protocol.  
288 The inelastic deformation at the top of Specimen C3 are attributed to the increased flexural  
289 demands at the same location once local buckling forms and progresses near the column base.

290 The proper representation of the member end boundary conditions has potential implications  
291 on the expected steel column unloading stiffness deterioration under reversed cyclic loading. In  
292 particular, Fig. 10b shows a comparison of the unloading stiffness-to-the initial elastic stiffness  
293 ratio for Specimens C1 and C3 with respect to the peak drift amplitudes of a symmetric cyclic  
294 loading protocol. Up to 3% drift, the unloading stiffness deterioration of Specimen C1 is more than  
295 double compared to that of Specimen C3. This is primarily due to the simultaneous loss of the

296 torsional restraint in fixed-end columns, such as Specimen C1. In particular, Specimen C1  
297 experienced twisting angles ( $\theta_z$ ) almost two times larger than those observed in Specimen C3.  
298 From Fig. 10b, at 4% drift, representative of low-probability of occurrence earthquakes, Specimen  
299 C1 (fixed-fixed) under predicts the rate of unloading stiffness deterioration compared to Specimen  
300 C3 (fixed-flexible). This is attributed to the fact that the weak-axis bending demands, triggered by  
301 the large out-of-plane deformations due to in-plane bending, are not adequately captured in fixed-  
302 end columns (see Fig. 8c). In fact, Specimen C1 experienced 40% less out-of-plane deformations  
303 compared to Specimen C3.

304 At drift-ratios less than 3%, fixed-flexible specimens, including those with  $L_b/r_y=79$ , did not  
305 experience significant twisting. These findings contradict recent observations from Ozkula et al.  
306 (2017) where specimens with even lower member slenderness  $L_b/r_y \approx 70$  experienced lateral  
307 torsional buckling at similar lateral drift demands. This indicates that (a) the expected failure  
308 modes in steel columns may be fairly misleading if fixed-fixed boundary conditions are  
309 considered; (b) the current CSA S16-09 seismic provisions may be fairly conservative by limiting  
310  $L_b/r_y \sim 60$  for Type-D steel MRFs; hence, this limit could be revisited in future editions.

### 311 **Effect of compressive axial load**

312 The effect of the applied compressive axial load on the column stability is evaluated by  
313 comparing the hysteretic behavior of Specimens C1 ( $P/P_y=0.2$ ) and C2 ( $P/P_y=0.5$ ). Both  
314 specimens were subjected to a symmetric lateral loading protocol. Referring to Figs. 3a and 3b, it  
315 is evident that when the applied compressive axial load increases, the rate of cyclic and in-cycle  
316 flexural strength deterioration of the column increases considerably; therefore, its plastic  
317 deformation capacity decreases. This agrees with prior studies (MacRae et al. 1990; Ozkula et al.  
318 2017). Notably, Fig. 3b indicates that Specimen C2 was still able to develop an appreciable plastic

319 rotational capacity even though the  $P/P_{cr} > 50\%$ . Therefore, this member should not be treated as  
320 a force-controlled element as per ASCE-41-13 (ASCE 2014). This has direct implications for the  
321 seismic retrofit of existing tall buildings in which steel columns with stocky cross-sections, are  
322 treated as force-controlled elements if  $P/P_{cr} > 50\%$  (Bech et al. 2015).

323 Referring to Fig. 4a, Specimen C1 shortened minimally (i.e.,  $\delta_z=0.6\% L$ ) at 2% drift compared  
324 to Specimen C2 that shortened by 4%  $L$  (see Fig. 4b) at the same drift amplitude. This was due to  
325 severe web and flange local buckling in the presence of high compressive axial load demands.  
326 This suggests that in modern capacity-designed steel-frame buildings with MRFs, an upper limit  
327 on the axial compressive load demands should be set. For instance, the seismic provisions in New  
328 Zealand (SNZ 2007) limit the compressive axial load demands to 50%  $P_y$  for Category 1 (i.e.,  
329 highly ductile) column members. The Canadian seismic provisions (CSA 2009) prohibit the use  
330 of  $P/P_y > 30\%$  in steel columns as part of Type-D ductile steel MRFs. The test results and a  
331 corroborating parametric finite element analysis study (Elkady and Lignos 2015a; Elkady 2016)  
332 suggest that the latter limit seems to be more rational. Notably, the AISC (2010a) seismic  
333 provisions and the steel specification (AISC 2010b) do not impose such a limit.

### 334 **Effect of lateral loading sequence**

335 Representative first-cycle envelope curves are shown in Fig. 11 for three pairs of nominally  
336 identical specimens subjected to the two different lateral loading histories: Specimens C3 and C5;  
337 Specimens C7 and C8; and Specimens C9 and C10. From this figure, for drifts up to about 2%  
338 (i.e., drifts associated with service- and/or design-basis seismic events), the employed lateral  
339 loading protocol does not practically affect the first-cycle envelope curve of a steel column. This  
340 is important if the objective is to evaluate the immediate occupancy of a steel-frame building after  
341 a design-basis seismic event. On the other hand, for drifts larger than 2%, specimens subjected to

342 a symmetric lateral loading protocol deteriorate much faster in flexural strength than those  
343 subjected to a collapse-consistent loading history. In particular, the plastic deformation capacity  
344 of steel columns subjected to the latter protocol is twice larger than that of nominally identical  
345 columns subjected to the former (see Fig. 11). This is attributed to the extent of inelastic cumulative  
346 damage due to the relatively large number of inelastic cycles of a symmetric cyclic loading history.

347 Interestingly, from Fig. 4 at 4% drift, test specimens subjected to a collapse-consistent loading  
348 protocol shortened, on average, by 0.6%  $L$ . This is five time less than the average amount of axial  
349 shortening measured in nominally identical columns subjected to a symmetric cyclic loading  
350 history (i.e., 2.7%  $L$ ). Therefore, experimental data from symmetric loading histories would be  
351 overly conservative for the performance-based seismic evaluation of steel-frame buildings  
352 subjected to low probability of occurrence earthquakes.

353 In brief, the aforementioned facts underscore the importance of utilizing realistic loading  
354 histories for the calibration of component deterioration models employed for the earthquake-  
355 induced collapse assessment of frame buildings. Such protocols should capture the ratcheting  
356 behavior that a building and its structural components experience prior to structural collapse. These  
357 findings are in agreement with past collapse-related studies (FEMA 2009; Krawinkler 2009;  
358 Lignos et al. 2011; Suzuki and Lignos 2014, 2015).

### 359 **Effect of bidirectional lateral loading**

360 The experimental program offers the opportunity to characterize the hysteretic behavior of steel  
361 columns subjected to bidirectional lateral loading coupled with compressive axial load and further  
362 assess their performance with respect to nominally identical specimens subjected to unidirectional  
363 lateral loading. Referring to Fig. 3, three pairs of specimens are compared for this purpose;  
364 Specimens C3 and C6; Specimens C7 and C9; and Specimens C8 and C10. From this figure, for



365 all practical purposes, the rate of flexural strength deterioration as well as the plastic deformation  
366 capacity of the examined steel columns is not influenced by the bidirectional loading especially  
367 prior to the 3% drift amplitude in the column's strong-axis bending direction, regardless of the  
368 employed cross-section and the lateral loading history. At larger drifts, the observed differences in  
369 the column's flexural strength deterioration are on the order of 15% or less. These differences are  
370 primarily attributed to the generally larger out-of-plane deformations measured in the plastic hinge  
371 region near the column base of the specimens that experienced a bidirectional lateral loading (see  
372 Fig. 8b) compared to those that experienced unidirectional loading (see Fig. 8a).

373 Representative deduced moment-rotation relations with respect to the weak-axis of steel  
374 columns are shown in Fig. 12a and 12b for Specimens C6 and C10, respectively. In Fig. 12a, the  
375 moment-rotation behavior following the onset of local buckling at the column base is highlighted  
376 with a dashed line. From Fig. 12a, up to about 2% drift in the weak-axis orientation, the hysteretic  
377 behavior of Specimen C6 is fairly stable without any observed weak-axis flexural strength  
378 deterioration. Referring to Fig. 12b, Specimen C10 exhibited appreciable cyclic flexural strength  
379 deterioration in the weak-axis bending direction. This is due to the fairly large inelastic cycles that  
380 this column experienced in the same loading direction (i.e., 3% drift amplitude).

381 Referring to Fig. 3, if the objective is to develop simplified backbone component models for  
382 the nonlinear modeling of steel columns in line with ASCE-41-13 (ASCE 2014), no adjustments  
383 are needed for a steel column's plastic deformation capacity due to the bidirectional loading. This  
384 effect is only reflected in the column's flexural strength due to axial load-bi-directional bending  
385 interaction ( $P-M_x-M_y$ ). With reference to Figs. 4g and 4i, at a given drift amplitude in the strong-  
386 axis orientation, the amount of column axial shortening is practically not influenced by the  
387 bidirectional loading. Same findings hold true regardless of the employed cross-section.

388 The test results suggest that specimens subjected to bidirectional loading developed the center  
389 of local buckling further away from the column base compared to those subjected to unidirectional  
390 loading. For example, the center of local buckling was located at  $0.7 d$  from the column base for  
391 Specimen C10 compared to  $0.4 d$  for Specimen C8. This is attributed to the member P-delta  
392 demands about the column's weak-axis in the case of bidirectional loading.

393 Steel columns subjected to bidirectional lateral loading experienced larger out-of-plane  
394 deformations compared to those subjected to unidirectional lateral loading. This observation was  
395 more evident in W24x84 columns (e.g., Specimens C7 and C9) at story-drifts larger than 3% (see  
396 Figs. 8a and 8b). These specimens are more susceptible to out-of-plane instabilities due to their  
397 larger member slenderness ratio ( $L_b/r_y$ ) and torsional slenderness ratio ( $\lambda_{LTB}$ ) compared to  
398 W24x146 columns (see Table 1). This caused the unloading stiffness of Specimen C9 to deteriorate  
399 more than that of Specimen C7 (see Fig. 3f and 3h).

#### 400 **Column plastic hinge length and comparisons with available empirical equations**

401 The column plastic hinge length,  $L_{PH}$ , was systematically evaluated for all the test specimens  
402 based on the uniaxial strain gauge measurements recorded along their height. Referring to Fig. 13,  
403  $L_{PH}$  is defined as the distance between the column base and the cross-section at which the uniaxial  
404 engineering strain exceeds the measured engineering yield strain,  $\epsilon_y$ . From the same figure, this  
405 location is traced by conducting a linear interpolation between the engineering strain  
406 measurements at cross-section level #2 ( $\epsilon_{2-2}$ ), which is located at 305mm (12 inches) from the  
407 column base; and cross-section level #3 ( $\epsilon_{3-3}$ ), which is located at 1270mm (50 inches) from the  
408 column base. Representative  $L_{PH}$  evolutions for four specimens subjected to various loading  
409 histories are shown in Fig. 14. In this figure,  $L_{PH}$  is normalized with respect to the employed cross-  
410 section depth,  $d$ . From this figure, prior to the onset of local buckling near the column base, the

411 progression of  $L_{PH}$  is due to the cyclic hardening of the steel material. This becomes evident for  
412 the more compact cross-sections (see Figs. 14a-b) due to the delayed onset of local buckling.  
413 Referring to Fig. 14, after the local buckling formation, the plastic hinge length stabilizes due to  
414 the localization of plastic strains within the buckled region. This is confirmed in Fig. 15 that shows  
415 the normalized  $L_{PH}$  for all the tested specimens at the end of each test and at a reference story-drift  
416 of 2% that in most cases local buckling did not occur.

417 Referring to Figs. 14a and 14b, while the applied compressive axial load increases the larger  
418 the plastic hinge length becomes. In particular, Specimen C2 developed a 15% larger plastic hinge  
419 length compared to Specimen C1. This is due to the second-order moment that pushes the  
420 maximum moment demands (i.e., first and second order moment) further away from the column  
421 base. This was also observed in specimens subjected to bidirectional lateral loading (see Fig. 14d).  
422 In this case, the weak-axis bending demands due to the out-of-plane deformations are larger  
423 compared to those subjected to unidirectional loading (see Fig. 14c).

424 Referring to Fig. 15, Specimens C1 to C6 that utilized a W24x146 cross-section developed a  
425 plastic hinge length in the range of  $1.6 d$  to  $1.9 d$ , regardless of the employed lateral loading history  
426 and the member's end boundary conditions. These values are in agreement with the current seismic  
427 provisions in New Zealand (SNZ 2007), which specify a minimum plastic hinge length of  $1.5 d$   
428 for Category 1 and 2 members (equivalent to highly ductile and moderately ductile members as  
429 per AISC (2010a)). On the other hand, Specimens C7 to C10 that utilized a W24x84 cross-section  
430 developed a plastic hinge length in the range of  $1.25 d$  to  $1.85 d$ . Notably, SNZ (2007) specifies a  
431 lower minimum plastic hinge length of  $1.0 d$  for Category 3 members (i.e., equivalent to non-  
432 compact cross-sections as per AISC (2010a)). Similar to the SNZ (2007), the plastic hinge length  
433 may be used to evaluate the steel column stability requirements in terms of the cross-section

434 restraint spacing, against out-of-plane deformations and twisting, within the member's yielded  
 435 regions. A similar approach may be adopted in future revisions of the current North American  
 436 seismic provisions for steel MRFs (CSA 2009; AISC 2010a).

437 The expected plastic hinge length of the test specimens are calculated based on the empirical  
 438 equation developed by Kemp (1996) as follows,

$$L_{PH} = 0.067 \left( \frac{60}{\lambda_{eff}} \right)^{1.5} L_i$$

439 where,

$$\lambda_{eff} = k_f k_w k_d \left( \frac{L_i}{r_{yc}} \right) \gamma, \quad \gamma = \sqrt{\frac{F_y \text{ (MPa)}}{250}}, \quad k_f = \left( \frac{b_f}{2t_f} \right) \frac{\gamma}{9}, \quad k_w = \left( \frac{h_w}{t_w} \right) \frac{\gamma}{70} \quad (1)$$

$k_d = 1.0$ , for bare steel beams

440 in which,  $L_i$  is the distance from the cross-section with the maximum flexural strength to the  
 441 nearest inflection point;  $r_{yc}$  is the radius of gyration of the compressive region. The  $L_{PH}$  values  
 442 computed by Eq. (1) were, on average, equal to  $1.67 d$  and  $1.54 d$  for W24x146 and W24x84  
 443 columns, respectively. Referring to Fig. 15, these values are fairly close to the average ones  
 444 obtained from the measurements of the two groups of specimens (i.e., less than 5% difference).  
 445 Note that Eq. (1) is applicable to cross-sections with  $5 < b_f/2t_f < 11$  and  $39 < h/t_w < 85$ . The cross-  
 446 sections that were utilized in the test program fall into this range. Although it is difficult to  
 447 generalize the experimental findings as the tests cover only a limited range of local slenderness  
 448 ratios, it seems that Kemp's equation can be employed to estimate the plastic hinge length of steel  
 449 columns that utilize slender cross-sections near the current compactness limits for highly ductile  
 450 and moderately ductile members according to AISC (2010). However, this finding should be  
 451 further verified for stockier members. The authors are currently evaluating this issue through  
 452 parametric finite element analyses (Elkady and Lignos 2015a, 2017).

453 **Column axial shortening and comparisons with available predictive equations**

454 Column axial shortening is typically neglected in column stability checks in North America. It  
455 directly relates to the cumulative plastic rotation ( $\sum\theta_{pl}$ ) that a member experiences during reversed  
456 cyclic loading (MacRae et al. 1990), which in turn depends on the employed lateral loading history.  
457 Figure 16 shows the measured amount of axial shortening for all the specimens at selected  $\sum\theta_{pl}$   
458 values of 0.1, 0.25 and 0.5 rads. To put these values into perspective, they correspond roughly to  
459 the amount of cumulative plastic rotation measured at the first cycle of 1%, 2% and 4% drift  
460 amplitudes of an equivalent symmetric cyclic loading protocol. In the context of this paper,  $\sum\theta_{pl}$   
461 is computed by assuming an elastic-perfectly plastic hysteretic behavior and a yield rotation,  $\theta_y = Z_x$   
462  $f_{ye} (I-P/P_y)/K_e$  ( $Z_x$  is the plastic section modulus;  $f_{ye}$  is the expected yield stress of the steel  
463 material). From Fig. 16, at  $\sum\theta_{pl} < 0.25$  rads (i.e., equivalent to 2% drift), nominally identical  
464 specimens experience the same amount of axial shortening at a given cumulative plastic rotation  
465 regardless of the employed lateral loading protocol (i.e., collapse-consistent versus symmetric  
466 cyclic and/or bidirectional versus unidirectional). In particular, column axial shortening is less than  
467 1% of the member length,  $L$ , at  $\sum\theta_{pl} < 0.25$  rads. Therefore, it should not become a controlling  
468 issue for design-basis seismic events (i.e., 10% probability of occurrence in 50 years). However,  
469 column axial shortening grows exponentially at  $\sum\theta_{pl} \sim 0.50$  rads (i.e., equivalent to 4% drift).  
470 Therefore, this failure mode could become controlling for collapse prevention during seismic  
471 events with low probability of occurrence (i.e., 2% probability of occurrence over 50 years).

472 MacRae et al. (1990) utilized the experimental data from small-scale cantilever column testing  
473 to develop an empirical equation to estimate the amount of column axial shortening ( $\Delta_{axial}$ ),

$$\begin{aligned}
\Delta_{axial} &= 0.446 \frac{P}{2.54 P_y} \frac{A}{A_w} L_{PH} \sum \theta_{pl} && \text{for } \frac{P}{P_y} \leq \frac{2.54 A_w}{A} \\
&= 0.446 L_{PH} \sum \theta_{pl} && \text{for } \frac{P}{P_y} > \frac{2.54 A_w}{A}
\end{aligned} \tag{2}$$

in which  $A_w$  is the web area,  $A$  is the gross area, and  $L_{PH}$  is the column plastic hinge length. The computed column axial shortening for all ten specimens based on Eq. (2) is superimposed in Fig. 16 for the three values of  $\sum \theta_{pl} = 0.1, 0.25, \text{ and } 0.5$  rads. For these calculations, the assumed plastic hinge length,  $L_{PH}$  is equal to the measured values for each specimen (see Fig. 15). For fixed-end columns (i.e., Specimens C1 and C2), the calculated axial shortening was multiplied by a factor of two to account for the simultaneous formation of plastic hinges at the member ends. Referring to Fig. 16, Eq. (2) seems to reasonably predict the axial shortening of columns subjected to a symmetric lateral loading protocol (i.e., C1, C2, C3, C6, C7, and C9), at cumulative plastic rotations of 0.25 rads or less. In this range, the  $\Delta_{axial} - \sum \theta_{pl}$  relation is fairly linear as implied by Eq. (2). However, if  $\sum \theta_{pl} > 0.25$ , Eq. (2) significantly underestimates the column axial shortening. This is due to its exponential increase with local buckling progression (Elkady and Lignos 2015a). This issue should be further considered in future studies.

### **Out-of-plane bracing force demands and comparisons with commonly used equations for predicting strength of nodal brace axial forces**

The 6-DOF test setup offers the opportunity to measure the out-of-plane bracing forces acting at the top end of a column specimen under unidirectional lateral loading. To the best of the authors' knowledge, the out-of-plane force demands have not been evaluated experimentally in prior studies. In that respect, the experimental data set is considered to be unique.

Figure 17 shows representative out-of-plane force demands,  $F_x$ , versus the true chord-rotation for six specimens. The out-of-plane force is normalized with respect to  $P_y$ . From Figs. 17,

495 W24x146 and W24x84 columns developed, on average, a maximum out-of-plane force of 1.5%  
496  $P_y$ . and 0.8%  $P_y$ , respectively, at their top end. As the  $L_b/r_y$  increases, the out-of-plane  
497 deformations near the plastic hinge region of a steel column increase; therefore, no significant out-  
498 of-plane forces are exerted at the column top end, regardless of the employed lateral loading  
499 protocol.

500 Section 6.4 of the ANSI/AISC 360-10 (AISC 2010b) specifies that the required nodal brace  
501 axial force strength,  $P_{rb}$ , shall be determined as the sum of the beam bracing axial force and beam-  
502 column bracing axial force,

$$503 \quad P_{rb} = 0.01 P_r + 0.02 M_r C_d / h_o \quad (3)$$

504 in which,  $P_r$  and  $M_r$  are the column's required axial and flexural strength, respectively;  $C_d = 2.0$   
505 for braces closest to the column inflection point; and  $h_o$  is the distance between the flange  
506 centroids. Similarly, Clause 9.2.5 of CSA (2009) specifies a lateral brace axial strength,  $P_b$ , larger  
507 than 2% of the factored compressive force,  $C_f$ , of the element being braced laterally,

$$508 \quad P_b = 0.02 C_f = 0.02 \cdot 1.1 f_{ye} A_{comp} \quad (4)$$

509 in which,  $f_{ye}$  is the expected yield stress; and  $A_{comp}$  is the cross-sectional area subjected to  
510 compressive stresses. The out-of-plane force demands that were measured during the testing  
511 program are utilized to assess the adequacy of the brace design axial forces calculated by Eqs. (3)  
512 and (4). These forces are calculated and superimposed in Fig. 17. In Eq. (3),  $P_r$  is assumed to be  
513 equal to the applied compressive axial load ratio to the corresponding specimen; and  $M_r$  is assumed  
514 to be equal to the reduced plastic flexural strength based on AISC (2010b)  $P$ - $M$  interaction  
515 equations. In Eq. (4),  $A_{comp}$  is calculated by assuming that 65% and 100% of the cross-section  
516 depth is under compression when subjected to 20%  $P_y$  and 50%  $P_y$ , respectively. These values

517 were estimated from a stress distribution that was obtained once flexural yielding initiated in the  
518 respective cross-section. Referring to Fig. 17, the lateral brace design axial force as per CSA (2009)  
519 and AISC (2010b) for the W24x146 columns seems adequate for story-drift-ratios up to about 2%.  
520 However, at larger drift demands, the out-of-plane axial force demands exceed the lateral bracing  
521 design axial force based on Eq. (3) and (4) by 35% and 15%, respectively. On the other hand, the  
522 lateral brace design axial force seems fairly conservative for all the W24x84 columns regardless  
523 of the employed lateral loading history and the corresponding lateral drift demands. These  
524 observations suggest that the lateral brace design axial force requirements for steel columns in  
525 MRFs should be carefully revisited.

## 526 **Summary and Conclusions**

527 This paper presents findings and design implications based on 10 full-scale tests of deep and  
528 slender (with local slenderness near the AISC 341-10  $\lambda_{hd}$  limits) W24 (i.e., 600mm deep) first-  
529 story steel columns subjected to various cyclic loading histories. The test specimens represent  
530 interior first-story steel columns in capacity-designed steel MRFs. Several key parameters,  
531 including the member end boundary conditions, loading sequence, local web and member  
532 slenderness are interrogated. The effects of bidirectional versus unidirectional lateral loading  
533 histories were also examined by conducting tests on nominally identical specimens. The lateral  
534 loading histories were either symmetric cyclic or collapse-consistent such that ratcheting prior to  
535 structural collapse was considered. The main findings are summarized as follows,

- 536 • Qualitatively, the test specimens with fixed-flexible boundary conditions followed a similar  
537 damage progression. Web and flange local buckling (at displacements corresponding to 1.5%-  
538 2% story-drift) formed at a distance of  $0.5 d$  to  $0.7 d$  from the column base. Subsequently, local  
539 buckling caused column axial shortening, which in turn triggered out-of-plane deformations



540 that became maximum at the center of the plastic hinge region near the column base. The out-  
541 of-plane deformations caused considerable weak-axis bending demands due to member P-delta  
542 effects. Notably, these deformations were not evident in fixed-end test specimens. The out-of-  
543 plane deformations were often followed by twisting at drifts larger than 3%. The twist angle  
544 magnitude is dependent on the member and torsional slenderness ratios.

545 • The experimental program suggests that it may be fairly misleading to characterize the  
546 hysteretic behavior of steel columns under multi-axis loading with simplified fixed-fixed  
547 boundary conditions. In this case, the torsional restraint at both member ends is lost  
548 simultaneously after the onset of local buckling, which is not typical for first-story columns in  
549 capacity-designed steel MRFs due to the employed strong-column/weak-beam ratio; and  
550 therefore, member geometric instabilities are more likely to occur at fairly small lateral drifts  
551 compared to reality. For instance, at story-drifts up to 3% rads, test specimens with  $L_b/r_y=79$   
552 and fixed-flexible boundary conditions were able to maintain 80% of their maximum flexural  
553 strength as well as 70% of their elastic stiffness. The same specimens experienced minimal  
554 twisting up to this drift range.

555 • The tests suggest that the current CSA S16-09 (CSA 2009) standards may be fairly  
556 conservative by setting an  $L_b/r_y \sim 60$  limit for the steel column design in Type-D steel MRFs  
557 (i.e., equivalent to SMFs according to the AISC 2010a provisions). A modified upper limit for  
558 the member and torsional slenderness ratios should be adopted in future versions of the CSA  
559 (2009) and AISC (2010a) provisions for collapse prevention of SMFs. This requires additional  
560 research studies.

561 • Axial shortening is a controlling failure mode in steel columns undergoing reversed cyclic  
562 loading. At story-drifts representative of design-basis earthquakes (i.e., 2% rads), axial

563 shortening ranged from 0.3%-0.5%  $L$  for specimens subjected to a compressive axial load of  
564 20%  $P_y$ . However, a W24x146 column subjected to 50%  $P_y$ , experienced 2.5%  $L$  axial  
565 shortening at the same drift-ratio. This indicates that an upper limit should be set to the  
566 allowable compressive axial load for the seismic design of steel columns in steel SMFs in future  
567 revisions of the AISC (2010) provisions. In that respect, the employed 30%  $P_y$  limit according  
568 to the Canadian seismic provisions (CSA 2009) for ductile steel MRFs seems to be rational.

569 • The tests reveal that the column axial shortening is strongly dependent on the cumulative plastic  
570 rotation. This agrees with MacRae et al. (1990). MacRae's column axial shortening predictive  
571 empirical equation seems adequate for drift-ratios up to 2% or less. In this range, column axial  
572 shortening is linearly dependent on the cumulative plastic rotation. In the examined cases  
573 herein, the same equation seems to under predict the column axial shortening by more than 50%  
574 at drifts larger than 2%. In this drift range, the axial shortening increases exponentially with  
575 respect to the cumulative plastic rotation; this is due to the rapid progression of web local  
576 buckling in the plastic hinge region.

577 • A W24x146 steel column subjected to a symmetric cyclic loading history coupled with a  $P/P_y$   
578 = 50% (i.e.,  $P/P_{cr} > 50%$ ) developed an appreciable plastic deformation capacity prior to the  
579 loss of its axial load carrying capacity. Although inconclusive, this suggests that the ASCE/SEI  
580 41-13 (ASCE 2014) recommendations for force-controlled elements may be fairly  
581 conservative. This issue should be examined in future studies.

582 • The plastic deformation of steel columns subjected to a collapse-consistent loading protocol  
583 was at least twice larger than those subjected to a symmetric cyclic loading protocol. Notably,  
584 at large drifts (i.e., larger than 4%), steel columns subjected to a collapse-consistent loading  
585 protocol shortened 5 times less than those subjected to a symmetric loading protocol. These

586 findings underscore the importance of utilizing realistic loading histories for characterizing the  
587 “ratcheting” hysteretic behavior (drifting in one direction) of structural components from the  
588 onset of damage through structural collapse.

589 • The test results suggest that, steel columns subjected to bidirectional lateral loading develop the  
590 center of the local buckling wave further away from the column base compared to those  
591 subjected to unidirectional lateral loading. This is due to the increased weak-axis flexural  
592 demands because of the weak-axis lateral drift as well as the increased member P-delta. These  
593 effects were more pronounced for W24x84 columns in which  $L_b/r_y = 79$ . However, if the  
594 objective is to develop simplified backbone component models for nonlinear modeling of steel  
595 columns to conduct nonlinear static analysis of steel MRFs, no adjustments are necessary to the  
596 plastic deformation capacity of steel columns due to bidirectional lateral loading.

597 • The developed plastic hinge length near the column base was in the range of  $1.25 d$  to  $1.85 d$   
598 for W24x84 columns. Stockier W24x146 columns developed, on average, a larger plastic hinge  
599 length of  $1.6 d$  to  $1.9 d$  due to material cyclic hardening prior to the onset of geometric  
600 instabilities. These values are fairly consistent with the ones reported in the New Zealand  
601 seismic provisions for the design of steel MRFs (SNZ 2007). It was found that the empirical  
602 equation developed by Kemp (1996) can be employed to estimate the expected plastic hinge  
603 length of steel columns that utilize slender cross-sections near the current compactness limits  
604 for highly ductile and moderately ductile members according to AISC (2010). This conclusion  
605 should be verified for stockier members in future studies.

606 • Comparisons of measured and calculated out-of-plane bracing force demands in steel columns  
607 generally confirm expectations only for the stockier W24x146 columns ( $L_b/r_y=51$  and  
608  $\lambda_{LTB}=0.28$ ) up to story-drifts of 2% or less, regardless of the employed lateral loading protocol.

609 At larger drift demands, the out-of-plane axial force demands exceeded the lateral bracing  
610 design axial force according to the CSA (2009) and AISC (2010b) specifications by 15% and  
611 35%, respectively, for the same cross-sections. On the other hand, the calculated lateral brace  
612 design axial force seems to be fairly conservative for all the W24x84 columns ( $L_b/r_y=79$  and  
613  $\lambda_{LTB}=0.42$ ) regardless of the corresponding lateral drift demands and the employed lateral  
614 loading history.

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**Table 1.** Test matrix summary and measured geometric and material properties

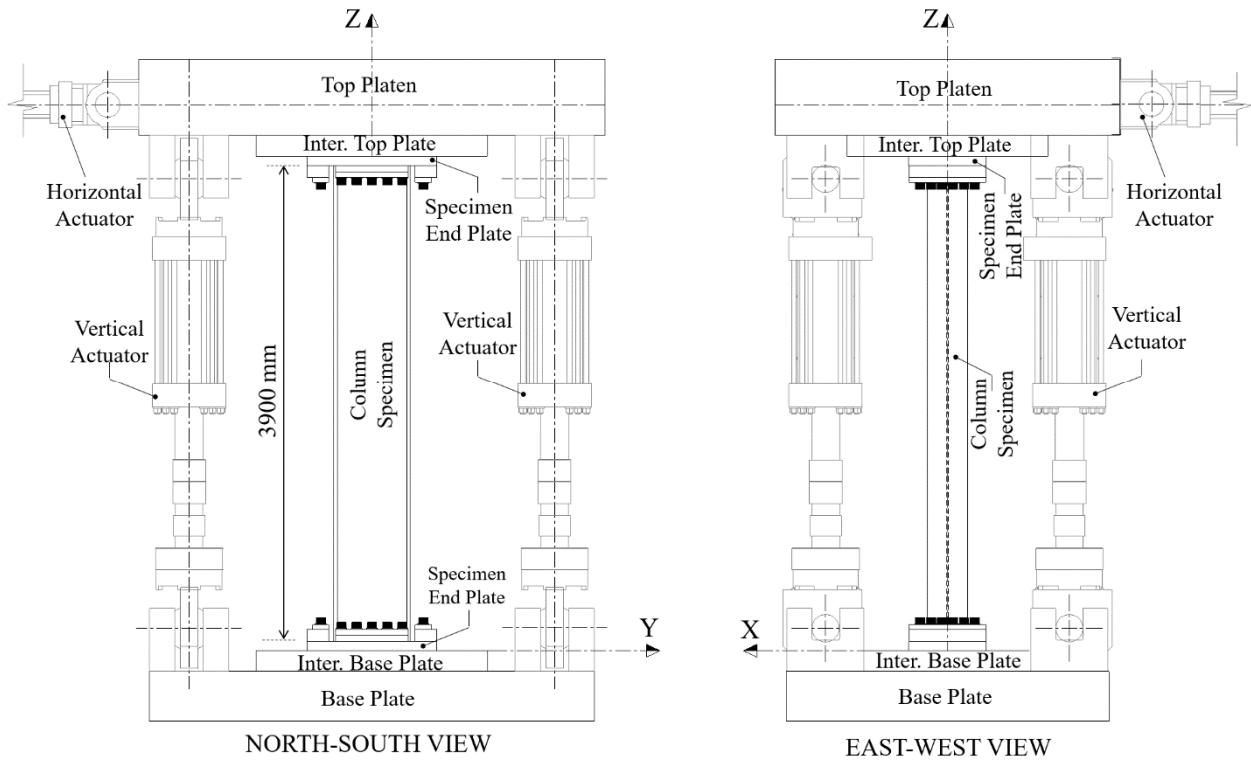
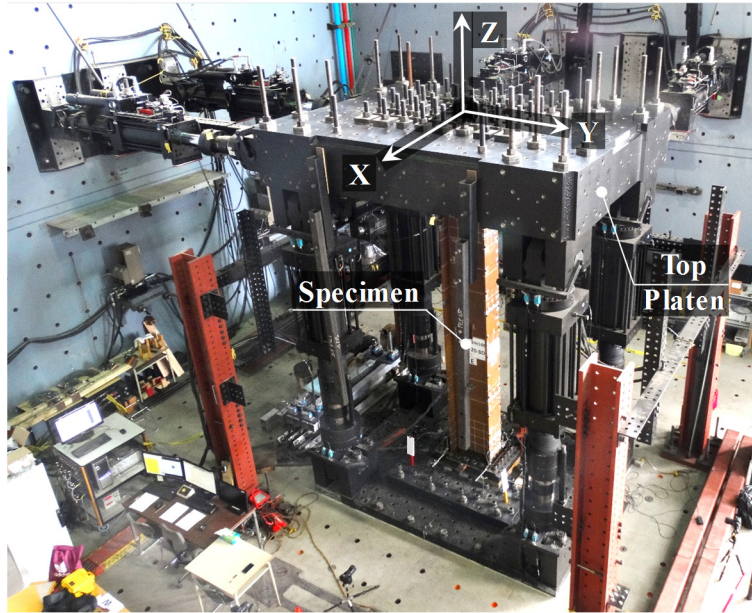
ID	Section size	Lateral loading protocol	$\frac{P}{P_y}$	BCs in the strong-axis direction	Cross-section and member properties <sup>a</sup>						Measured material properties <sup>b</sup>				
					$\frac{b_f}{2t_f}$	$\frac{h}{t_w}$	$\frac{L_b}{r_y}$	$J$ [mm <sup>4</sup> ]	$C_w$ [mm <sup>6</sup> ]	$\lambda_{LTB}$	$E$ [MPa]	Flange		Web	
												$f_{yf}$	$f_{uf}$	$f_{yw}$	$f_{uw}$
C1	W24x146	SYM (UD)	-0.2	Fixed-Fixed	6.1	33.3	51.7	5.1x10 <sup>6</sup>	1.4x10 <sup>13</sup>	0.28	190481	414	509	415	502
C2	W24x146	SYM (UD)	-0.5	Fixed-Fixed	6.1	33.1	51.5	5.1x10 <sup>6</sup>	1.4x10 <sup>13</sup>	0.28	190481	414	509	415	502
C3	W24x146	SYM (UD)	-0.2	Fixed-Flexible	6.1	33.5	51.5	5.0x10 <sup>6</sup>	1.4x10 <sup>13</sup>	0.32	190481	414	509	415	502
C4	W24x146	CPS (UD)	-0.2	Fixed-Flexible	6.1	33.3	51.7	5.1x10 <sup>6</sup>	1.4x10 <sup>13</sup>	0.28	190481	414	509	415	502
C5	W24x146	CPS (UD)	-0.2	Fixed-Flexible	6.0	32.5	52.1	5.2x10 <sup>6</sup>	1.4x10 <sup>13</sup>	0.30	204413	368	483	378	479
C6	W24x146	SYM (BD)	-0.2	Fixed-Flexible	5.9	32.2	52.3	5.3x10 <sup>6</sup>	1.4x10 <sup>13</sup>	0.30	204413	368	483	378	479
C7	W24x84	SYM (UD)	-0.2	Fixed-Flexible	6.1	47.0	79.2	1.6x10 <sup>6</sup>	3.3x10 <sup>13</sup>	0.42	195203	332	507	345	508
C8	W24x84	CPS (UD)	-0.2	Fixed-Flexible	6.1	47.0	79.2	1.6x10 <sup>6</sup>	3.3x10 <sup>13</sup>	0.42	195203	332	507	345	508
C9	W24x84	SYM (BD)	-0.2	Fixed-Flexible	6.1	47.7	79.4	1.5x10 <sup>6</sup>	3.2x10 <sup>13</sup>	0.42	195203	332	507	345	508
C10	W24x84	CPS (BD)	-0.2	Fixed-Flexible	6.1	47.4	79.6	1.5x10 <sup>6</sup>	3.3x10 <sup>13</sup>	0.42	195203	332	507	345	508

<sup>a</sup>  $h$ : web height;  $t_w$ : web thickness;  $b_f$ : flange width;  $t_f$ : flange thickness;  $J$ : torsion constant;  $C_w$ : warping constant

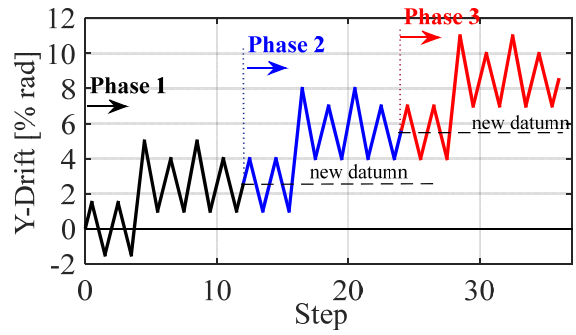
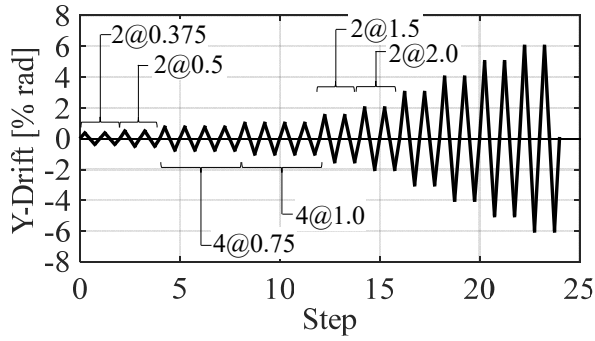
$$\lambda_{LTB} = (Z_x f_y / M_{cr})^{0.5}; \quad M_{cr} = C_1 \pi^2 E I_y / (k_y L)^2 [C_w / I_y (k_y / k_w)^2 + G J (k_y L_e)^2 / (\pi^2 E I_y)]^{0.5}$$

where  $k_w=1.0$ ,  $k_y=0.5$ , and  $C_1=2.76$  and  $2.08$  for fixed-fixed and fixed-flexible specimens, respectively

<sup>b</sup>  $E$ : elastic modulus;  $f_{yf}$ : flange yield stress;  $f_{uf}$ : flange ultimate stress;  $f_{yw}$ : web yield stress;  $f_{uw}$ : web ultimate stress

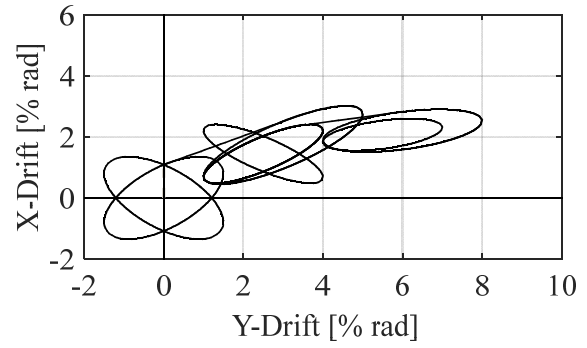
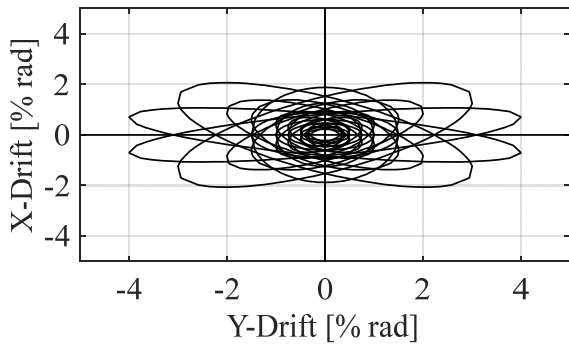


770 Fig. 1. Description of the 6-DOF test setup at École Polytechnique de Montréal



771 (a) unidirectional symmetric protocol

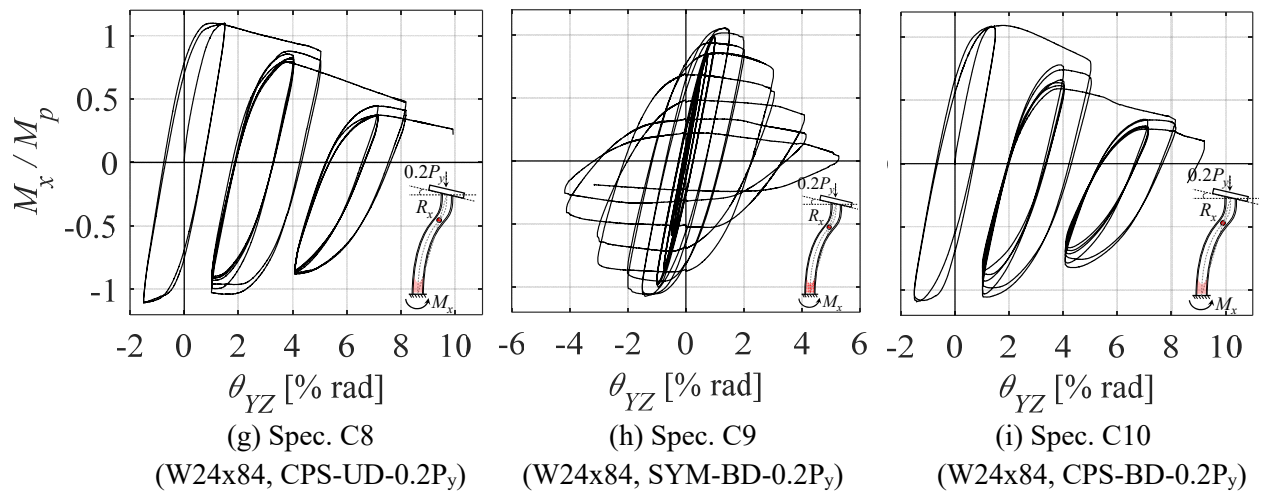
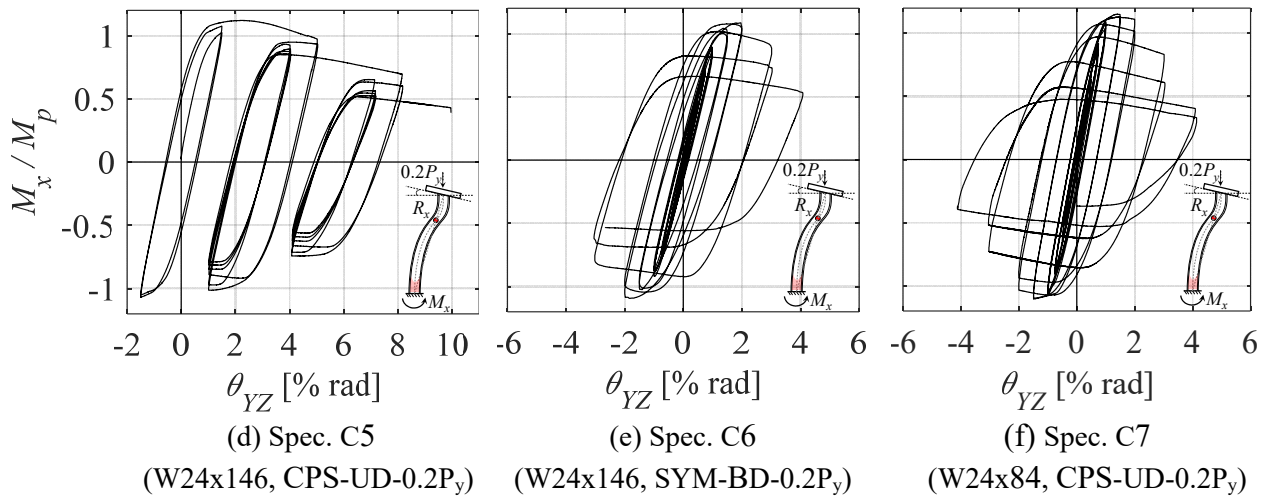
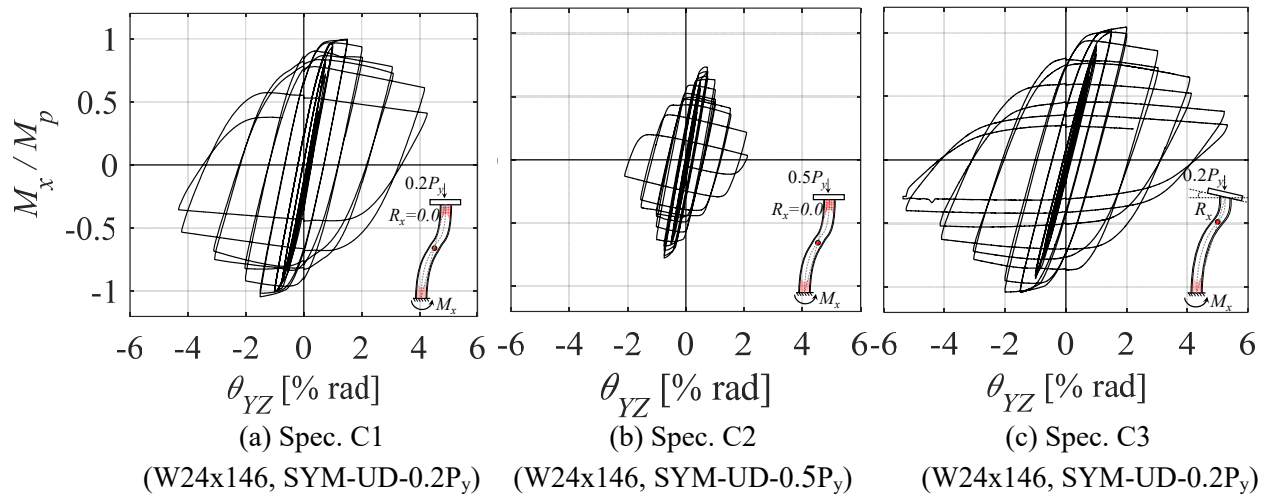
(b) unidirectional collapse-consistent protocol



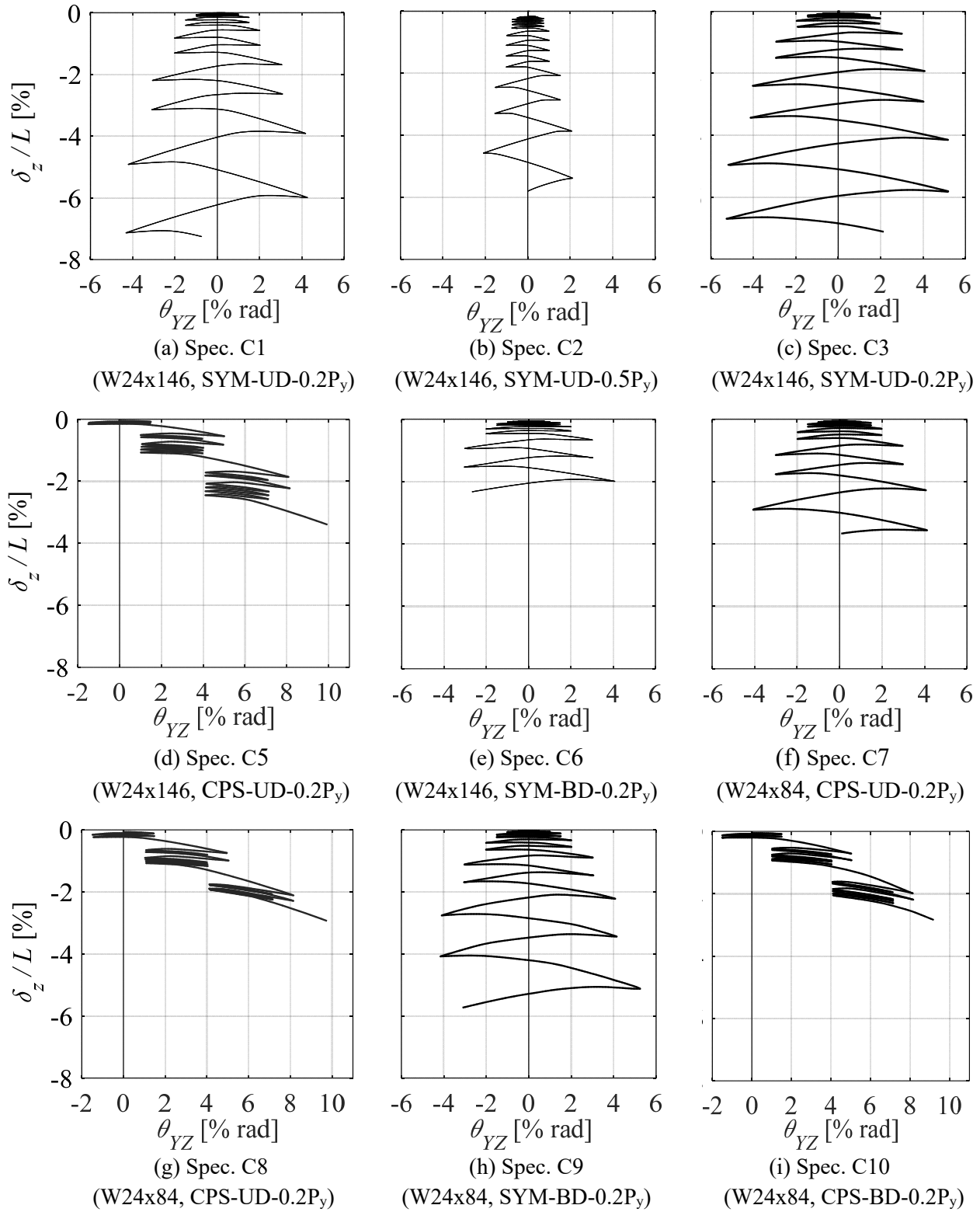
772 (c) bidirectional symmetric protocol

(d) bidirectional collapse-consistent protocol

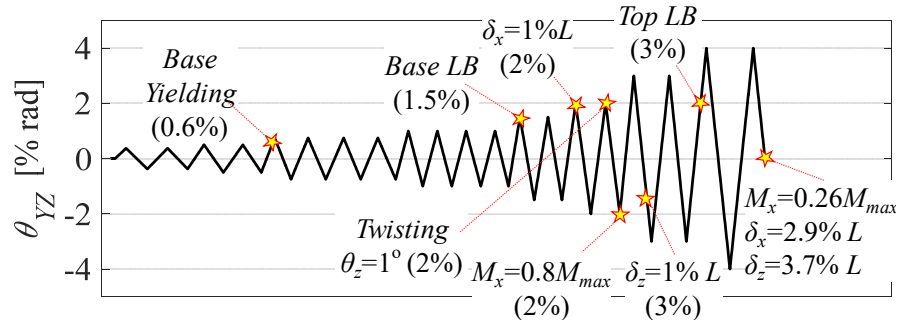
773 **Fig. 2.** Lateral loading protocols utilized in the experimental program



780 **Fig. 3.** Normalized column base end-moment versus true chord-rotation in the strong-axis  
 781 direction

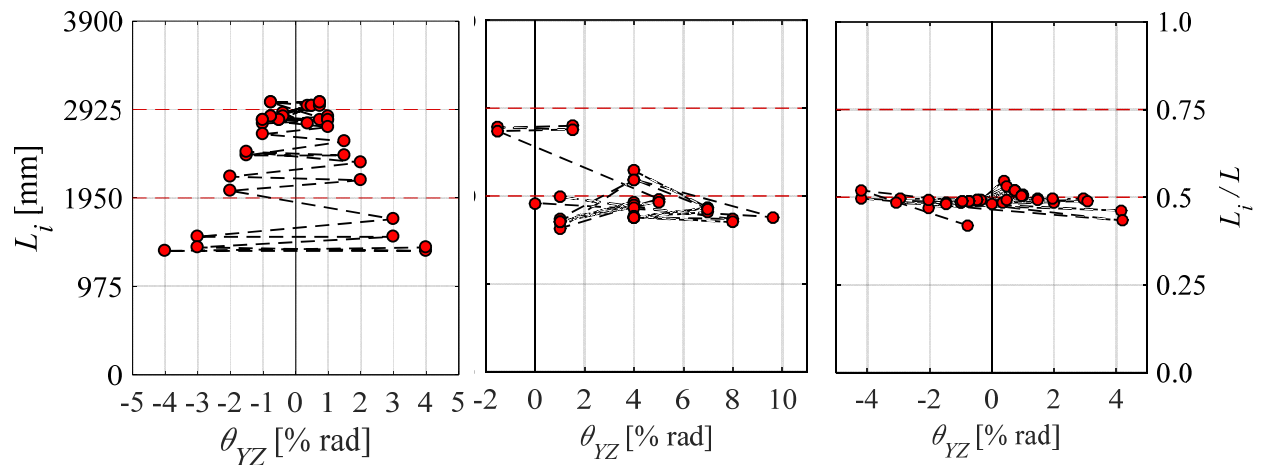


788 **Fig. 4.** Normalized column axial shortening versus true chord-rotation in the strong-axis direction



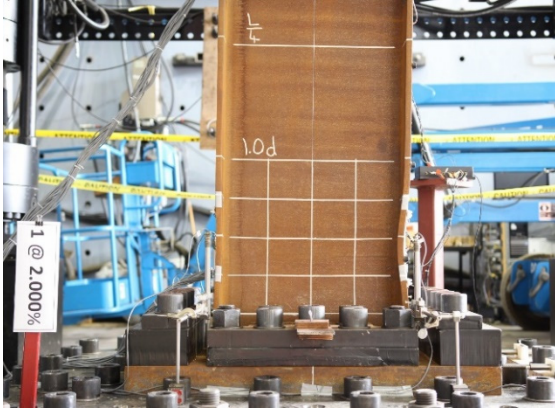
789 **Fig. 5.** Applied drift history in the strong-axis for Specimen C7 with key damage states indicated



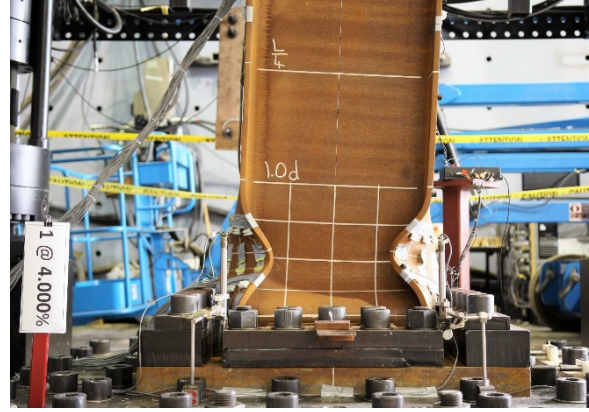


790 (a) Spec. C7 (Fixed-Flexible, SYM-UD) (b) Spec. C8 (Fixed-Flexible, CPS-UD) (c) Spec. C1 (Fixed-Fixed, SYM-UD)  
 791

792 **Fig. 6.** Inflection point location history for specimens with various boundary conditions

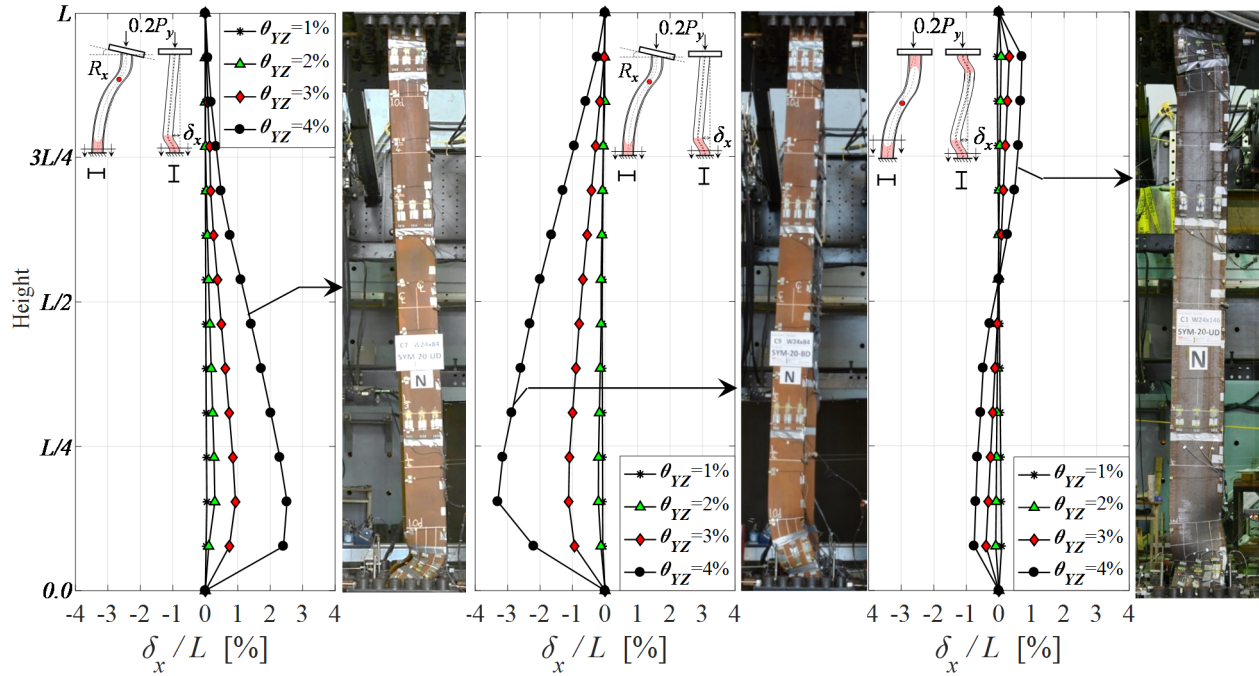


793 (a) 1<sup>st</sup> cycle, 2% drift amplitude



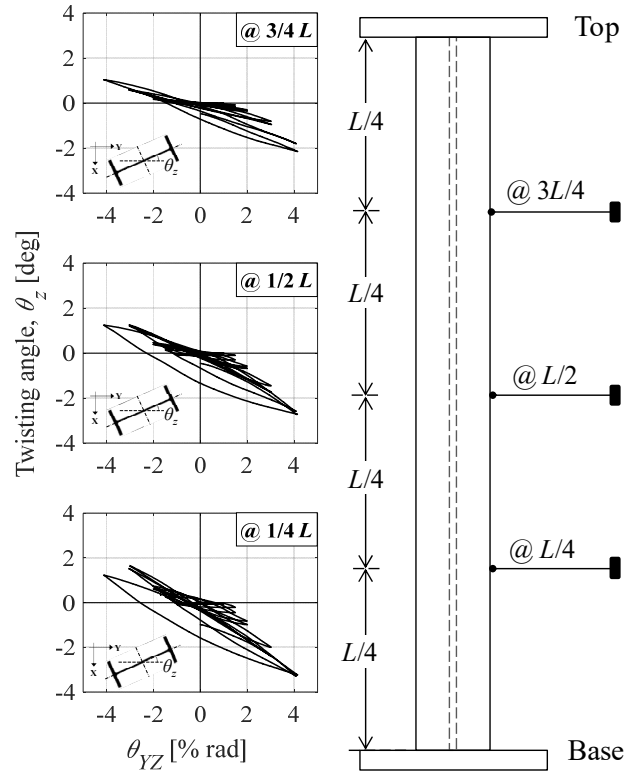
794 (b) 1<sup>st</sup> cycle, 4% drift amplitude

794 **Fig. 7.** Local buckling progression near the base of Specimen C7



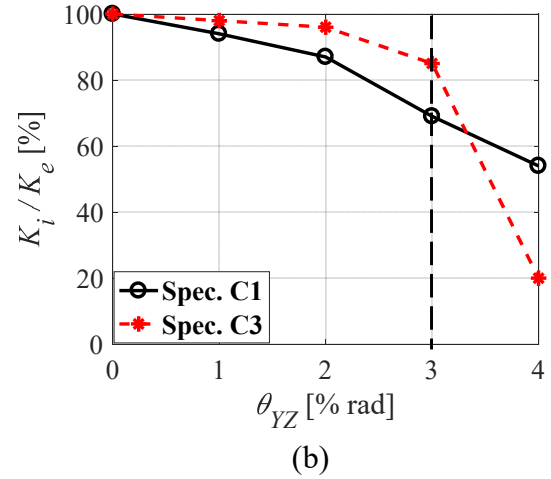
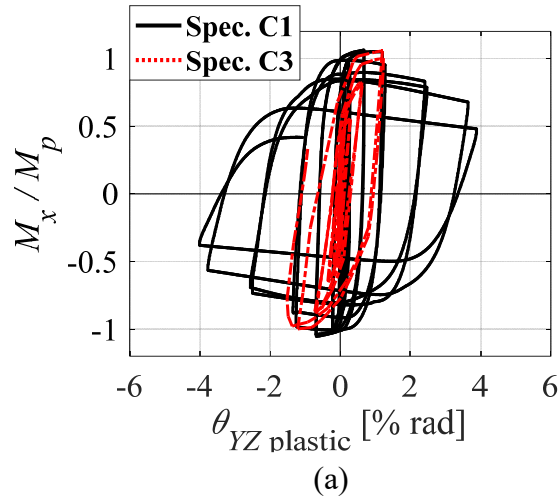
795 (a) Spec. C7  
 796 (Fixed-Flexible, SYM-UD) (b) Spec. C9 (Fixed-Flexible, SYM-BD) (c) Spec. C1 (Fixed-Fixed, SYM-UD)

797 **Fig. 8.** Out-of-plane deformation profiles at selected drift amplitudes for selected specimens



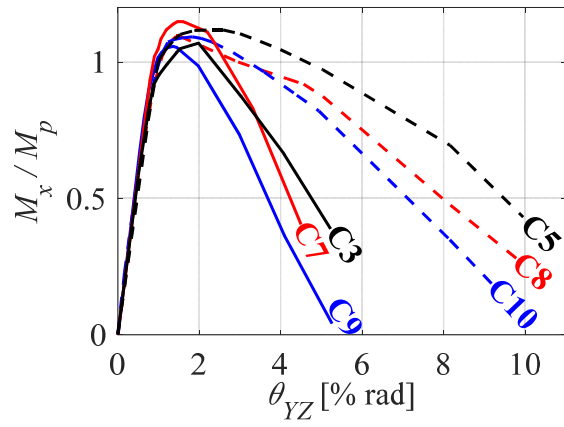
798 **Fig. 9.** Twisting angle versus true chord-rotation at different cross-sectional levels of Specimen

799 C7

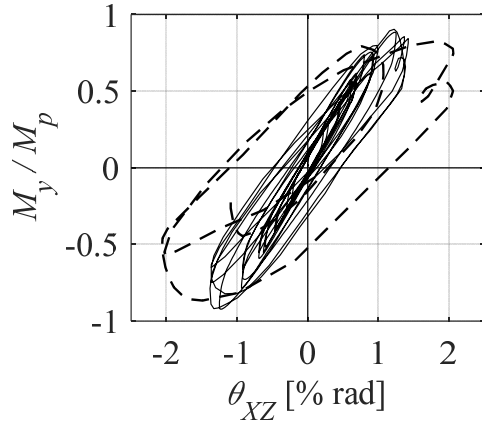


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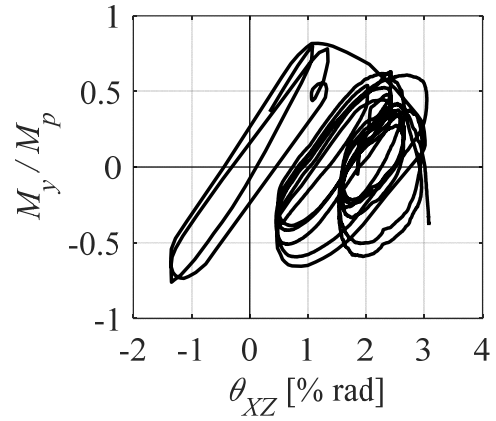
801 **Fig. 10.** Specimens C1 (fixed-fixed boundary conditions) and C3 (fixed-flexible boundary  
 802 conditions): (a) column top end moment versus plastic rotation; (b) normalized unloading stiffness  
 803 at peak drift amplitudes



804 **Fig. 11.** First-cycle envelopes for all specimens with fixed-flexible boundary condiditons subjected  
 805 to symmetric cyclic (*solid line*) and collapse-consistent loading protocols (*dashed lines*)



(a) Specimen C6

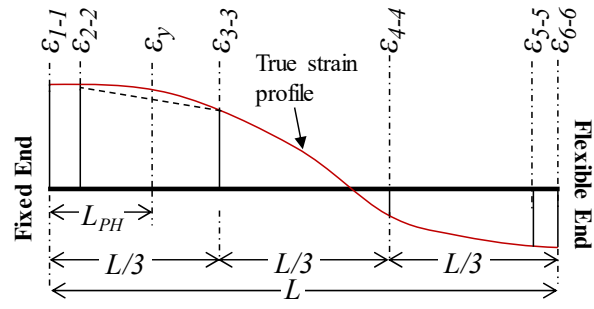


(b) Specimen C10

806

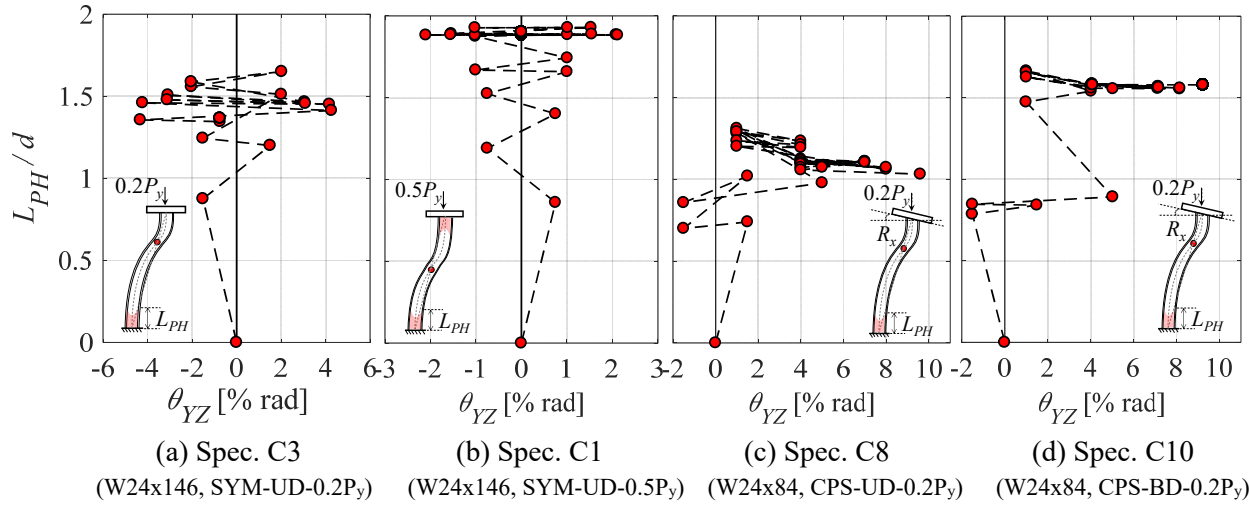
807 **Fig. 12.** Normalized column base weak-axis moment versus true chord-rotation in the weak-axis

808 direction



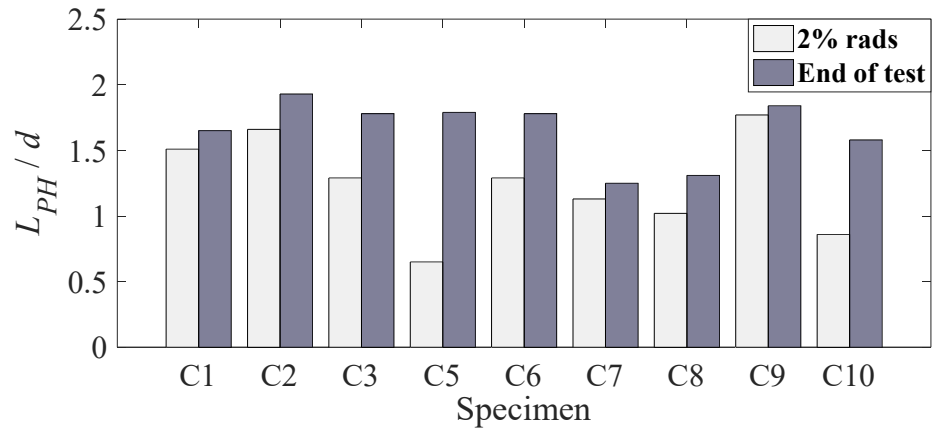
809 **Fig. 13.** Illustration of the plastic hinge length computation using strain gauge measurements



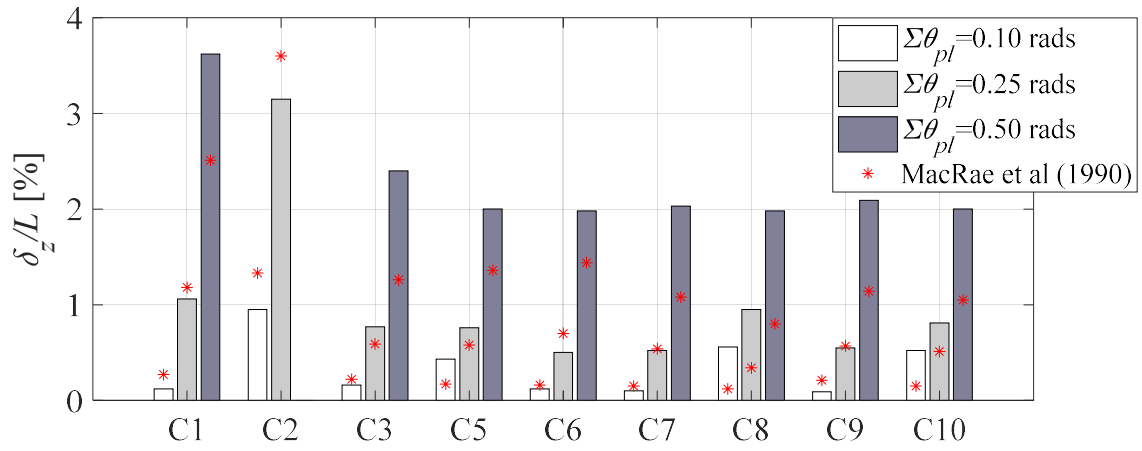


810  
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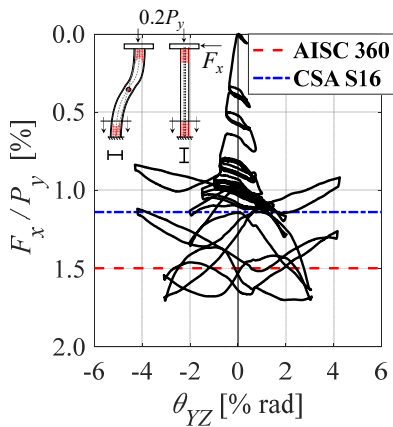
812 **Fig. 14.** Plastic hinge length at peak drifts versus true chord-rotation for selected specimens



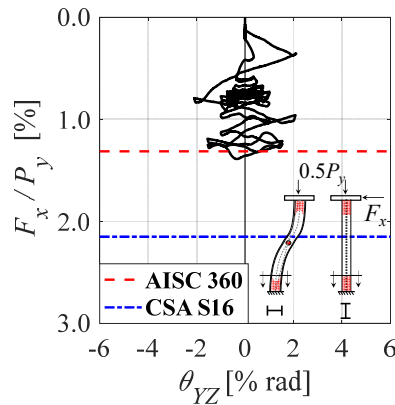
813 **Fig. 15.** Progression of plastic hinge length at selected drift amplitudes



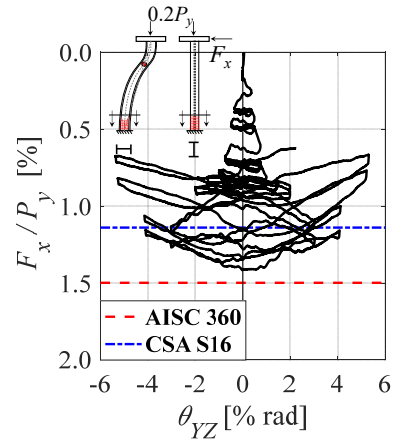
814 **Fig. 16.** Normalized column axial shortening at different levels of cumulative plastic rotation



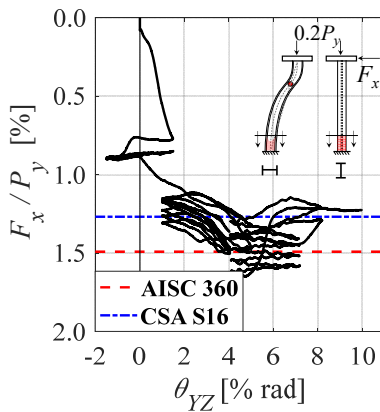
815 (a) Spec. C1 (W24x146)



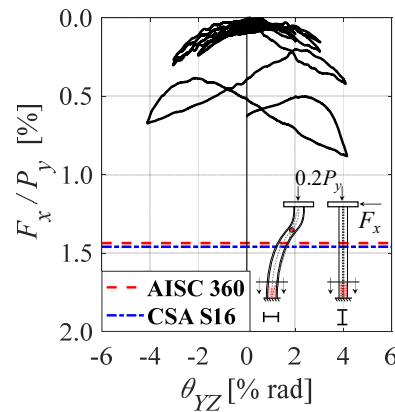
(b) Spec. C2 (W24x146)



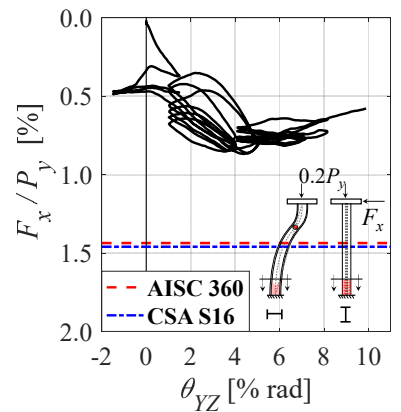
(c) Spec. C3 (W24x146)



816 (d) Spec. C5 (W24x146)



(e) Spec. C7 (W24x84)



(f) Spec. C8 (W24x84)

817 **Fig. 17.** Measured out-of-plane force demands at column top end versus true chord-rotation for  
818 selected specimens