

1 A methodology for measurement-system design combining information from
2 static and dynamic excitations for bridge load testing

3 Numa J. Bertola^{1,2}

4 Ian F.C. Smith^{1,2}

5 **Abstract**

6 Managing infrastructure assets is challenging for developed countries because of demand for
7 increases in capacity, the scarcity of economic and environmental resources as well as ageing.
8 Due to conservative approaches to construction design and practice, infrastructure often has
9 hidden reserve capacity and its estimation may improve asset-management decisions. Static
10 and dynamic bridge load testing has the potential to support engineers in their evaluation of
11 infrastructure reserve capacity if monitoring data are associated with a robust structural-
12 identification methodology. As choices of sensor types and locations directly influence
13 structural-identification outcomes, sensor-placement methodologies have recently been
14 developed to ensure successful model-updating results. Due to the nature of static and dynamic
15 measurements, sensor-placement methodologies are usually developed independently.
16 However, both types of load testing are used to update the same bridge behavior model.
17 Therefore, when sensor-placement strategies are established independently, redundant sensor
18 information is likely. In this study, two measurement-system design methodologies are
19 proposed. First, a new methodology for sensor-placement for dynamic load testing is presented,
20 where expected information gain of natural frequencies is used to prioritize sensor-location
21 selections. Then, a measurement-system-design methodology combining information of both
22 static and dynamic load testing is proposed. Finally, the methodology is evaluated using a full-
23 scale bridge. A well-designed measurement system based on expected information gain
24 enhances system identification and reserve-capacity estimation.

25 **Keywords:**

26 Structural identification; error-domain model falsification; optimal sensor placement; joint
27 entropy; modal identification; effective independence method.

¹ ETH Zurich, Future Cities Laboratory, Singapore-ETH Centre, 1 CREATE Way, CREATE Tower, 138602 Singapore.

² EPFL, Applied Computing and Mechanics Laboratory, CH-1015 Lausanne, Switzerland; E-mail: numa.bertola@epfl.ch.

28 **1 Introduction**

29 The annual global expenditure of the construction industry represents more than \$10 trillion
30 [1]. Accounting for 40% of this total, civil infrastructure must be managed efficiently. Due to
31 safe design and construction practices, infrastructure often has reserve capacity that is well
32 above the margins created by safety factors in code requirements. However, accurate reserve-
33 capacity assessment is challenging because of uncertainties in material, geometry and boundary
34 conditions. Measurement data, collected through monitoring, has the potential to support
35 engineers for reserve-capacity assessment.

36 Structural identification involves interpreting field measurements to improve knowledge of
37 structural behavior [2]. Although model-free strategies may successfully detect damage and
38 perform behavior interpolation [3], a model-based approach must be used when extrapolation
39 is needed, such as when infrastructure reserve-capacity is assessed [4].

40 In such cases, field measurements are carried out and the data are used to improve model-
41 prediction accuracy. Due to the use of complex behavior models, for instance finite-element
42 (FE) models, numerous assumptions are required and this leads to several sources of
43 uncertainties. Traditional approaches to structural identification, such as residual minimization
44 and Bayesian model updating, usually assume that uncertainties have zero-mean Gaussian
45 forms [5]–[7]. Since structural models are designed to be safe, the zero-mean assumption is not
46 satisfied in this context [8]. Furthermore, the estimation of prediction-error correlations
47 between sensor locations is challenging as usually, little information is available [9]. Although
48 modifications to traditional implementations of Bayesian model updating are possible, they
49 lead to complex system-identification formulations [10], [11]. Introduced by Goulet and Smith
50 [12], error-domain model falsification (EDMF) is an easy-to-use structural-identification
51 approach. Compared with traditional Bayesian updating, EDMF provides more accurate (albeit
52 less precise) model-parameter identification since it represents explicitly systematic
53 uncertainties in a way that is compatible with practical engineering knowledge [13]. Recently,
54 EDMF has been applied to full-scale case studies to assess the reserve-capacity and to evaluate
55 the worst-case load capacity [14], [15].

56 Structural-identification outcomes are directly dependent on the design of the measurement
57 system. Surprisingly, sensor types and positions are typically chosen using only qualitative
58 rules of thumb arising from engineering experience. Quantitative studies on optimal sensor
59 placement have been recently carried out to maximize the information gain by sensor
60 configurations for both static and dynamic load tests [16], [17].

61 Although load testing is often used for identification, static and dynamic load tests can provide
62 unique information. Information may also be redundant [18]. Due to their nature, measurement
63 systems for static and dynamic tests are typically designed independently. Measurement-
64 system-design methodologies should involve information from static and dynamic tests
65 simultaneously in order to maximize the total expected information gain.

66 For both static and dynamic tests, finding the optimal sensor configuration is usually
67 formulated as a simple optimization task. The computational complexity of the sensor-
68 placement algorithm is exponential with respect to the number of sensors [19]. Although
69 global-search optimization algorithms have been proposed [20], most researchers have
70 preferred to use sequential searches (greedy algorithms) to reduce the computational effort
71 when there are more than a few sensors [21].

72 Several approaches, such as either minimizing the information entropy in posterior model-
73 parameter distributions [22], [23] or maximizing information entropy in multiple-model
74 predictions [24], [25] have been developed to evaluate sensor locations in terms of their
75 performance for model-parameter estimations. Most authors have disregarded the mutual
76 information between sensor locations, leading to sensor clustering issues [26]. Furthermore, a
77 constant uncertainty level at all locations is assumed.

78 Once first sensor locations are selected, Papadimitriou and Lombaert [27] included the effect
79 of spatially-correlated prediction errors, thus reducing information-entropy values of
80 neighboring sensors and thus avoiding sensor clusters. Papadopoulou et al. [28] introduced a
81 methodology involving a hierarchical algorithm to examine potential locations. Mutual
82 information between sensor locations was explicitly accounted for in an objective function that
83 maximized the joint entropy. This sensor placement algorithm, combining a hierarchical
84 algorithm with joint-entropy maximization, explicitly incorporated systematic uncertainties
85 and was successfully applied to sensor-placement studies for wind-around-building sensor
86 placement [29]. The methodology has later been adapted for structural identification in order
87 to account for several sensor types and a modification was proposed to include mutual
88 information between several static load-test configurations [30].

89 As these strategies required uncertainty-distribution quantification to evaluate possible sensor
90 locations, these need engineers to provide input information. For dynamic tests, some
91 researchers preferred to use only modal information, such as mode-shape vectors, to identify
92 optimal sensor locations, mostly based on the Fisher Information Matrix (FIM) [31]. Several
93 sensor-placement objective functions for modal identification have been proposed, for
94 instance, modal kinetic energy (MKE) [32], effective independence method (EFI) [33], modal

95 assurance criterion (MAC) [34], QR-decomposition [35] and mutual information [36].
96 Relations between these objective functions have been explored [37]. Several studies compared
97 these objective functions in terms of information gain and for lattice structures [38], timber-
98 frame structures [39] and bridges [40]. These last studies concluded that the EFI method is the
99 best sensor-placement objective function for modal identification. In the EFI method, each
100 mode shape is assumed to have the same performance on structural identification. However,
101 some case studies showed that this assumption is incorrect [19], [41] because information
102 gained on model parameters and modelling errors may differ significantly between mode
103 shapes.

104 This paper contains a description of a measurement-system-design methodology for dynamic
105 load testing using expected information from natural frequencies to prioritize sensor-location
106 selections using the EFI method. Although not directly linked to sensor locations, the expected
107 performance of mode-shapes is assessed using natural-frequency predictions. Once the
108 expected information gain by each mode shape is quantified, sensor-location selections in EFI
109 are prioritized to improve the identification of useful mode shapes for model updating without
110 requiring estimation of mode-shape-vector uncertainties. Then, measurement-system-design
111 methodologies for dynamic and static load testing are combined to minimize the redundant
112 information gain between measurement systems. Finally, a full-scale case study is used to
113 evaluate this approach. The novelty of this paper lies in the prioritization of the mode-shape
114 identification in the EFI method using the expected information gain of natural frequencies.
115 Additionally, this approach proposed a combination of measurement-system-design
116 methodologies for static and dynamic load testing in order to minimize the redundancy in
117 information gain between them.

118 The study is organized as follows. Background methodologies are shown in Section 2. Section
119 3 includes a description of the methodology for measurement-system design using expected
120 information from dynamic load testing in the EDMF framework. Section 4 shows how this
121 information is combined with the static load testing for the first time in a measurement-system
122 design methodology. A full-scale case study is described in Section 5 with optimal
123 measurement-system results presented in Section 5.4 for dynamic load testing and in Section
124 5.5 for both static and dynamic load testing.

125 **2 Background**

126 In this section, background methodologies that have been developed previously are presented.
127 First, the structural-identification methodology, called error-domain model falsification, is
128 presented in Section 2.1. After this, the sensor-placement algorithm for static measurements,
129 called hierarchical algorithm, is described in Section 2.2.

130 **2.1 Structural identification – Error-domain model falsification**

131 Error-domain model falsification (EDMF) is a recently developed methodology for structural
132 identification [12]. Finite-element (FE) model predictions are compared with field
133 measurements in order to identify plausible model instances of a parameterized model class. A
134 model instance is generated by assigning a unique combination of parameter values to a model
135 class, which consists of a FE parametric model including characteristics such as material
136 properties, geometry, boundary conditions and excitations.

137 Model predictions at location i , $g_i(\Theta)$ are generated by assigning a vector of parameter values
138 Θ to the selected model class. Assuming R_i to be the real response of a structure—unknown in
139 practice—and \hat{y}_i to be the measured value at a sensor location i among n_y monitored locations,
140 model uncertainty $U_{i,g}$ and the measurement uncertainty $U_{i,y}$ are first estimated and then
141 connected to the real behavior using the following equation:

$$142 \quad g_i(\Theta) + U_{i,g} = R_i = \hat{y}_i + U_{i,\hat{y}} \quad \forall i \in \{1, \dots, n_y\} \quad (1)$$

143 Following [42], modeling and measurement uncertainties are combined in a unique source $U_{i,c}$
144 using Monte-Carlo simulations and Eq. (1) is transformed in Eq. (2). The residual r_i presents
145 the difference between the model prediction and the field measurement at a sensor location i .

$$146 \quad g_i(\Theta) - \hat{y}_i = U_{i,c} = r_i \quad (2)$$

147 In the EDMF implementation, the identification process starts with the generation of an initial
148 model set (IMS) that consists of n_Ω instances $\Omega = \{\Theta_1, \dots, \Theta_{n_\Omega}\}$. EDMF selects plausible
149 model instances by falsifying those for which residuals exceed threshold bounds given
150 combined uncertainties and a target reliability of identification, which is typically fixed at 95%
151 [12]. Model instances with residuals that do not exceed threshold bounds at each sensor
152 location are included in the candidate model set (CMS). The set of candidate models is defined
153 to be those models satisfying the inequalities in using Eq. (3).

$$154 \quad \Omega'' = \{\Theta \in \Omega'' \mid \forall i \in \{1, \dots, n_y\} u_{i,low} \leq r_i \leq u_{i,high}\} \quad (3)$$

155 where Ω'' is the candidate model set (CMS) built of model instances that have not been
 156 falsified. The thresholds, $u_{i,low}$ and $u_{i,high}$, represent lower and upper bounds expressing the
 157 shortest intervals through the probability density function (PDF) of combined uncertainties
 158 $f_{U_i}(u_i)$ at a measurement location i , including a probability of identification ϕ^{1/n_y} . The Šidák
 159 correction $1/n_y$ [43] maintains a constant level of confidence when multiple sensor
 160 measurements are compared with model-instance predictions (Eq. (4)).

$$161 \quad \forall i = 1, \dots, n_y: \phi^{1/n_y} = \int_{u_{i,low}}^{u_{i,high}} f_{U_i}(u_i) du_i \quad (4)$$

162 Since little information is usually available to describe the combined-uncertainty distribution,
 163 candidate models are set to be equally likely [44]. Thus, they are assigned an equal probability
 164 as expressed in Eq. (5).

$$165 \quad \Pr(\Theta \in \Omega'') = \frac{1}{\int_{\theta \in \Omega''} d\theta} \quad (5)$$

166 Falsified model instances are assigned a null probability (Eq. (6)).

$$167 \quad \Pr(\Theta \notin \Omega'') = 0 \quad (6)$$

168 Consequently Θ'' , is the vector of random variables describing the realistic model-parameter
 169 values of candidate model instances given field measurements. Its PDF is defined using Eq.
 170 (7).

$$171 \quad f_{\Theta''} = \begin{cases} \frac{1}{\int_{\theta \in \Omega''} d\theta}, & \text{if } \Theta \in \Omega'' \\ 0, & \text{otherwise} \end{cases} \quad (7)$$

172 If all initial model instances are falsified, the entire model class is falsified ($\Omega'' = \emptyset$). This
 173 means that no model instance is compatible with sensor measurements given combined
 174 uncertainties. This result is usually an indication of incorrect assumptions in the model-class
 175 definition [45]. This particular situation highlights an important advantage of EDMF compared
 176 with residual minimization. In such cases, EDMF results lead to a re-evaluation of assumptions
 177 and a new model class is generated, avoiding wrong parameter identification.

178 **2.2 Sensor-placement algorithm – Hierarchical algorithm**

179 Prior to measuring a structure, a sensor-placement strategy has the potential to identify optimal
 180 measurement systems when a limited knowledge of model-parameter values is available. Once
 181 the numerical model is built and the model class is selected, prediction data from a population
 182 of model instances are generated. A set of model predictions is a typical input to evaluate the
 183 expected information gain by sensor locations, such as prediction variability.

184 The information entropy, from information theory, was introduced as a sensor-placement
 185 objective function for system identification [22]. At each sensor location i , the range of model-
 186 instance predictions is divided into $N_{l,i}$ intervals, where the interval width W_i is constant. W_i
 187 is equal to the difference between upper and lower bounds of the combined source of
 188 uncertainty $U_{i,c}$. The probability that the model-instance prediction $g_{i,j}$ falls inside the j^{th}
 189 interval equals to $P(g_{i,j}) = m_{i,j}/N_{MI}$, where $m_{i,j}$ is the number of model instances falling
 190 inside the j^{th} interval, and N_{MI} is the total number of model instances. At a sensor location i , the
 191 information entropy $H(g_i)$ is evaluated using Eq. (8).

$$192 \quad H(g_i) = -\sum_{j=1}^{N_{l,i}} P(g_{i,j}) \log_2 P(g_{i,j}) \quad (8)$$

193 Papadopoulou et al. [28] proposed the joint entropy as a new sensor-placement objective
 194 function to quantify the redundancy of information gain between sensor locations. The joint
 195 entropy $H(g_{i,i+1})$ assesses the information entropy between sets of predictions, taking into
 196 account the mutual information between them. For a set of two sensors, it is defined using Eq.
 197 (9).

$$198 \quad H(g_{i,i+1}) = -\sum_{k=1}^{N_{l,i+1}} \sum_{j=1}^{N_{l,i}} P(g_{i,j}, g_{i+1,k}) \log_2 P(g_{i,j}, g_{i+1,k}) \quad (9)$$

199 where $k \in \{1, \dots, N_{l,i+1}\}$ and $N_{l,i+1}$ is the maximum number of prediction intervals at the $i+1$
 200 location and $i+1 \in \{1, \dots, n_s\}$ with the number of potential sensor locations n_s . An alternative
 201 form expresses the joint entropy to be equal to the sum of the individual information entropies
 202 of sets of predictions minus the mutual information between sensor i and $i+1$ $I(g_{i,i+1})$ as
 203 presented in Eq. (10).

$$204 \quad H(g_{i,i+1}) = H(g_i) + H(g_{i+1}) - I(g_{i,i+1}) \quad (10)$$

205 $I(g_{i,i+1})$ is unknown in practice and can be calculated only if individual information entropies
 206 and the joint entropy of sensor locations i and $i+1$ are known. The hierarchical algorithm [28]
 207 is a sequential algorithm (greedy search). Therefore, the sensor-location selection is not
 208 reevaluated in its subsequent selections. At each iteration, the hierarchical algorithm re-
 209 evaluates the joint-entropy objective function of remaining sensor locations and selects the
 210 location with the largest value. The algorithm stops when all sensor locations are selected.

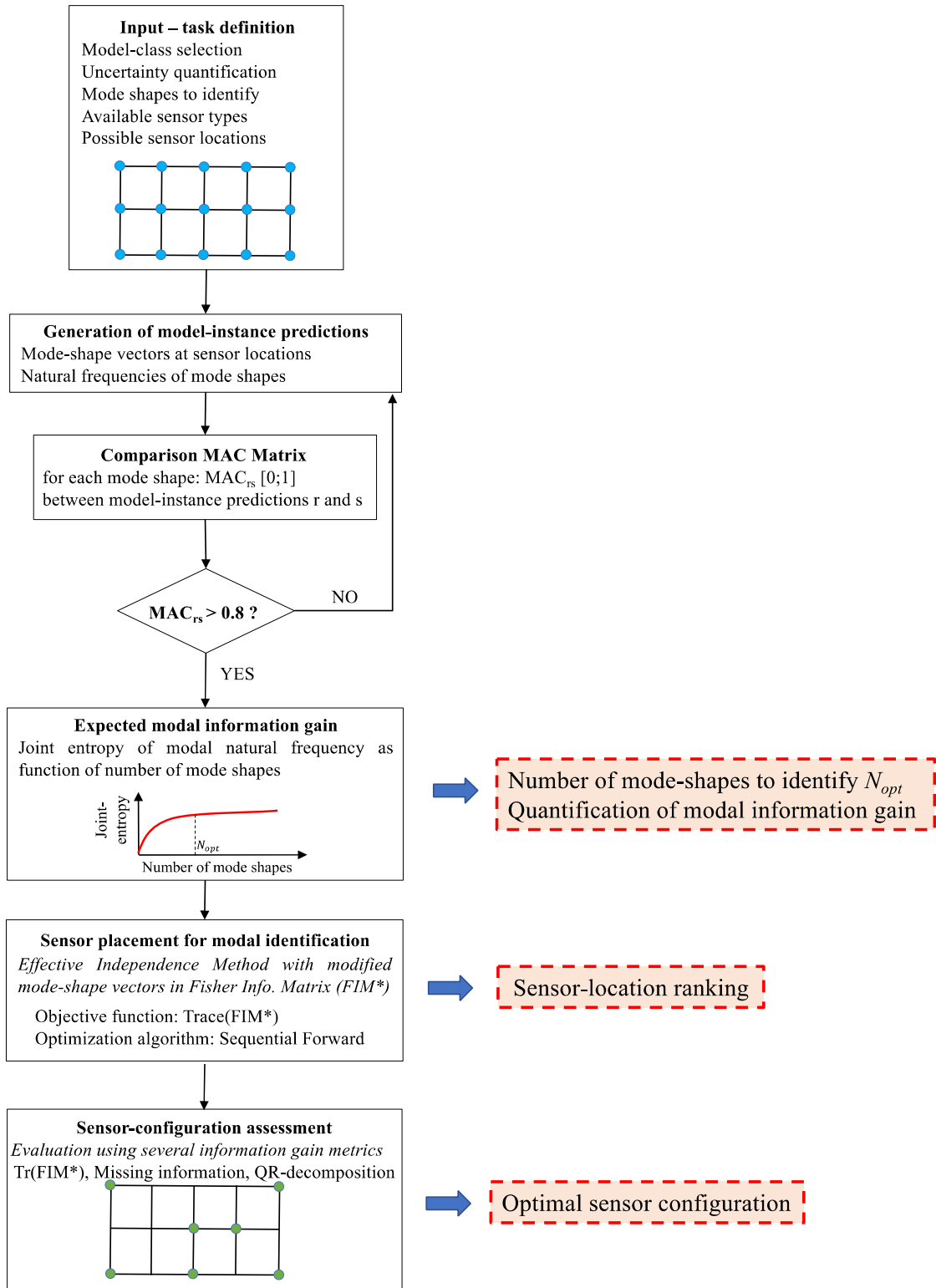
211 Bertola et al. [30] proposed a modification of the hierarchical algorithm to take into account
 212 mutual information between static load tests based on joint entropy. For a sensor location i and
 213 model predictions of two static load tests, the joint entropy evaluation is described in Eq. (11),
 214 in which $j \in \{1, \dots, N_{l,i_l}\}$ and N_{l,i_l} is the maximum number of intervals at the location i
 215 associated with a load test l , $k \in \{1, \dots, N_{l,i_{l+1}}\}$ and $N_{l,i_{l+1}}$ is the maximum number of intervals

216 at the location i associated with another load test $l + 1 \in \{1, \dots, n_{LT}\}$ with the number of
 217 potential load tests n_{LT} . The hierarchical algorithm is able to evaluate the expected information
 218 gained by a measurement system, consisting of a sensor configuration and a set of load tests.

$$219 \quad H(g_{i_l, i_{l+1}}) = - \sum_{k=1}^{N_{l, i_{l+1}}} \sum_{j=1}^{N_{l, i_l}} P(g_{i_l, j}, g_{i_{l+1}, k}) \log_2 P(g_{i_l, j}, g_{i_{l+1}, k}) \quad (11)$$

220 **3 Measurement-system-design methodology for dynamic load** 221 **testing**

222 In this section, the methodology for sensor-placement for dynamic load testing is developed.
 223 Figure 1 presents the steps of the methodology. The methodology is divided into four phases.
 224 Each phase is presented in a subsection below. First, the task of measurement-system design is
 225 defined such as the model-class, selection, the generation of model-instance predictions or the
 226 engineering decisions (Section 3.1). The expected information gain of each natural frequency
 227 is assessed to determine the optimal number of natural frequencies for falsification (Section
 228 3.2). This phase involves the identification of the most useful modes for structural
 229 identification. However, natural-frequency predictions are not directly linked to a sensor
 230 location. Therefore, the next phase involves using a sensor-placement strategy to rank sensor
 231 locations (Section 3.3). To take advantage of the information gained in the second phase, the
 232 sensor-placement strategy prioritizes locations that help identify useful modes for falsification.
 233 Finally, Section 3.4 presents information-gain metrics that are used to define the optimal sensor
 234 configuration. These metrics help evaluate the number of sensors required to identify
 235 accurately natural frequencies of the most useful modes.



236

237 Figure 1 Measurement-system-design methodology for dynamic load tests.

238 3.1 Task definition

239 Before assessing sensor locations, several steps are required. This section describes important
240 steps in the definition of the sensor-placement task definition. The first step is to define the
241 input information. The finite-element model of the structure is built to obtain reliable
242 quantitative predictions of measurable variables such as mode-shape vectors at each possible
243 sensor location. Engineers choose sensor types and the number of sensors available in the study
244 and select possible locations. The number of mode shapes is estimated using engineering
245 judgement, signal-to-noise ratio evaluation and the maximum number of available sensors.

246 Model parameters, which have the highest impact on predictions, are selected using a
247 sensitivity analysis. Due to geometrical and mathematical simplifications of the numerical
248 model, significant non-parametric uncertainties are always involved. Model uncertainties must
249 be estimated because the expected information-gain assessment of natural frequencies is
250 influenced by them (Section 3.2).

251 Multiple model instances are generated using a sampling technique to obtain a discrete
252 population of model-parameter values within plausible ranges. For each model instance, typical
253 outputs are mode-shape vectors at each sensor location and a natural frequency for each mode
254 shape. Then, for each mode shape, a population of model predictions is generated. Model
255 instances constitute the initial model set, which is the dataset used in the sensor-placement
256 strategy to rank possible sensor locations (Section 3.2).

257 The last step of the task-definition phase involves evaluating if model-instance predictions for
258 each mode shape are compatible. By compatible, it is assumed that mode-shape vectors φ
259 between two model instances r and s have a modal assurance criterion (MAC) larger or equal
260 to 0.8 (Eq. (12)) [34].

$$261 \quad \text{MAC}_{rs} = \frac{(\varphi_r^T \varphi_s)^2}{(\varphi_r^T \varphi_r)(\varphi_s^T \varphi_s)} \geq 0.8 \quad (12)$$

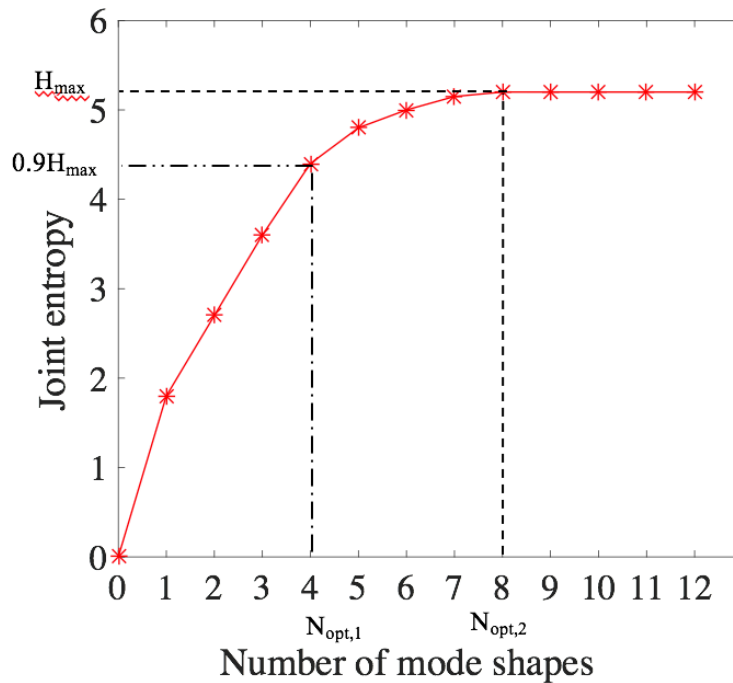
262 This condition ensures that model instances are comparable in terms of predictions. While this
263 condition is typically fulfilled [46], it may happen that the appearance order of mode-shapes is
264 inversed since model-parameter values influence natural-frequency values.

265 3.2 Assessment of natural-frequency expected information gain

266 The expected information gained by natural frequencies can be assessed using the joint entropy
267 in a similar way to Section 2.2 for static measurements. In this case, the output variable g_m is
268 the natural-frequency predictions of the mode m and is not related to sensor locations.

269 Following Eq. (8) to (10), the expected information gain of sets of natural-frequency
 270 predictions can be quantified.

271 The joint-entropy assessment provides two features to the sensor-placement strategy (Section
 272 3.3). First, a subset of useful mode shapes can be selected by removing mode shapes that do
 273 not significantly contribute to the joint entropy. Then, an estimation of the contribution of each
 274 mode shape can be assessed through the joint-entropy increase when the mode shape is added
 275 to the subset. Figure 2 presents a sketch of typical joint-entropy evaluation as function of
 276 number of mode shapes. Natural frequencies are ranked by expected information gain. The
 277 joint entropy increases until it reaches a maximum value, corresponding to the maximum
 278 expected information gain by dynamic load testing. In this example, mode shapes selected after
 279 the 8th position do not provide useful additional information as the joint entropy does not
 280 increase significantly. Therefore, only $N_{opt,2}$ mode shapes only are identified. Similarly, another
 281 asset manager may decide to only use $N_{opt,1}$ modes that correspond to 90% of the maximum
 282 joint entropy to reduce the number of sensors to install on the bridge and therefore reduce the
 283 cost of the monitoring. The information provided by each mode shape can be quantified in a
 284 similar way by assessing the increase of joint entropy when the mode shape m is added to the
 285 system.



286
 287 Figure 2 Sketch of the joint entropy of natural frequencies as function of number of mode
 288 shapes.

289 3.3 Modified effective independence method for modal identification

290 3.3.1 Equations of motion

291 Any sensor-placement methodology for modal identification starts with the equation of motion
292 for a linear structural system. Using n degree of freedom (DOFs) is formulated in Eq. (13) as
293 function of time t .

$$294 \quad \mathbf{M}\ddot{\mathbf{u}}(t) + \mathbf{C}\dot{\mathbf{u}}(t) + \mathbf{K}\mathbf{u}(t) = \mathbf{F}(t) \quad (13)$$

295 where $\mathbf{u}(t) \in [n \times 1]$ is the vector of displacement responses, \mathbf{M} , \mathbf{C} and $\mathbf{K} \in [n \times n]$ are the
296 mass, damping and stiffness matrices respectively, and $\mathbf{F}(t) \in [n \times 1]$ is the vector of
297 excitations. A typical assumption is that the structural system is characterized by proportional
298 damping [39]. Eq. (14) can be rewritten in Eq. (15).

$$299 \quad \ddot{\boldsymbol{\xi}}(t) + \bar{\mathbf{C}}\dot{\boldsymbol{\xi}}(t) + \boldsymbol{\Lambda}\boldsymbol{\xi}(t) = \boldsymbol{\Phi}^T \mathbf{F}(t) \quad (14)$$

300 With

$$301 \quad \boldsymbol{\Lambda} = \text{diag}\{\omega_1^2, \dots, \omega_n^2\} \in [n \times n] \quad (15)$$

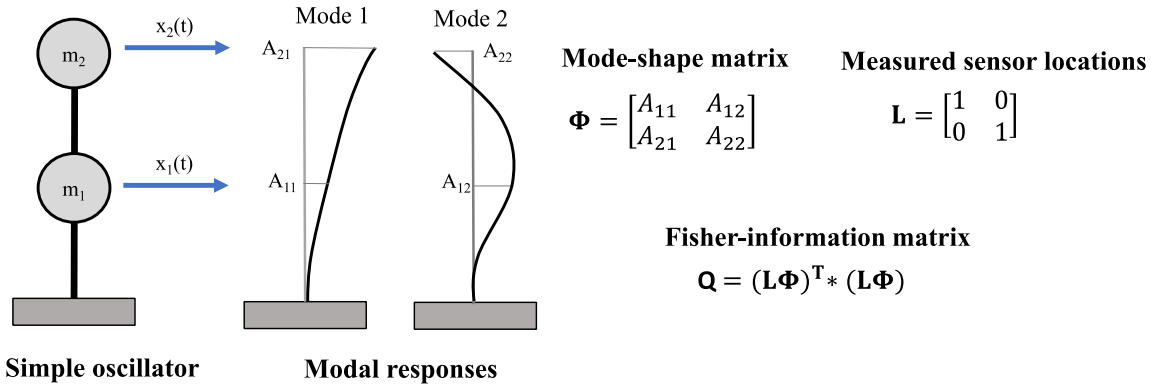
$$302 \quad \bar{\mathbf{C}} = \text{diag}\{2\zeta\omega_1, \dots, 2\zeta\omega_n\} \in [n \times n] \quad (16)$$

303 where $\boldsymbol{\xi}(t) \in [n \times 1]$ is the vector of modal coordinates, ω_j is the j^{th} natural frequency, ζ_j
304 designates the j^{th} modal damping ratio, $\boldsymbol{\Phi} \in [n \times n]$ is the matrix of mode shapes, $\boldsymbol{\Lambda}$ is the
305 matrix of natural frequencies and $\bar{\mathbf{C}}$ is the modified damping matrix. Moreover, the matrix of
306 mode shapes may be $\boldsymbol{\Phi} \in [m \times m]$ if a subset of $m < n$ mode shapes are involved. The
307 structural response in the original coordinate space is thus: $\mathbf{u}(t) = \boldsymbol{\Phi}\boldsymbol{\xi}(t)$.

308 3.3.2 Traditional effective independence method

309 The aim of a sensor-placement methodology for modal identification is to build the best
310 estimate of the structural response $\boldsymbol{\xi}(t)$. This implies the minimization of the covariance matrix
311 of the estimated errors. Following [31], maximizing the Fisher Information Matrix (FIM)
312 would lead to the minimization of the covariance matrix and thus to the best estimate of $\boldsymbol{\xi}(t)$.
313 The FIM $\mathbf{Q} \in [n \times n]$ is computed using Eq. (17) in which $\mathbf{L} \in [n \times n]$ is a Boolean matrix
314 that maps the sensor locations to the DOFs. Figure 3 presents FIM calculation. The case of a
315 simple oscillator with two possible sensor locations and two mode shapes is used for simplicity
316 purposes. A numerical example is added to show the FIM calculation in Figure 3.

$$317 \quad \mathbf{Q} = (\mathbf{L}\boldsymbol{\Phi})^T(\mathbf{L}\boldsymbol{\Phi}) \quad (17)$$



Numerical application $\Phi = \begin{bmatrix} 0.3 & -0.4 \\ 1 & 0.5 \end{bmatrix}$ $\mathbf{Q} = \begin{bmatrix} 1.09 & 0.38 \\ 0.38 & 0.41 \end{bmatrix}$ $\text{tr}(\mathbf{Q}) = 1.5$
 $\det(\mathbf{Q}) = 0.303$

318

319 Figure 3 Fisher-information-matrix (FIM) calculation for the case of a simple oscillator with
 320 two sensor locations and two mode shapes.

321 The traditional effective independence method (EFI), introduced by Kammer [33], evaluates
 322 possible sensor locations through both the linear independence between target mode shapes
 323 and the intensity of measured value. Using a sequential sensor placement with Forward (FSSP)
 324 or Backward (BSSP) greedy strategies for sensor placement, sensor configurations are
 325 evaluated through FIM properties such as the trace, or the determinant. An important
 326 characteristic of the FIM is that the matrix is singular in the case of the number of sensors being
 327 lower than the number of mode shapes. In such situations, the FIM determinant cannot be used
 328 as an objective function for sensor placement.

329 An important assumption of the traditional EFI method is that each mode is equally useful for
 330 the model-parameter identification. If this assumption is correct, each natural frequency should
 331 have the same influence on the model falsification. However, several case studies show that
 332 this assumption is incorrect [19], [41] as the variability of frequency predictions is significantly
 333 influenced by the model parameters to identify. The next section describes how the EFI method
 334 is modified to take into account the importance of mode shapes for falsification in the sensor-
 335 location ranking.

336 3.3.3 Modified effective independence method

337 A solution proposed by others, for example [16], [23], involves using parameter uncertainties
 338 and a Bayesian framework to identify sensor locations that will reduce the posterior distribution
 339 of model parameters. This proposal necessitates evaluations of the combined uncertainty $U_{i,c}$

340 for mode-shape vectors. Such recommendations are currently missing in the literature due to
 341 difficulties associated with uncertainties of mode shapes.

342 The strategy proposed in this study is to use the information provided during the modal
 343 expected information gain (Section 3.2) to weight mode-shape vectors according to the
 344 expected information that they provide. Each column in the mode-shape matrix Φ ,
 345 corresponding to mode-shape m , is therefore adjusted according to its contribution in the joint-
 346 entropy evaluation. The contribution is estimated as the difference of joint entropy ΔH_m when
 347 the mode shape m is added to the system. Therefore, an importance Boolean matrix $\mathbf{I}_{mp} \in$
 348 $[n \times n]$ is added to the FIM calculation, in which $I_{mp_{m,m}} = \Delta H_m$. \mathbf{I}_{mp} will thus prioritize
 349 sensor locations in the EFI method.

350 The assessment of the modal-frequency joint entropy does not involve sensor locations. In
 351 order to accurately measure natural frequencies, possible sensor locations must be able to
 352 reconstruct accurately the mode shapes. A simple example is the following. If a local mode
 353 shape is involved and possible sensor locations are not present in the specific area, it is not
 354 possible to identify this mode shape and it should not be involved in the sensor placement.
 355 Therefore, a second identifiability Boolean matrix $\mathbf{I}_{nt} \in [n \times n]$ is added to the FIM
 356 calculation where the identifiability of each mode shape is a ratio between the maximum
 357 deformation of the mode shape m at possible sensor locations, $A_{L,max}$, divided by the global
 358 maximum deformation of this mode shape, A_{max} . As both amplitude values are influenced by
 359 model predictions, mean predictions are used. For a mode shape m , the identifiability matrix
 360 is calculated as $I_{nt_{m,m}} = \text{mean}(A_{L,max_m}(\boldsymbol{\theta})) / \text{mean}(A_{max_m}(\boldsymbol{\theta}))$ and is bounded between 0
 361 and 1.

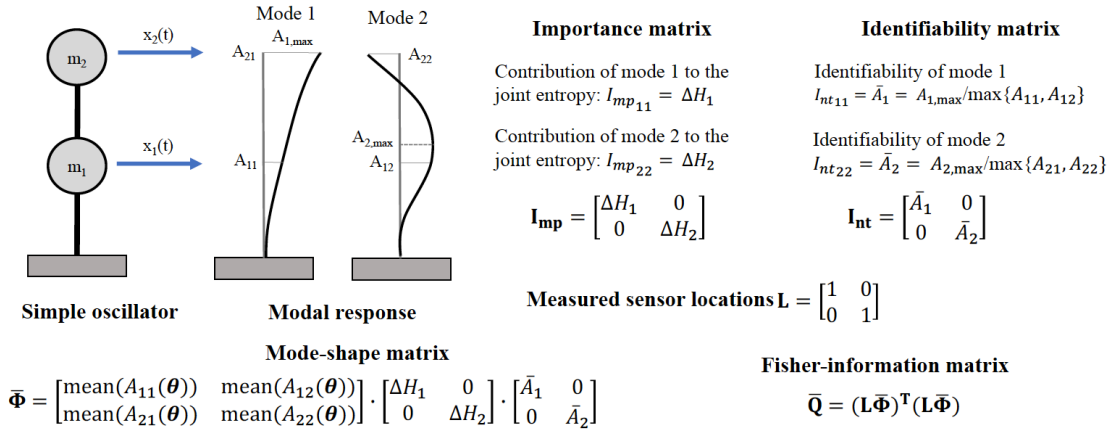
362 Mode-shape vectors are influenced by modal parameters to identify. As mode-shape vectors of
 363 model instances are comparable (Section 3.1), the average value of mode-shape vectors is taken
 364 as an approximation. The modified mode-shape vector for modal identification is presented in
 365 Eq. (18).

$$366 \quad \bar{\Phi} = \Phi \left(\text{mean}(g_i(\boldsymbol{\theta})) \right) \cdot \mathbf{I}_{mp} \mathbf{I}_{nt} \quad (18)$$

367 The modified FIM can be calculated using Eq. (19). Figure 4 presents steps of the calculation
 368 of the modified FIM for a simple oscillator, where the importance-factor and identifiability
 369 matrices are used to weigh the mode-shape matrix Φ . When compared with the traditional FIM
 370 calculation (Figure 3), the modified FIM calculation involves the expected performance in
 371 terms of information gain of mode-shape vectors. In the numerical example, the value of joint
 372 entropy of the second mode shape is small, showing a smaller importance of this mode for

373 model parameter identification. By including the importance matrix, FIM properties such as
 374 the trace are then reduced; i.e. $\text{tr}(\bar{\mathbf{Q}})$ equals to 1.18, while $\text{tr}(\mathbf{Q})$ equals to 1.5 in Figure 3 when
 375 no important matrices are taken into account. This result shows that sensor locations useful to
 376 identify the second mode shape such as A_1 will not have priority since this mode does not
 377 contribute significantly to the information gain.

378
$$\bar{\mathbf{Q}} = (\mathbf{L}\bar{\Phi})^T(\mathbf{L}\bar{\Phi}) \quad (19)$$



379 **Numerical application** $\Phi = \begin{bmatrix} 0.3 & -0.4 \\ 1 & 0.5 \end{bmatrix}$ $\mathbf{I}_{mp} = \begin{bmatrix} 1 & 0 \\ 0 & 0.5 \end{bmatrix}$ $\mathbf{I}_{nt} = \begin{bmatrix} 1 & 0 \\ 0 & 0.95 \end{bmatrix}$ $\bar{\mathbf{Q}} = \begin{bmatrix} 1.09 & 0.38 \\ 0.18 & 0.09 \end{bmatrix}$ $\text{tr}(\bar{\mathbf{Q}}) = 1.18$
 380 $\det(\bar{\mathbf{Q}}) = 0.068$

380 Figure 4 Modified Fisher-information matrix (FIM) calculation using the simple-oscillator case
 381 and joint-entropy evaluations of mode shapes.

382 Since the number of sensors involved in this study is low, the trace of $\bar{\mathbf{Q}}$ is used as an objective
 383 function for sensor placement. The determinant of $\bar{\mathbf{Q}}$ is equal to zero in cases where number of
 384 sensors is lower than the number of mode shapes needing identification. Additionally, a
 385 sequential forward (FSSP) is used as an optimization strategy, following [21].

386 3.4 Information-gain metrics for modal identification

387 Once the objective function for the sensor placement and the optimization algorithm are
 388 selected, sensor locations are ranked. To define the optimal sensor configuration, quantification
 389 of information gain metrics must be carried out. In this section, a strategy involving two metrics
 390 is proposed to evaluate sensor configurations in terms of their ability to reconstruct mode
 391 shapes. The missing information and QR decomposition are shown to be complementary.
 392 Information-gain metrics and are presented in Sections 3.4.1 and 3.4.2.

393 3.4.1 Missing information

394 The first metric, called missing information MI , represents a comparison the information
 395 provided by a sensor configuration k with the maximum information that can be gained if the
 396 sensor configuration includes all possible sensor locations. The metric, inspired by the mutual-
 397 information objective function introduced by Stephan [36], compares the FIM of a sensor
 398 configuration k $\bar{\mathbf{Q}}_k$ with the FIM of the sensor configuration K including all possible sensor
 399 locations $\bar{\mathbf{Q}}_K$. This metric is calculated using Eq. (20). When MI tends to zero, the maximum
 400 possible information gain, represented by sensor configuration K , is obtained by the sensor
 401 configuration k . A non-zero value of MI means that adding sensors to the sensor configuration
 402 will provide more information. Therefore, the information-gain metric MI must be minimized.

$$403 \quad MI(k) = \frac{\|\bar{\mathbf{Q}}_k - \bar{\mathbf{Q}}_K\|}{\|\bar{\mathbf{Q}}_k + \bar{\mathbf{Q}}_K\|} \in [0,1] \quad (20)$$

404 3.4.2 QR decomposition

405 The second information-gain metric, called QR decomposition (QRD), provides an evaluation
 406 of the linear independence of mode-shape vectors. The term QR comes from the decomposition
 407 of a matrix A into the product $A = QR$, where Q is an orthogonal matrix and R an upper
 408 diagonal matrix. Introduced by Schedlinski and Link [35], the metric uses rows of the mode-
 409 shape matrix Φ instead of columns used in EFI. For N_m modes, the linear independence of each
 410 pair of rows $\langle \psi_n, \psi_m \rangle \in [1 \times n]$ is evaluated using Eq. (21). The matrix $\mathbf{LI} \in [N_m \times N_m]$ is
 411 therefore a triangular matrix, in which the $LI_{n,m}$ is equal to zero if mode-shape vectors n and m
 412 are perfectly orthogonal and equal to one if they are linearly dependent. The information-gain
 413 metric QRD sums $LI_{n,m}$ elements of a sensor configuration k (Eq. (22)). When QRD tends to
 414 zero, all pairs of mode-shape vectors are orthogonal and therefore, each mode shape is
 415 distinguished. Non-zero values signify that some mode shapes cannot be differentiated and,
 416 therefore, adding sensors to the configuration can convey additional information in terms of
 417 the linear independence of mode-shape vectors. Therefore, the information-gain metric QRD
 418 must be minimized.

$$419 \quad LI_{n,m} = \frac{(\psi_n^T \psi_m)^2}{(\psi_n^T \psi_n)(\psi_m^T \psi_m)} \quad (21)$$

$$420 \quad QRD_k = \sum_i^{N_m} \sum_j^{N_m} LI_{i,j} \quad (22)$$

421 **4 Methodology for measurement system design combining** 422 **information from static and dynamic testing**

423 In order to monitor a structure, engineers may perform dynamic and static tests. Both tests
424 differ in terms of the nature of the data collected (time-series data for dynamic tests and single
425 measurements at sensor locations for static tests) and numerical models for predictions.
426 Nevertheless, information collected during monitoring are often used to update the same
427 parameter values of the numerical behavior model. Therefore, sensor-placement
428 methodologies should be combined in order to avoid redundant information gain between
429 dynamic and static load testing. In this section, a methodology is proposed to design
430 measurement systems that are intended for including static and dynamic load testing (Figure
431 5).

432 The strategy begins with the building of a numerical model of the structure. Primary parameters
433 to identify are chosen and non-parametric uncertainties are estimated. The engineer then selects
434 available sensor types and sensor locations for both static and dynamic tests. Additionally,
435 possible static load tests are designed.

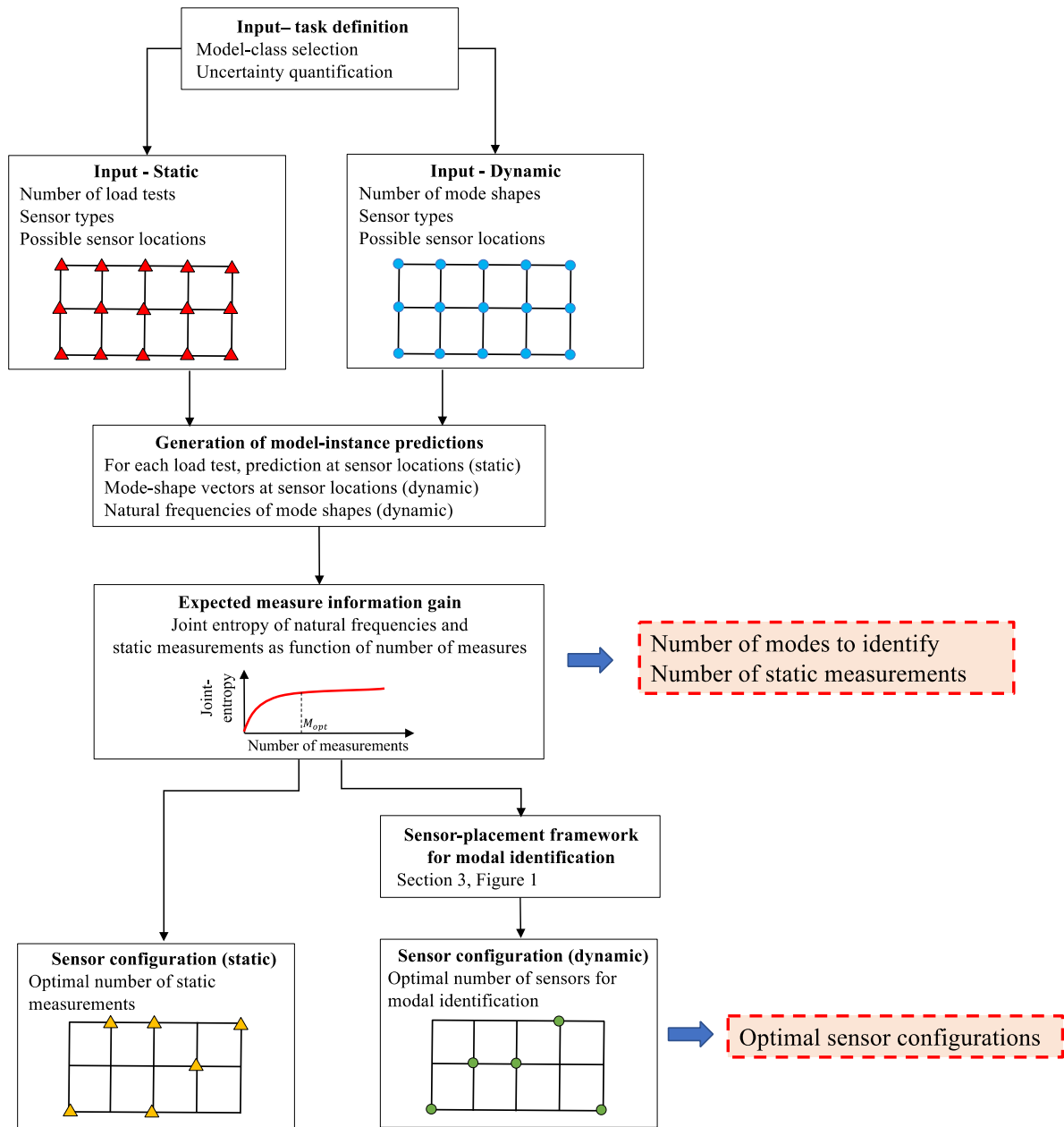
436 Once engineering decisions are made, model instances are generated for both static predictions,
437 such as deflection at static-sensor locations for each static load test and dynamic predictions
438 such as natural frequencies and mode-shape vectors at dynamic-sensor locations. In order to
439 compare the information entropy of each type of measurement (i.e. static measurements and
440 natural frequencies), an important limitation of the methodology is that both tests must use the
441 same model class; the same dataset in terms of model-parameter values must be compared. In
442 this situation, predictions of static measurements and natural frequencies of dynamic tests can
443 be compared using the joint-entropy metric for information gain as presented in Section 3.2.
444 Mathematical details are presented in Section 4.1.

445 The optimal sensor configuration for the static test is explicitly obtained from the joint-entropy
446 assessment, following the methodology presented in Section 2.2. Concerning the optimal
447 sensor configuration for dynamic test, Sections 3.3 and 3.4 must still be performed. However,
448 the importance matrix \mathbf{I}_{mp} is adapted from the joint-entropy assessment using both static and
449 dynamic tests, while the identifiability matrix \mathbf{I}_{nt} remains unchanged as the same possible
450 dynamic-sensor locations are used. When these steps are completed, the optimal sensor
451 configuration for dynamic testing is found.

452 **4.1 Joint-entropy assessment**

453 The information provided by static measurements and natural frequencies, which in this section
454 are called measurements, is assessed the same way as in Sections 2.2 and 3.2. In this situation,
455 the output variable, g , is either the natural-frequency predictions of the mode m or the static
456 predictions at sensor location, i , such as deflection. Following Eq. (8) to (10), the expected
457 information gain of sets of natural-frequency predictions can be quantified.

458 The expected information gain provided by each measurement can be quantified using the
459 joint-entropy assessment. This result makes two contributions to the sensor-placement
460 strategies. First, useful static measurements are directly found and thus, the optimal static
461 sensor configuration is revealed. Secondly, useful mode shapes and their information-gain
462 quantification are provided for the sensor-placement strategy for modal identification as
463 presented in Section 3, where the importance factor of each mode shape is evaluated using the
464 joint-entropy assessment combining static and dynamic testing.



465

466

467

Figure 5 Measurement-system-design methodology combining information from static and dynamic load testing.

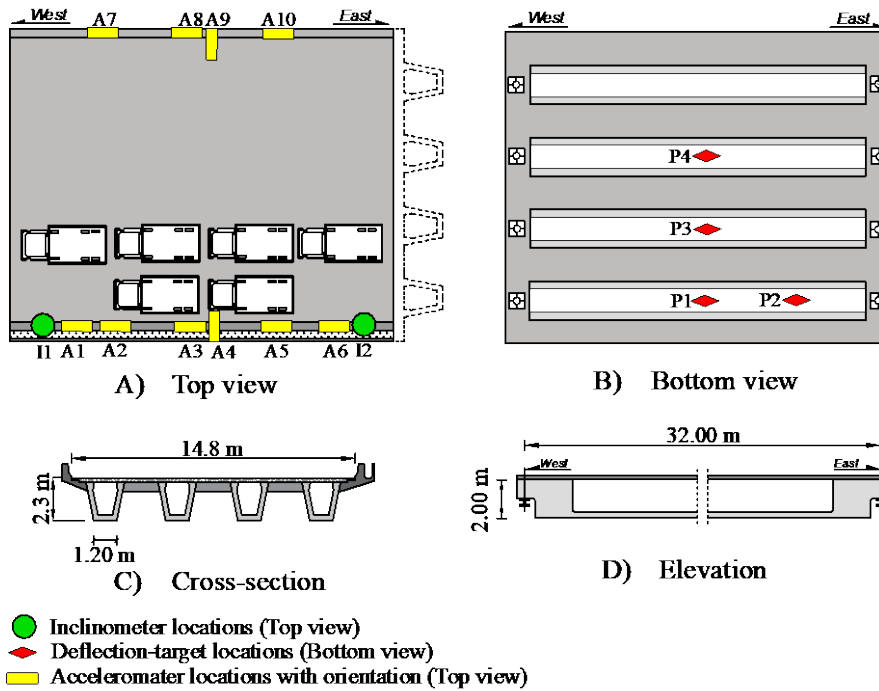
468 **5 Case study**

469 Engineering details of the full-scale case study are presented in Section 5.1. Sections 5.2 and
470 5.3 show the selection of the model class and of the mode shapes. Optimal measurement
471 systems are presented in Section 5.4 for dynamic load testing and in Section 5.5 for both static
472 and dynamic load testing.

473 **5.1 Bridge presentation**

474 The full-scale case study is a 32-year-old bridge in Singapore. The pre-stressed pre-cast
475 concrete structure has four beams carrying three unidirectional traffic lanes over a simply-
476 supported span of 32 meters. Principal characteristics of the bridge are presented in Figure 6
477 including the static load test and possible sensor locations. Sensor locations were chosen based
478 on engineering judgement and signal-to-noise-ratio estimation. As this sensor configuration
479 could not be modified, this study focuses on evaluating the performance of selected sensor
480 locations.

481 For the static load test, sensors included two inclinometers (I_i) on the south parapet and four
482 deflection targets (P_i) on the girders (Figure 6: A; B). A laser tracker was positioned on the
483 road below the bridge and was used to measure deflections at target locations. To monitor the
484 dynamic behavior of the bridge, 10 accelerometers were installed on parapets (Figure 6: A).
485 Eight free-vibration tests were carried out on the bridge using a single truck at several speeds.
486 Additionally, ambient-vibration measurements were recorded during 15 minutes when no truck
487 was running. Additional information concerning the monitoring is presented in [47].



488
 489 Figure 6 Bridge geometry showing sensor locations and the static load test: A) Top view; B)
 490 Bottom view; C) Cross-section; D) Elevation.

491 5.2 Model-class selection

492 The selection of the model class is an important step of structural identification, where primary
 493 parameters are chosen and uncertainties are estimated. Four model parameters have the largest
 494 influence on model predictions: the Young modulus of the concrete E_{con} ; the density of the
 495 concrete ρ_{con} ; the rotational stiffness of the bearing devices K_{rot} ; and the vertical stiffness of the
 496 bearing devices K_{lon} . Plausible ranges of model-parameter values are estimated using
 497 engineering judgement and are presented in Table 1. Non-structural elements, such as the
 498 asphalt pavement, are included in the numerical model to reduce model-simplification
 499 uncertainties.

500 Table 1 Model-parameter initial ranges for structural identification.

Primary parameters	Symbol	Initial ranges
Equivalent Young's modulus of concrete beams and deck (GPa)	E_{con}	[20 – 42]
Concrete equivalent density (kg/m^3)	ρ_{con}	[1800 – 3000]
Rotational stiffness of bearing devices ($\log(\text{Nmm}/\text{rad})$)	K_{rot}	[9 – 13]
Vertical stiffness of bearing devices ($\log(\text{N}/\text{mm})$)	K_{lon}	[8 – 11]

501

502 Table 2 presents the upper and lower bounds of model-class uncertainties and measurement
 503 uncertainties. Measurement uncertainties include the sensor precision based on manufacturing
 504 specifications and site conditions, the measurement repeatability that is usually estimated by
 505 conducting multiple series of tests on site, and the influence of the sensor installation. Model-
 506 class uncertainties are estimated using engineering judgement, technical literature, and local
 507 knowledge. Since a limited number of parameters can be sampled to generate the initial model
 508 set, an additional uncertainty source estimated using stochastic simulation is included.

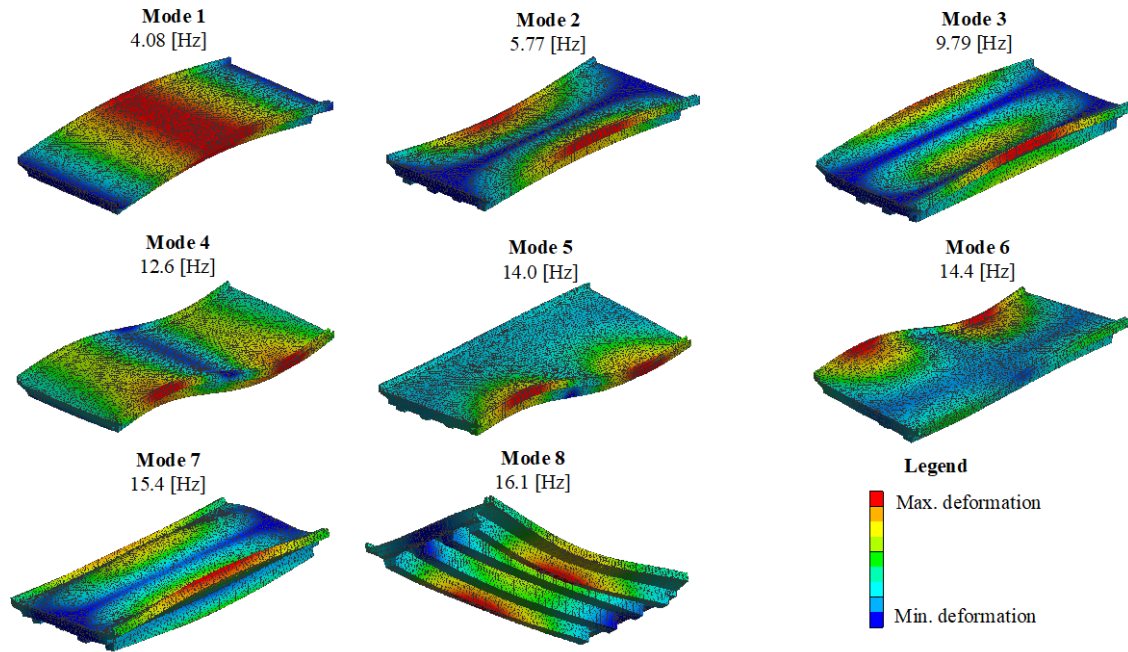
509 Table 2 Estimations of model and measurement uncertainties.

Uncertainty source	Displacements – (P)		Rotations – (I)		Accelerometer – (S)	
	Min	Max	Min	Max	Min	Max
Model simplifications (%)	-5	13	-5	13	-8	5
Mesh refinement (%)	-1	1	-1	1	0	2
Additional uncertainty (%)	-1	1	-1	1	-1	1
Sensor precision	-0.05 mm	0.05 mm	-1 μ rad	1 μ rad	-0.1 Hz	0.1 Hz
Repeatability	-0.15 mm	0.15 mm	-4 μ rad	4 μ rad	-0.05 Hz	0.05 Hz
Sensor installation (%)	-	-	-5	5	-2	2

510

511 5.3 Mode-shape selection

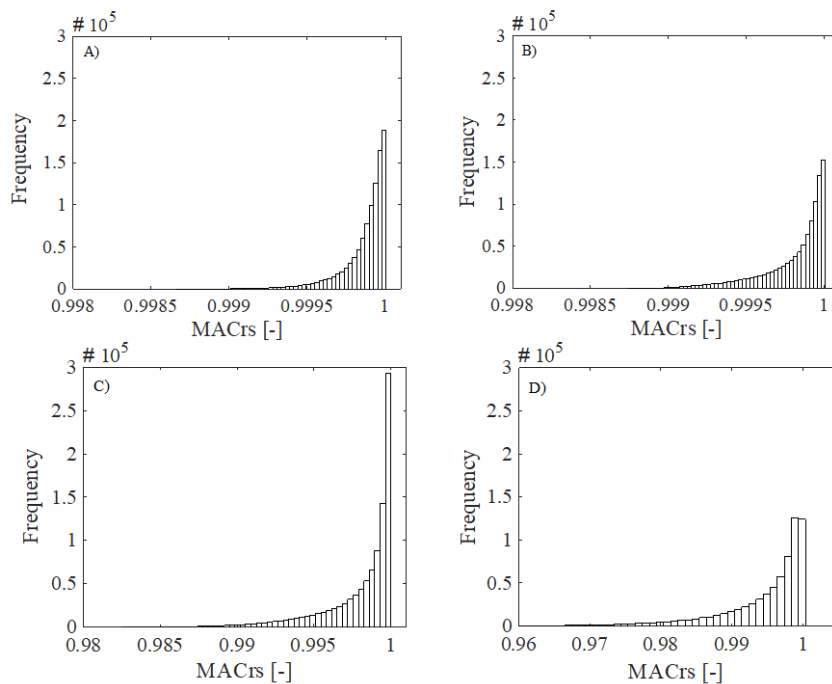
512 The eight first mode shapes are selected for the modal identification based on global behavior
 513 and low natural-frequency values (Figure 7). Within this 4-parameter space, 1,000 initial model
 514 instances were generated using Latin Hypercube Sampling (LHS). Model-instance predictions
 515 consist of deflection and inclination measurements for the static load testing (Figure 6) and the
 516 natural frequency of each mode shape and the mode-shape vector at each possible
 517 accelerometer location.



518

519 Figure 7 Visualization of mode shapes with their respective mean natural frequency.

520 Before designing a measurement system, the comparability of model-instance predictions for
 521 each mode shape must be checked, following Section 3.1. Figure 8 presents the distribution of
 522 MAC values between model instances r and s for respectively, mode 1 (Figure 8A), mode 2
 523 (Figure 8B), mode 4 (Figure 8C) and mode 7 (Figure 8D). Horizontal axes are scaled between
 524 the minimum values of MAC_{rs} and 1. Following Eq. (12), MAC_{rs} must be above 0.8. For all
 525 mode shapes, this condition is satisfied and thus, the model-instance generation is validated.
 526 Additionally, as the values in the distribution are almost equal to 1, this means that model
 527 instances do not significantly influence the shape of the mode but more the amplitude.
 528 Therefore, the implicit assumption of Eq. (18), taking the mean value as a good approximation
 529 of the modal behavior, is reasonable.



531
 532 Figure 8 Modal assurance criterion (MAC) for model instances i and j for specific modes. A)
 533 mode 1; B) mode 2; C) mode 4; D) mode 7.

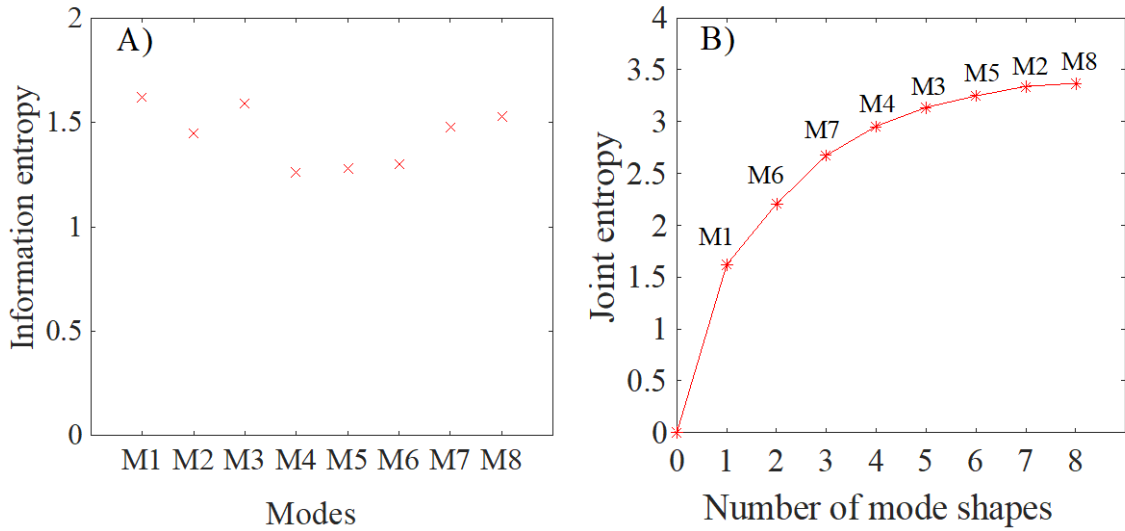
534 5.4 Measurement-system design for dynamic load testing

535 In this subsection, the methodology presented in Section 3 for sensor placement for dynamic
 536 load testing is applied to the Singapore case study. First, the mode-shape joint entropy is
 537 assessed in Section 5.4.1. Then, the addition of the importance matrix and identifiability matrix
 538 to the FIM calculation is compared with the traditional approach in Section 5.4.2.

539 5.4.1 Joint-entropy assessment of natural-frequency expected information gain

540 This section contains an assessment of the expected information gain of natural-frequency
 541 predictions of the eight mode shapes presented in Section 5.3. The information entropy is
 542 presented in Figure 9A, showing the expected information gain of each mode shape
 543 independently. Each mode shape displays similar performance with information-entropy
 544 values between 1.28 and 1.62, meaning they have close expected performance for structural
 545 identification as the information entropy is bounded between 0 and 10. Mode shape 1 shows a
 546 slightly larger information entropy and will be selected as the best mode shape in the first
 547 iteration of the hierarchical algorithm (Section 2.2). Figure 9B presents the joint entropy
 548 evaluation of mode-shape sets as function of number of mode shapes. The joint entropy
 549 increases significantly with first mode shapes (M1, M6, M7, M4) added to the set and then, the

550 expected information gain is limited. Although mode 1 is selected in the first position,
 551 subsequent mode-shape selections do not follow the mode-shape order. Higher mode shapes,
 552 such as modes 6 and 7, have more additional information when mode 1 is already involved. As
 553 each mode shape increases the joint entropy evaluation, each mode shape provides unique
 554 information. Therefore, all mode shapes will be involved in the measurement-system design
 555 for modal identification (Section 5.4.2).



556
 557 Figure 9 A) Information entropy (Eq. (8)) of natural frequencies of mode shapes; B) Joint
 558 entropy (Eq. (9)) of the natural frequency of mode-shape set as function of the number of
 559 modes.

560 5.4.2 Sensor-placement for modal identification

561 In this section, the sensor placement based on modal identification is performed. In order to
 562 justify the proposition to modify the mode-shape vectors using importance-factor and
 563 identifiability matrices (Eq. (18)), several importance-factor scenarios will be compared. Table
 564 3 presents five scenarios that are compared as well as equal importance factors in terms of
 565 objective-function evaluation (Section 5.4.2.1), sensor-location ranking (Section 5.4.2.2) and
 566 information-gain metrics (Section 5.4.2.3). The first importance-factor scenario $\mathbf{I}_{mp,1}$ includes
 567 only an importance matrix based on the information-entropy results (Figure 9A), as introduced
 568 in Section 3.3.3. The second importance-factor scenario $\mathbf{I}_{mp,2}$ uses only the joint entropy as
 569 importance factors based on results of Figure 9B. As redundancy of information gain inevitably
 570 occurs, the mode-1 result is corrected using Eq. (10) and the information entropy of the first
 571 two mode shapes is selected. The third importance-factor scenario is the identifiability matrix
 572 \mathbf{I}_{nt} , as introduced in Section 3.3.3. The fourth and fifth importance-factor scenarios take into

573 account both importance-factor and identifiability matrices using the information entropy
 574 $\mathbf{I}_{\text{mp},1} * \mathbf{I}_{\text{nt}}$ and the joint entropy $\mathbf{I}_{\text{mp},2} * \mathbf{I}_{\text{nt}}$ respectively.

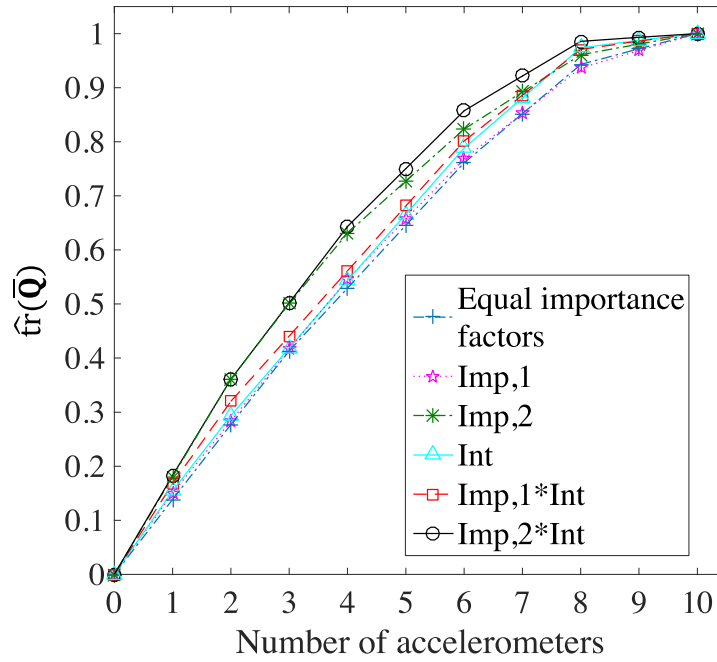
575 Table 3 Mode-shape-vector importance factors.

Mode-shape-vector importance-factor scenarios					
Modes	Information entropy (IE) [-] $\mathbf{I}_{\text{mp},1}$	Joint entropy gain (JE) [-] $\mathbf{I}_{\text{mp},2}$	Identifiability (ID) [-] \mathbf{I}_{nt}	IE and ID [-] $\mathbf{I}_{\text{mp},1} * \mathbf{I}_{\text{nt}}$	JE and ID [-] $\mathbf{I}_{\text{mp},2} * \mathbf{I}_{\text{nt}}$
1	1.62	0.91	0.98	1.57	0.89
2	1.45	0.09	0.91	1.33	0.08
3	1.59	0.18	0.87	1.38	0.16
4	1.29	0.28	0.28	0.35	0.08
5	1.28	0.11	0.83	1.06	0.09
6	1.30	0.62	0.53	0.69	0.33
7	1.48	0.46	0.45	0.67	0.21
8	1.53	0.03	0.51	0.78	0.02

576

577 *5.4.2.1 Objective-function evaluation*

578 For each importance-factor scenario, the objective-function evaluation (trace of the modified
 579 FIM $\bar{\mathbf{Q}}$) is presented in Figure 10 with respect to the number of sensors. In order to compare
 580 the importance-factor scenarios, results are normalized using the maximum objective-function
 581 evaluation of each importance-factor scenario $\hat{\text{tr}}(\bar{\mathbf{Q}})$. For all importance-factor scenarios, the
 582 objective function reaches a near-maximum value from the 8th sensor added to the sensor
 583 configuration. This shows that more than eight sensors provide no new information. For any
 584 number of sensors, scenario $\mathbf{I}_{\text{mp},2}$ and $\mathbf{I}_{\text{mp},2} * \mathbf{I}_{\text{nt}}$ outperform other importance-factor scenarios.
 585 This result is validated using information-gain metrics (Section 5.4.2.3).



586

587 Figure 10 Normalized trace of modified Fisher Information Matrix ($\bar{\mathbf{Q}}$) for scenarios as function
 588 of number of sensors.

589 *5.4.2.2 Sensor-location ranking*

590 In this section, the sensor-location ranking for each importance-factor scenario is presented.

591 Figure 11 shows the first five sensor locations selected for each importance-factor scenario.

592 Globally, sensor locations 3, 7, 8 and 10 have the largest expected information gain as they are

593 selected in first positions in most of importance-factor scenarios. Additionally, from the first

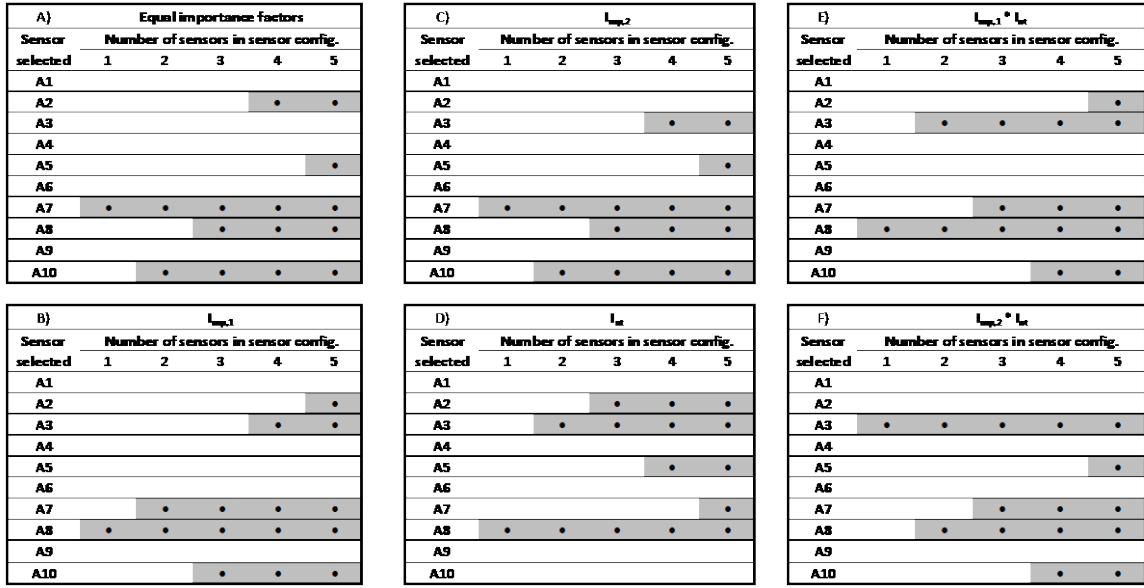
594 sensor selected, the importance-factor scenario $\mathbf{I}_{mp,1} * \mathbf{I}_{nt}$ differs in terms of sensor

595 configuration. This shows that the modification of the FIM influence sensor-location

596 evaluations to prioritize the identification of useful mode shapes for structural identification.

597 This result explains the difference in terms of objective-function evaluations (Figure 10) but

598 must be validated using information-gain metrics (Section 5.4.2.3).



599

600 Figure 11 Accelerometer-location (A_i) ranking for importance-factor scenarios presented in
 601 Table 3. A) Equal importance factors B) information entropy $I_{mp,1}$ only; C) Joint entropy $I_{mp,2}$
 602 only; D) Identifiability matrix only I_{nt} ; E) Information entropy and identifiability matrix
 603 $I_{mp,1} * I_{nt}$; F) Joint entropy and identifiability matrix $I_{mp,2} * I_{nt}$.

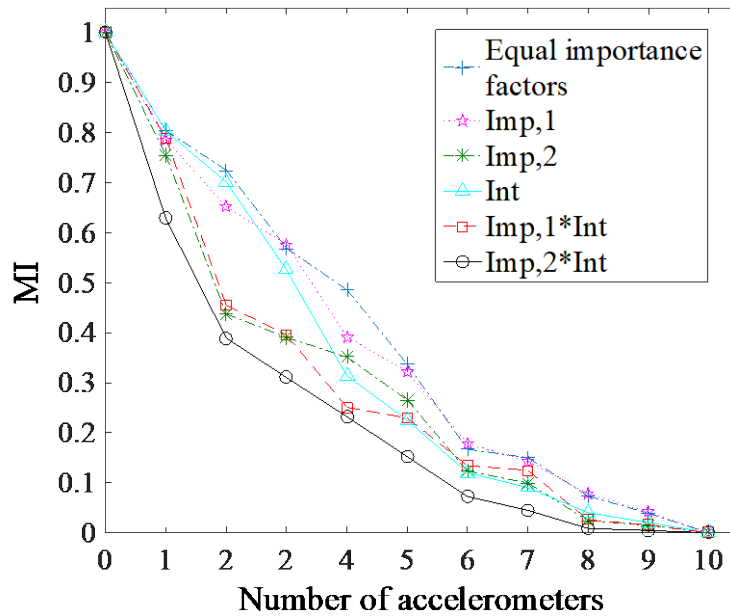
604 *5.4.2.3 Information-gain metrics for modal identification*

605 In this section, sensor configurations obtained with the importance-factor scenarios are
 606 compared using the two information-gain metrics introduced in Section 3.4. Figure 12 presents
 607 the missing information MI with respect to the number of accelerometers. Globally, as the
 608 missing information must be minimized, all importance-factor scenarios lead to an increase in
 609 information gain with the number of sensors. The traditional approach represented by the
 610 scenario with no importance factors is over-performed by all remaining scenarios, while the
 611 scenario $I_{mp,2} * I_{nt}$ outperforms all remaining scenarios, indicating a higher information gain.
 612 Additionally, for this importance-factor scenario, the missing-information metric does not
 613 significantly decrease when six sensors are used in the sensor configuration, while the drop in
 614 missing information is negligible from eight sensors in the sensor configuration. Based on the
 615 missing information as an inverse information-gain metric for modal identification, 6 to 8
 616 sensors are recommended in the sensor configuration. Next, results are validated using the
 617 second information-gain metric (QR-decomposition).

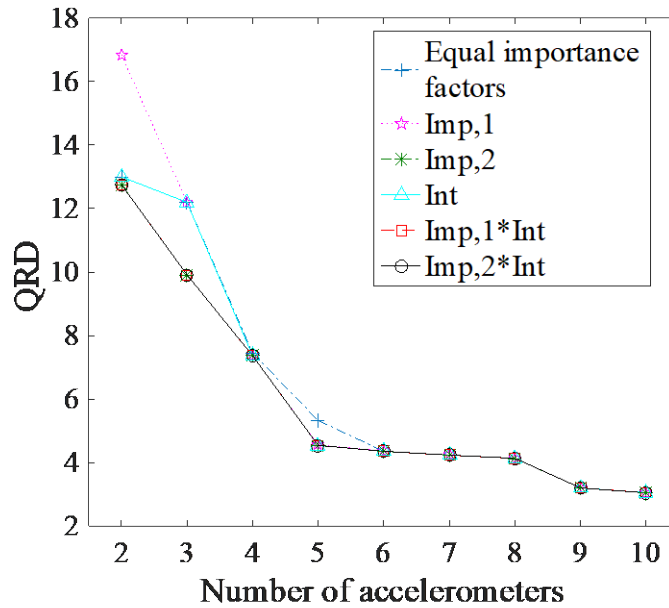
618 Figure 13 presents QR-decomposition information-gain metric (QRD) with respect to the
 619 number of accelerometers. The metric evaluates the linear independence between mode-shape
 620 vectors and must be minimized. Globally, all importance-factor scenarios lead to a decrease in

621 QRD with increasing number of sensors and they perform similarly on this information-gain
 622 metric. The QR-decomposition metric as an inverse information-gain metric for modal
 623 identification is almost constant between 5 to 8 sensors and shows a slight decrease when the
 624 9th sensor is added to the sensor configuration. Therefore, for all importance-factor scenarios,
 625 a sensor configuration of 6 to 10 sensors is recommended. The scenario $\mathbf{I}_{mp,2} * \mathbf{I}_{nt}$ performs as
 626 well as the equal-importance scenario. This shows that the prioritization of sensor-location
 627 selections to identify useful mode shapes for falsification does not affect significantly the
 628 ability to detect all mode shapes in the EFI method.

629 Regarding both information-gain metrics, the importance-factor scenario $\mathbf{I}_{mp,2} * \mathbf{I}_{nt}$ outperforms
 630 other scenarios. Using the importance and identifiability matrices lead to a prioritization of
 631 sensor-location selections to identify useful mode shapes for falsification. This important-
 632 factor scenario outperforms the traditional approach for modal identification, when all mode
 633 shapes are assumed to have the same performance for structural identification. Additionally, a
 634 sensor configuration of 6 to 9 sensors is recommended for structural identification based only
 635 on dynamic load testing.



636
 637 Figure 12 Missing-information metric (*MI*) for importance-factor scenarios as function of
 638 number of accelerometers.



639

640 Figure 13 QR-decomposition metric (QRD) for importance-factor scenarios as function of
 641 number of accelerometers.

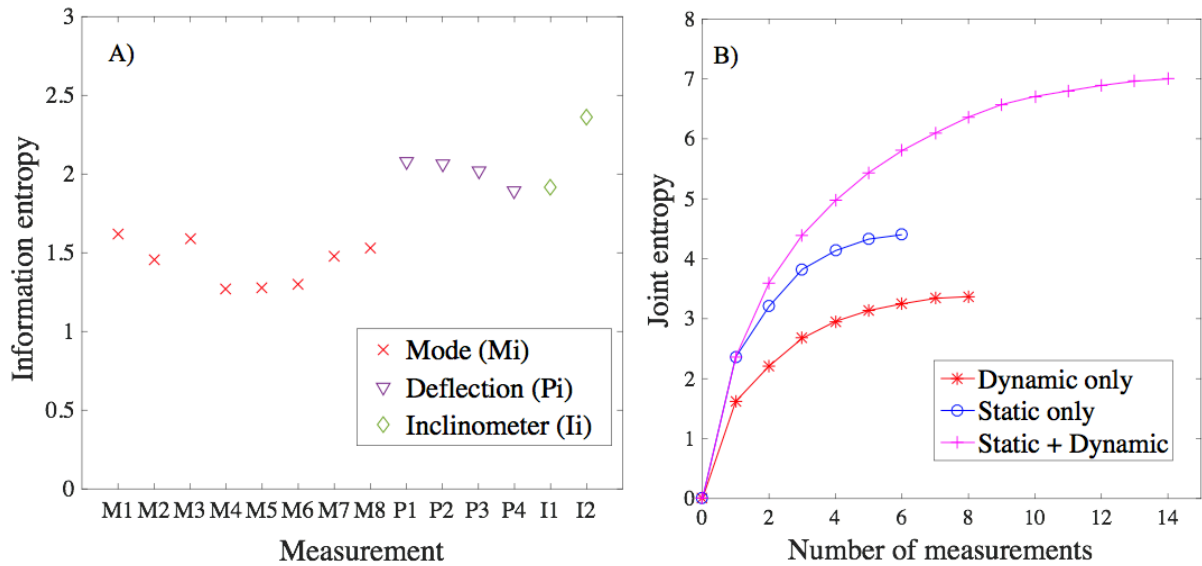
642 5.5 Measurement-system design including static and dynamic tests

643 In this subsection, the methodology presented in Section 4 for measurement-system design
 644 including information from static and dynamic tests is applied to the Singapore case study.
 645 First, the joint entropy of natural frequencies and static measurements is assessed in Section
 646 5.5.1. Then, the sensor-placement methodology for modal identification is performed in
 647 Section 5.5.2 to design the optimal measurement system for dynamic tests.

648 5.5.1 Joint-entropy assessment of static measurements and natural frequencies

649 In this section, the expected information gain from both static (deflection and inclination) and
 650 dynamic (natural frequency) measurements is assessed using the joint entropy as presented in
 651 Section 4.1. Figure 14A presents the information entropy of both static and dynamic
 652 measurements. Globally, static measurements show larger information-entropy values than
 653 natural frequencies. The inclinometer I2 presents the largest information-entropy value and is
 654 therefore the best measurement if measurements are used individually. Additionally, two
 655 sensor types are used for static tests. Figure 14B presents the joint-entropy assessment of
 656 measurement sets. Using both static and dynamic measurements lead to significantly larger
 657 joint entropy values than using static or dynamic measurements individually for any number
 658 of measurements in the measurement set. This shows that both tests provide unique information
 659 and should be used for structural identification of this case. Multi-sensor types have been

660 shown to reduce the mutual information between measurements [30]. The joint entropy of both
 661 tests increases for any measurements added to the measurement set. Each measurement
 662 provides useful additional information.



663
 664 Figure 14 A) Information entropy of measurements; B) Joint entropy as function of number of
 665 measurements.

666 Figure 15 presents the measurement ranking from the joint-entropy assessment when both
 667 static and dynamic measurements are used. As shown using information-entropy evaluation
 668 (Figure 14A), inclinometer I2 is selected as the best measurement for structural identification.
 669 Although presenting a smaller individual information entropy than other static measurements
 670 (Figure 14A), the first natural frequency is selected as the second-best measurement, showing
 671 that this dynamic measurement provides more additional information when associated with the
 672 static measurement I2 than any remaining static measurements. Then, the deflection
 673 measurement P1 is selected as the third-best measurement, showing that each type of
 674 measurements provides useful additional information. Once the measurement joint entropy is
 675 assessed, the optimal sensor configuration for the static test is directly obtained. However, the
 676 sensor-placement methodology for modal identification must be performed (Section 3.3) in
 677 order to define the optimal sensor configuration for dynamic tests.

Measurement selected	Number of measurements													
	1	2	3	4	5	6	7	8	9	10	11	12	13	14
M1		•	•	•	•	•	•	•	•	•	•	•	•	•
M2												•	•	•
M3													•	•
M4				•	•	•	•	•	•	•	•	•	•	•
M5							•	•	•	•	•	•	•	•
M6												•	•	•
M7						•	•	•	•	•	•	•	•	•
M8								•	•	•	•	•	•	•
P1			•	•	•	•	•	•	•	•	•	•	•	•
P2														•
P3											•	•	•	•
P4										•	•	•	•	•
I1					•	•	•	•	•	•	•	•	•	•
I2	•	•	•	•	•	•	•	•	•	•	•	•	•	•

678

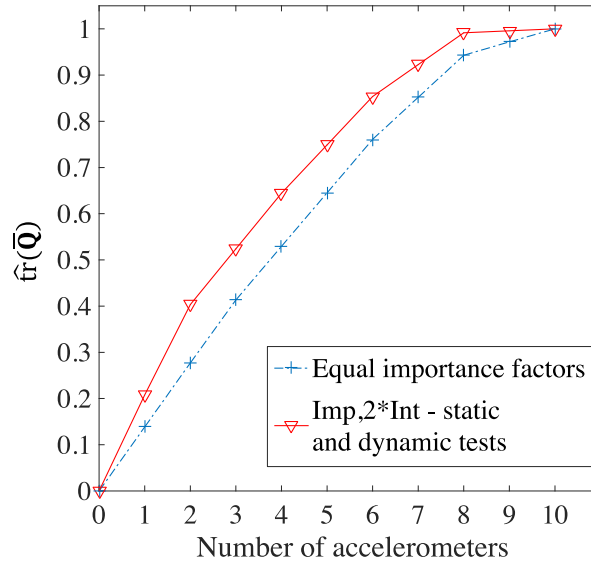
679 Figure 15 Measurement ranking for measurement-system design including static and dynamic
680 tests, where M_i are modes, P_i deflection-target and I_i inclinometer locations.

681 **5.5.2 Sensor-placement for modal identification**

682 Once the expected information gain of each mode shape is quantified, the same procedure as
683 presented in Section 5.4.2 is performed to obtain the dynamic measurement system. Compared
684 with the previous section, the main difference is that the importance-factor matrix is modified
685 with the new quantification of expected modal information gain from the joint-entropy
686 assessment of both tests (Figure 14B).

687 The important-factor evaluation for both static and dynamic measurements is compared in
688 terms of objective-function evaluation, sensor-location ranking and information-gain metrics
689 to the dynamic only importance-factor evaluation and the traditional approach where mode
690 shapes are taken to be equally important.

691 Figure 16 presents the normalized objective-function evaluation ($\widehat{\text{tr}}(\bar{\mathbf{Q}})$) with respect to the
692 number of accelerometers. Results show that the new importance-factor evaluation
693 outperforms the scenario with equal importance factors. This result confirms previous
694 observations (Section 5.4.2) where scenarios involving various importance factors lead to larger
695 objective-function evaluations than the traditional approach where mode shapes are taken as
696 equally important for structural identification.



697

698 Figure 16 Normalized trace of modified Fisher Information Matrix (\bar{Q}) for importance-factor
 699 scenarios as function of number of accelerometers.

700 Figure 17 presents sensor-location ranking for the three impact-factor scenarios for the five
 701 first sensor locations selected. Globally, sensor-location rankings of importance-factor
 702 scenarios differ, showing that the choice of importance factors significantly influences sensor-
 703 location evaluations. When importance-factor evaluations include dynamic tests individually
 704 and static and dynamic tests, the first two sensor locations selected remain the same. However,
 705 from the third sensor-location selection, sensor-location selections differ, showing that small
 706 difference in importance factor may significantly influence sensor-location selections.

A) Equal importance factors					
Sensor selected	Number of sensors in sensor config.				
	1	2	3	4	5
A1					
A2				•	•
A3					
A4					
A5					•
A6					
A7	•	•	•	•	•
A8			•	•	•
A9					
A10		•	•	•	•

B) $I_{mp,2} * I_{nt}$ - Dynamic test only					
Sensor selected	Number of sensors in sensor config.				
	1	2	3	4	5
A1					
A2					
A3	•	•	•	•	•
A4					
A5					•
A6					
A7			•	•	•
A8		•	•	•	•
A9					
A10				•	•

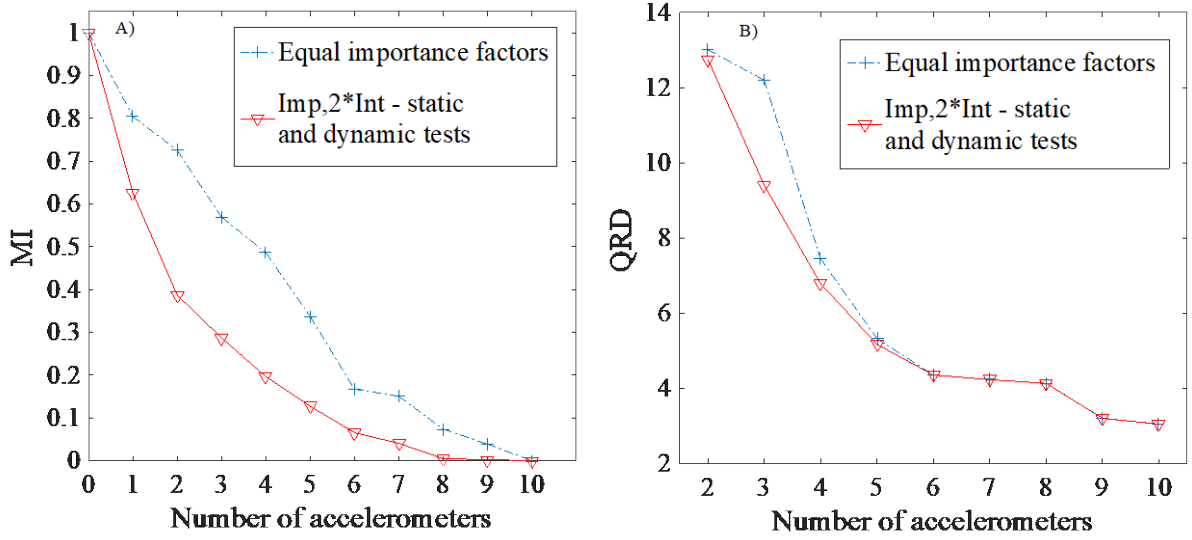
C) $I_{mp,2} * I_{nt}$ - Static + dynamic tests					
Sensor selected	Number of sensors in sensor config.				
	1	2	3	4	5
A1					
A2			•	•	•
A3	•	•	•	•	•
A4					
A5				•	•
A6					
A7					•
A8		•	•	•	•
A9					
A10					

707

708 Figure 17 Accelerometer-location (A_i) ranking for importance-factor scenarios of mode-shape
 709 matrix. A) Scenario of equal importance of mode shapes; B) Importance scenario of $I_{mp,2} * I_{nt}$
 710 using the joint entropy of dynamic test only; C) Importance scenario of $I_{mp,2} * I_{nt}$ using the joint
 711 entropy for static and dynamic tests.

712 Figure 18 presents MI and QRD metrics for modal identification for the two impact-factor
 713 scenarios with respect to the number of accelerometers. The missing information gain metric

714 MI (Figure 18A) shows a better performance if importance factors are used. As all mode-
 715 shapes are useful even if static measurements are added to dynamic measurements (Figure
 716 14B), the number of sensors needed to identify correctly all mode shape remains constant.
 717 Therefore, the recommended number of accelerometers (i.e. 6 to 9) remains similar to the
 718 previous analysis for dynamic load tests only (Section 5.4.2.3). The main difference is that
 719 accelerometer locations are modified since the sensor-location ranking is re-evaluated (Figure
 720 17).



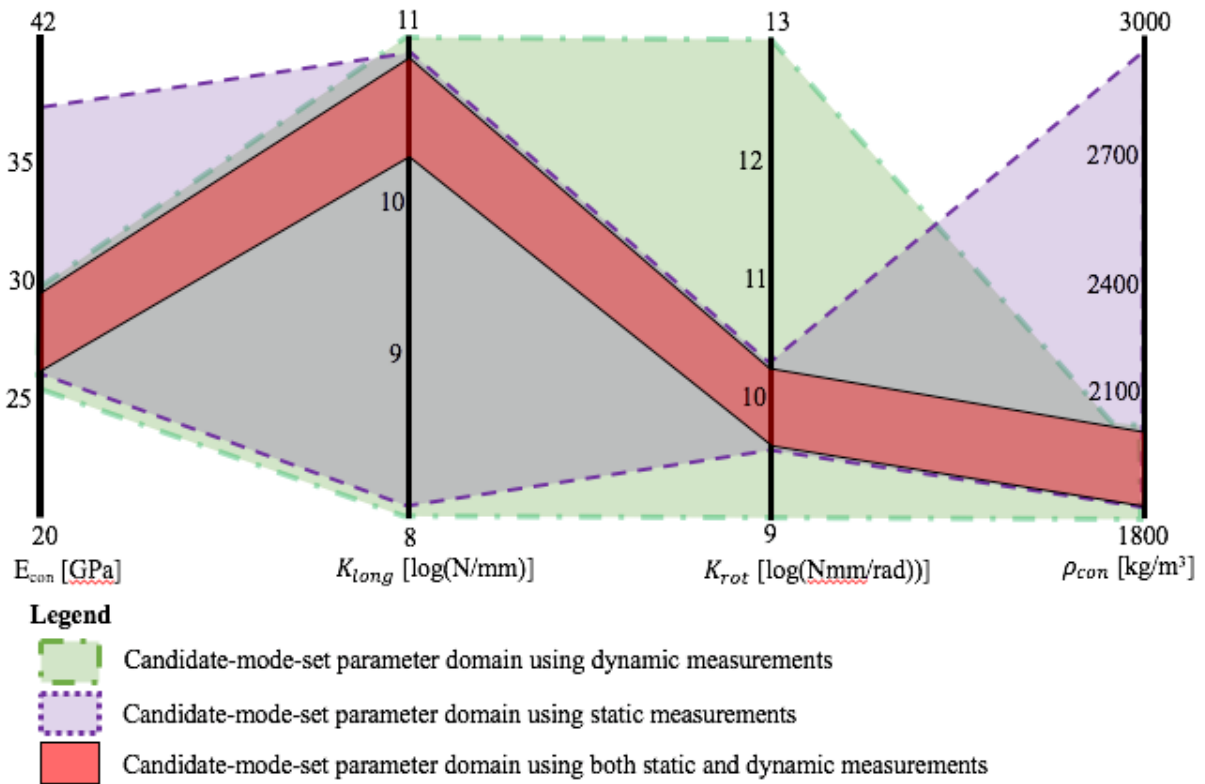
721
 722 Figure 18 A) Missing-information metric for importance-factor scenarios as function of
 723 number of accelerometers; B) QR-decomposition metric for importance-factor scenarios as
 724 function of number of accelerometers.

725 **5.6 Result corroboration using measurements taken from a full-scale**
 726 **bridge**

727 As mentioned in Section 5.1, this study is performed after the bridge was monitored.
 728 Performance of both static and dynamic measurements is assessed based on their ability to
 729 falsify model instances. In the joint-entropy assessment of static and dynamic measurements
 730 (Figure 14, Section 5.5.1), both tests are expected to provide unique information. Figure 19
 731 presents candidate-model-set domains obtained using static and dynamic measurements
 732 independently and static and dynamic measurements combined. Results are taken from Cao et
 733 al. [47], where a surrogate model was built to generate more initial model instances in order to
 734 better cover the model-parameter domain of candidate-model sets. Static measurements (i.e.
 735 all inclinometers and deflection targets) are useful to reduce parameter ranges for the first and
 736 third parameters (E_{con} and K_{rot}). They cannot help in the identification of the second and the
 737 fourth parameters (K_{long} and ρ_{con}). Figure 19 presents the boundaries of the parameter domains.

738 Among static candidate models, large parameter values of E_{con} are associated with small values
 739 of K_{Long} and inversely.

740 Dynamic measurements (i.e. all natural frequencies) reduce parameter ranges for the first and
 741 the fourth parameters (E_{con} and ρ_{con}), but have no impact on parameter ranges for the second
 742 and third parameters (K_{long} and K_{rot}). When static and dynamic measurements are combined,
 743 all model-parameter ranges are significantly reduced, showing a good structural identification
 744 even for the second model parameter K_{long} that is not well identified when static and dynamic
 745 tests are used independently. This shows that both tests provide useful information and when
 746 they are combined, the information gained is significantly increased. This result using bridge
 747 measurements corroborates the expected performance of static and dynamic tests presented in
 748 the sensor-placement study.



749
 750 Figure 19 Parallel-axis plot of candidate-model-set parameter domain using static
 751 measurements; dynamic measurements; static and dynamic measurements. Adapted from [47].

752 In the study of Cao et al. [47], 10 accelerometers were used to evaluate real bridge modal
 753 frequencies and then the model-falsification process was performed. However, the present
 754 study suggests that only six accelerometers provide significant new information, while the
 755 remaining four present a large amount of redundant information (Section 5.5). Natural-
 756 frequency evaluations using only the six best accelerometers are performed again and results

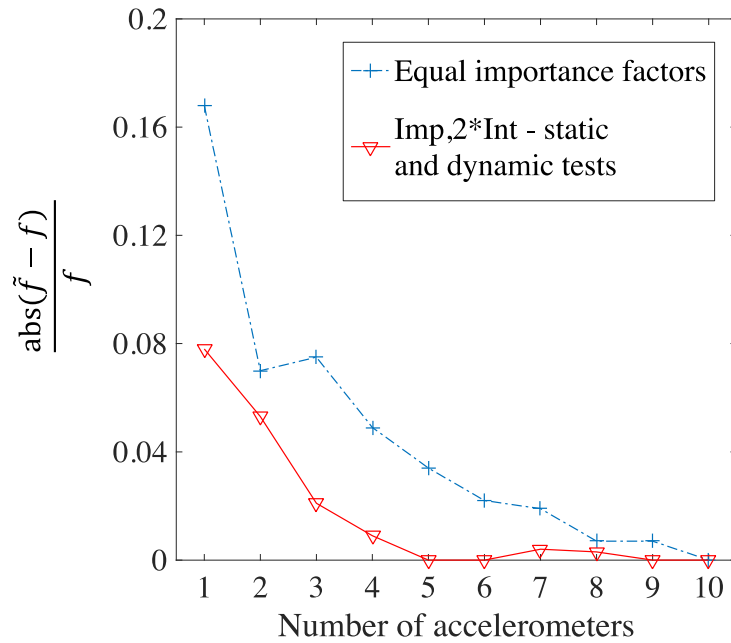
757 in terms of parameter ranges of candidate model sets are presented in Table 4. When using the
758 six best accelerometers instead of all accelerometers, small differences of parameter
759 identification are revealed for E_{con} and ρ_{con} . However, when static and dynamic measurements
760 are combined, changes in parameter-identification ranges are more important. Using the
761 accelerometer subset obtained with the proposed methodology does not significantly influence
762 the candidate-model set. Therefore, results show that the proposed methodology help reduce
763 the number of sensors without compromising the performance of structural identification.

764 Table 4 Identification of parameter ranges using static measurements; dynamic measurements
765 and static and dynamic measurements. Adapted from [47].

Load testing	E_{con}		K_{long}		K_{rot}		ρ_{con}	
	[Gpa]		[log(N/mm)]		[log(Nmm/rad)]		[kg/m ³]	
	min	max	min	max	min	max	min	max
Static only	28.1	37.0	9.2	12.8	8.4	8.9	1800	2900
Dynamic only – All accelerometers	26.7	29.9	9.0	13.0	8.0	11.0	1800	1950
Dynamic only – 6 best accelerometers	26.2	29.7	9.0	13.0	8.0	11.0	1800	2000
Static + Dynamic – All accelerometers	28.1	29.9	12.2	12.8	8.5	8.9	1800	1950
Static + Dynamic – 6 best accelerometers	28.1	29.7	12.2	12.8	8.5	8.9	1800	2000

766 Additionally, information-gain metrics (Figure 18) provide information related to the minimum
767 number of sensors to accurately identify mode shapes. Information-gain metrics show that 6
768 and 9 accelerometers are required for two scenarios of sensor placement respectively: $\mathbf{I}_{mp,2} * \mathbf{I}_{nt}$
769 and the equal importance factor. Figure 20 presents the error in the measured natural frequency
770 of the first mode with respect to the number of accelerometers. The error in measured natural
771 frequencies is evaluated as the difference between the frequency obtained when a subset of
772 sensors is used \tilde{f} with the frequency obtained considering all accelerometers f . Results are
773 normalized using f and are compared for sensor configurations selected by both scenarios
774 (Figure 17). The error in frequency estimation decreases while increasing the number of
775 accelerometers for both scenarios. Sensor configurations selected by the $\mathbf{I}_{mp,2} * \mathbf{I}_{nt}$ scenario
776 provide better approximation of the first natural frequency than the equal-importance scenario.
777 As the first mode provides the largest increase of joint entropy (Figure 14), this result
778 demonstrates that the $\mathbf{I}_{mp,2} * \mathbf{I}_{nt}$ scenario prioritizes sensor locations to identify useful modes.

779 Additionally, results show the $\mathbf{I}_{mp,2} * \mathbf{I}_{nt}$ scenario requires at least 5 sensors to approximate
 780 accurately the natural frequency while the equal-importance-factors scenario requires 8
 781 sensors, corroborating results of information-gain metrics. Globally, results show that the
 782 expected performance of accelerometers of this study is corroborated using field measurements
 783 [47].



784
 785 Figure 20 Error in the measured natural frequency for the first mode shape as function of the
 786 number of accelerometers.

787 6 Discussion

788 The following limitations of the work are recognized. First, the objective function for modal
 789 identification in the dynamic methodology employs only information on the diagonal of the
 790 FIM to select sensor locations. This choice was motivated by the fact that the number of sensors
 791 was low compared with the number of mode shapes. Therefore, the determinant of the FIM
 792 equals zero and thus cannot be used as sensor-placement objective function. When the number
 793 of sensors is larger than the number of useful mode shapes, the determinant of the FIM could
 794 be a more appropriate objective function for sensor placement (Section 3.3.3). The determinant
 795 involves complete information contents of the FIM, while the trace involves only information
 796 on the diagonal of the FIM.

797 Greedy search is used as optimization algorithm to reduce the computational time. However,
 798 it may lead to suboptimal sensor configurations, especially when there are few sensors. In order
 799 to combine information of both static and dynamic load testing, joint-entropy evaluations of

800 respective measurements need to be compared. This implies that the same initial model sets of
801 model-parameter values are used. Thus, a single model class is used and it should include all
802 model parameters influencing both static and dynamic predictions.

803 Selections of mode shapes and possible sensor locations are treated as engineering decisions.
804 However, it is recommended to include large sets in order to find optimal measurement
805 systems. Sensor locations that were involved in the previous study [47] are employed in order
806 to compare the expected performance of both static and dynamic tests with reality. Results of
807 the sensor-placement study are corroborated using real measurements. However, this is not a
808 firm result validation since sensor-placement strategies provide only a statistical advantage in
809 terms of expected performance. A large sample of case studies must be compared in order to
810 achieve result validation that would justify generalization of any sensor-placement
811 methodology. To validate a sensor configuration for modal identification, real mode-shape
812 vectors should be compared using a complete set of sensors and using a subset with only
813 important sensor locations.

814 The proposed measurement-system design framework takes into account only information-
815 gain metrics. In addition to the information gain, characteristics such as monitoring costs,
816 sensor installation and robustness of information gain to sensor failure should be evaluated to
817 design an optimal measurement system. Additionally, the criterion-weighting preferences of
818 asset managers must be supported. A framework using a multi-criteria-decision-analysis
819 approach and five performance metrics has recently been proposed in [48].

820 The success of any sensor-placement methodology depends directly on the quality of the
821 numerical behavior model that is used to obtain predictions. Before installing the sensor
822 configuration, the material constants and reliability of model assumptions should be verified
823 using visual inspection and non-destructive testing methods.

824 The information collected through load testing in the elastic domain is used to update the
825 structural behavior at limit-state conditions. If the critical limit state is the ultimate limit state
826 and the response is non-linear behavior that is difficult to model, as would be expected for a
827 reinforced concrete beams in bending, structural-identification methodologies may provide
828 limited information since the updated-parameter sensitivity is low. Further research is also
829 needed to recommend appropriate values of numerical-simulation uncertainties [49]. In such
830 situations, while the interest of measurement-system-design methodologies is reduced,
831 methodologies remain valid.

832 **7 Conclusions**

833 A well-designed measurement system enhances structural identification and reserve-capacity
834 assessment. The following are the specific conclusions of the study:

- 835 • The measurement-system-design methodology that is proposed in this paper for
836 dynamic load tests supports asset managers for tasks of finding optimal sensor
837 configurations and quantifying expected information gain of natural frequencies for
838 structural identification.
- 839 • Taking into account importance and identifiability matrices helps prioritize sensor
840 locations in the effective independence method (EFI) for modal identification without
841 requiring additional uncertainty information.
- 842 • Combining information from measurement-system-design methodologies for static and
843 dynamic load testing leads to measurement systems where the redundancy of
844 information gain is minimized in a unique way.

845 Future work will involve quantifying the expected influence of monitoring information gain
846 for both static and dynamic load testing to assess reserve capacity.

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855 **Conflict of interest**

856 The authors declare that they have no conflict of interest.

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