

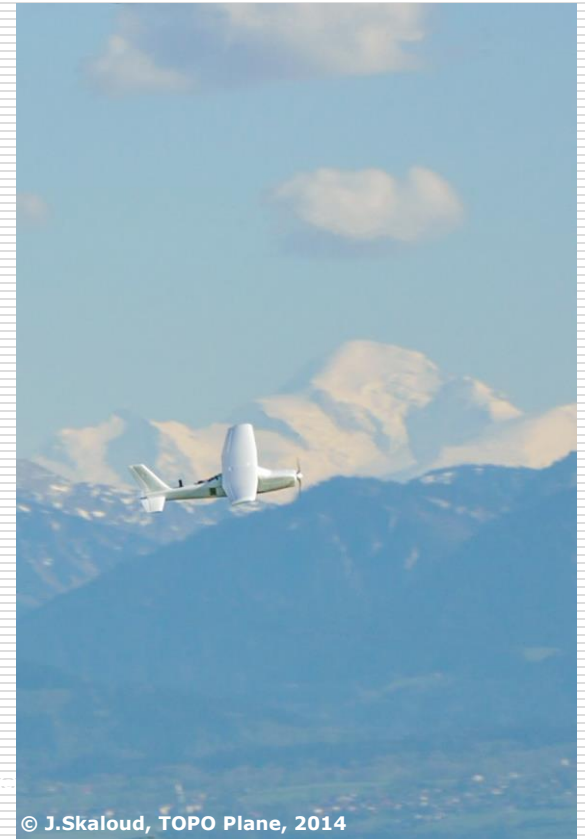
# Erhöhte Zuverlässigkeit der GPS Drohnennavigation dank einem integrierten Barometer

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# Motivation

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- River
- Highway
- Privat Movies
- ...



# What's on the menu

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- Barometer
  - Physics
  - Problems
  
- UAV Platform
  
- Methodology & testing
  - System calibration
  
- Performance with GNSS outage
  
- Conclusion



# Barometer – Altitude determined via pressure

- Model of a “perfect” gas

$$H_i - H_0 = -\frac{RT_0}{Mg} \log\left(\frac{P_i}{P_0}\right)$$

$R$ : universal gas constant

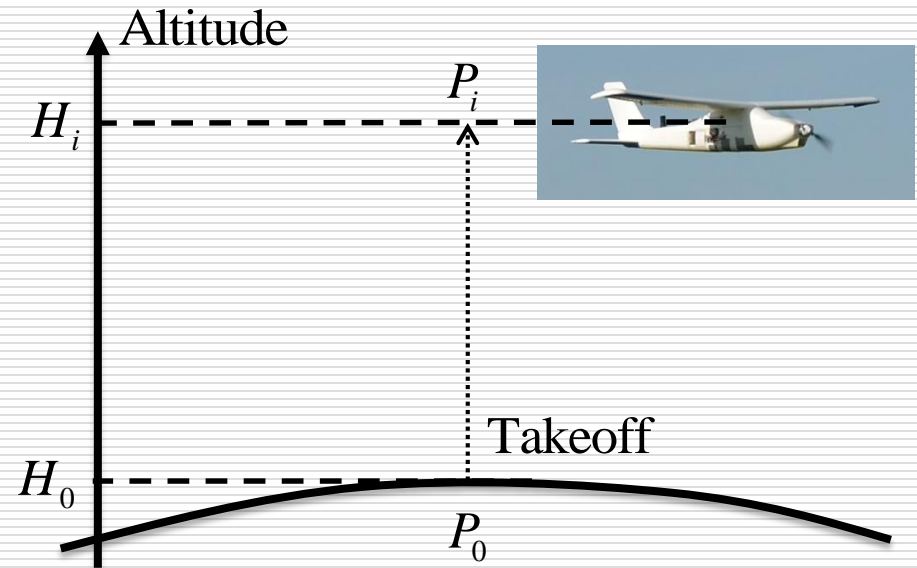
$g$ : gravitational acceleration

$M$ : molar mass of air

$T_0$ : air temperature

$P_0$ : reference pressure of takeoff

$P_i$ : pressure at certain height

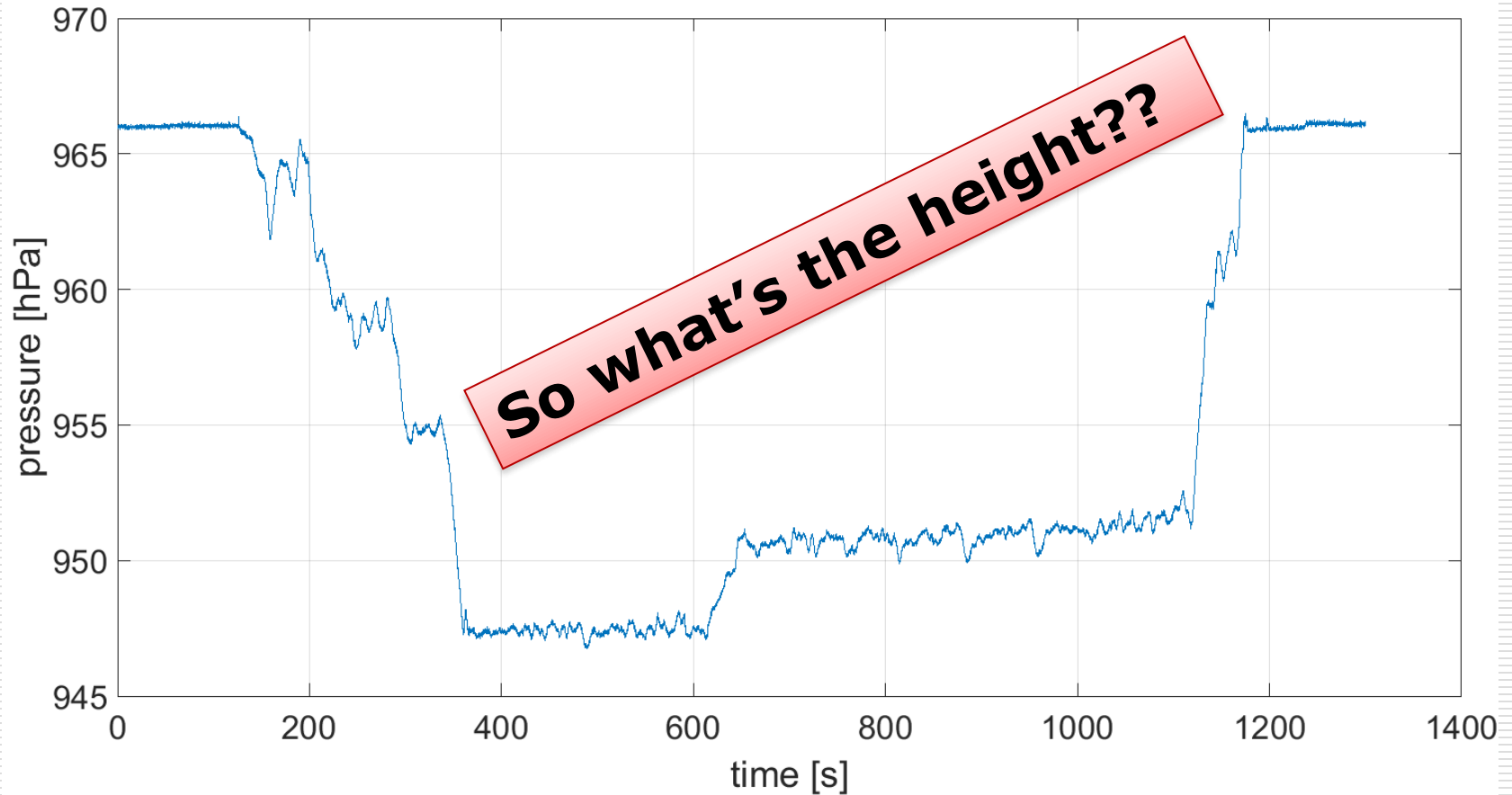


# Barometer – Influence of parameter variations

$$H_i - H_0 = -\frac{RT_0}{Mg} \log\left(\frac{P_i}{P_0}\right)$$

	Unit	Typical value	$\Delta$	$\Delta H$ [m]
$T_0$	K	293.15 (= 20°C)	$\pm 1$	$\pm 0.61$
$P_i$	hPa	950	$\pm 1$	$\pm 8.84$
$P_0$	hPa	970	$\pm 1$	$\pm 9.02$
$M$	kg/mol	0.029	$\pm 0.0001$	$\pm 0.62$
$g$	m/s <sup>2</sup>	9.80665	$\pm 0.001$	$\pm 0.02$

# Barometer – Observations on typical flight



# TOPO Plane – Characteristics

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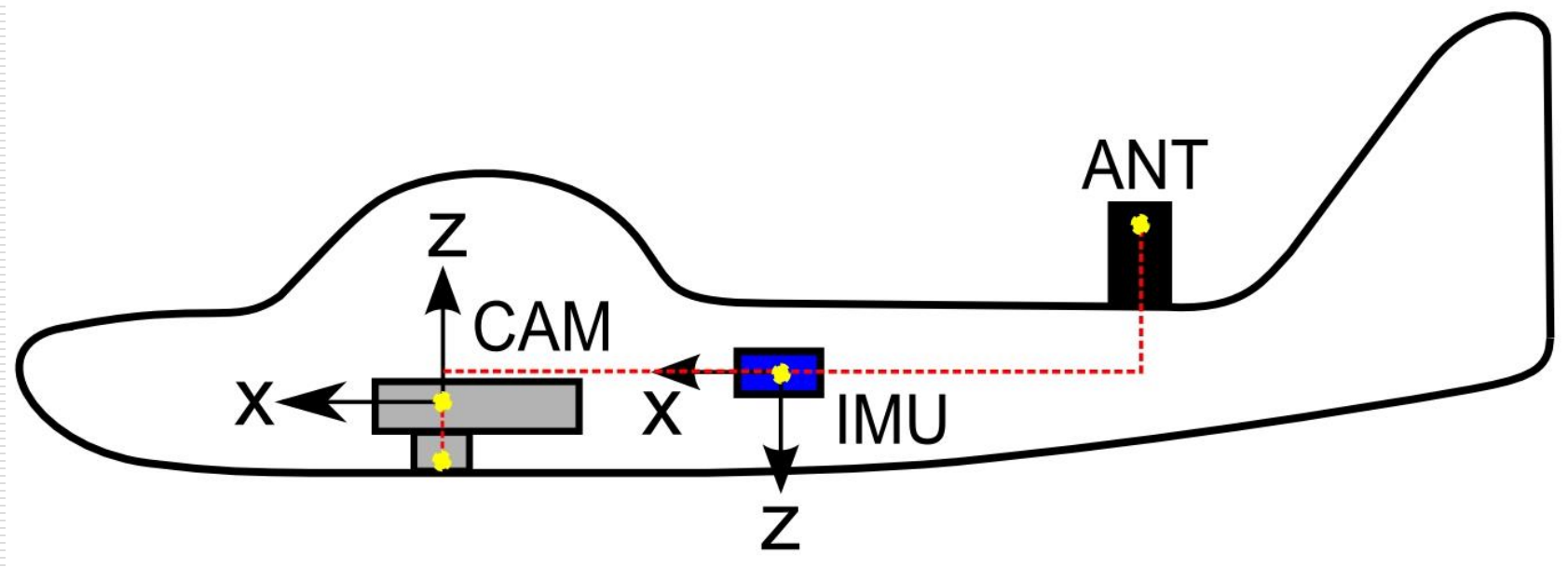
- ❑ Custom built
  - 150 € frame (as MAVinci)
  - Off-the-shelf components
  
- ❑ 1630 x 1170 mm
  
- ❑ Operational weight 2.8 kg
  
- ❑ Endurance of 40 min with 600g of payload
  
- ❑ Flying speed 16-20 m/s
  
- ❑ Pixhawk autopilot





# TOPO Plane – Characteristics

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# TOPO Plane – Payload

## ❑ Redundant-IMU (A)

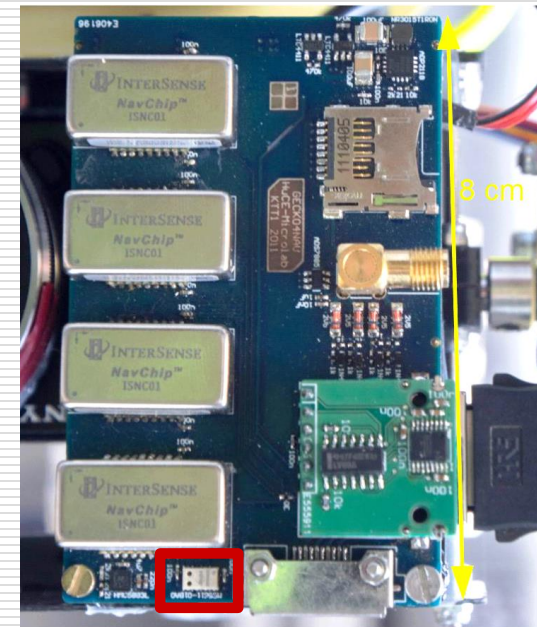
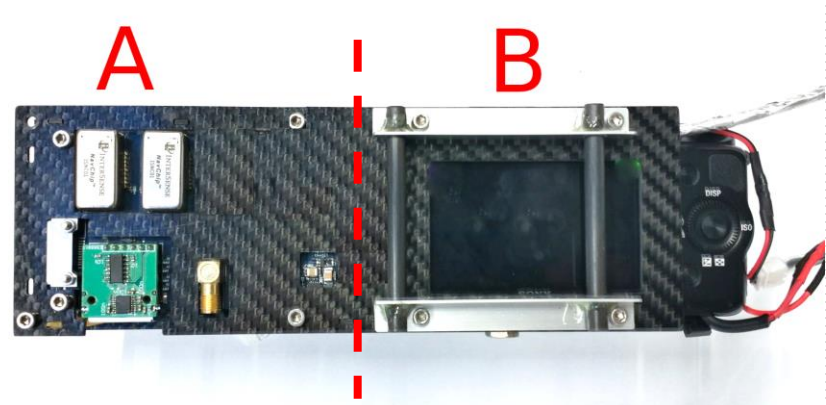
- FPGA board
- 1-4 x MEMS IMU
- 250 – 1000 Hz

## ❑ Camera (B)

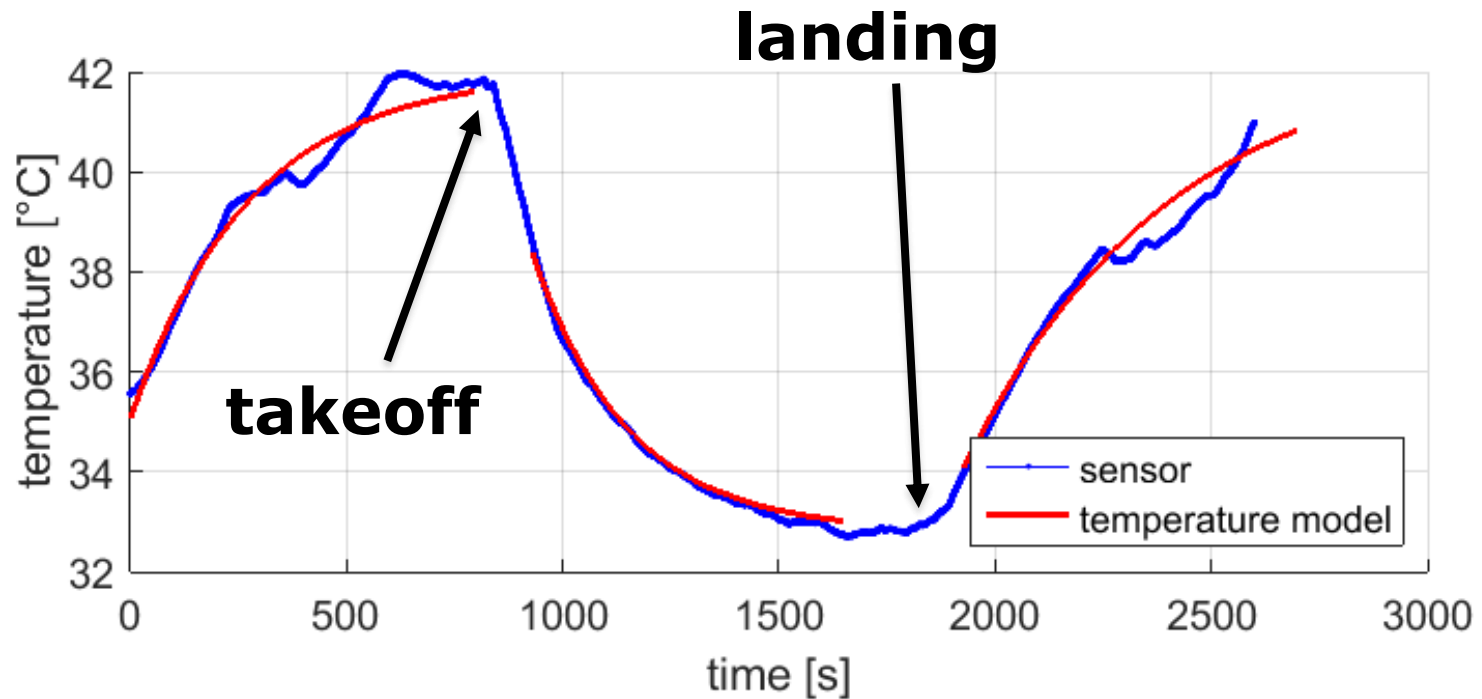
- Sony NEX 5T camera (16 Mpx)
- 16 mm lens (used in test)
- synchronization module (flash)

## ❑ GNSS

- multi freq., PPS, Event
- GPS/Glonass L1/L2 antenna



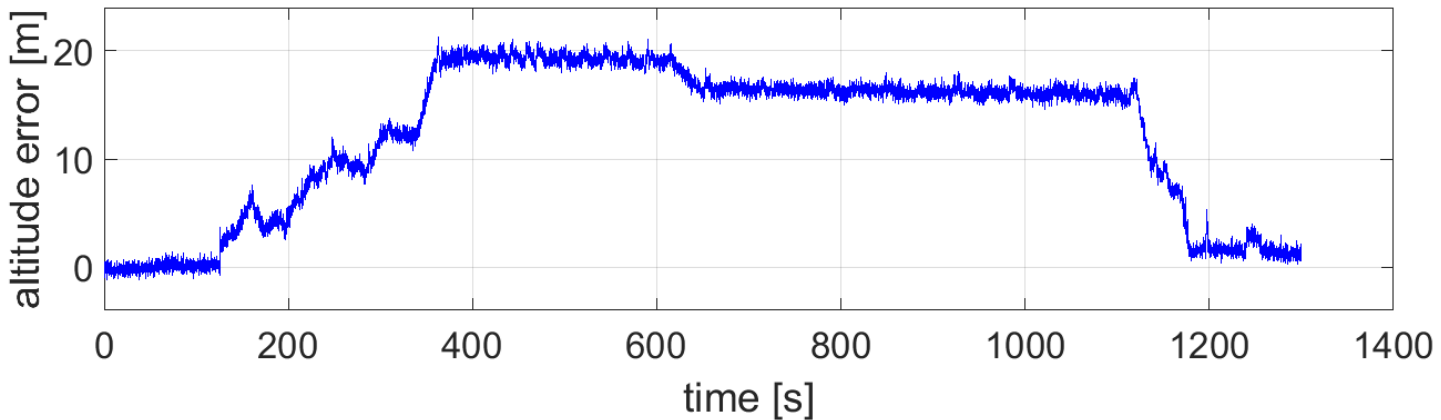
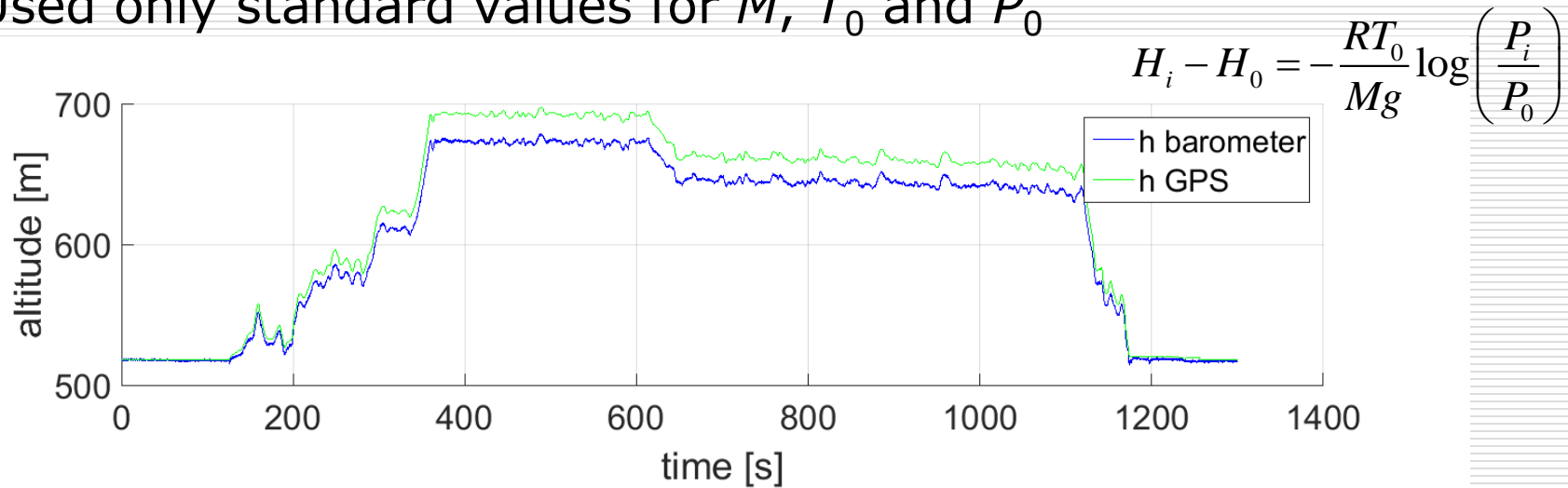
# Problem – Heating



$$H_i - H_0 = -\frac{RT_0}{Mg} \log\left(\frac{P_i}{P_0}\right)$$

# Problem – NO parameter calibration

- Used only standard values for  $M$ ,  $T_0$  and  $P_0$



# Problems

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- Influence of temperature
- Influence of humidity
- Influence of reference pressure

$$H_i - H_0 = -\frac{RT_0}{Mg} \log\left(\frac{P_i}{P_0}\right)$$

$$H_i - H_0 = -\frac{RT_0}{Mg} \log\left(\frac{P_i}{P_0}\right)$$

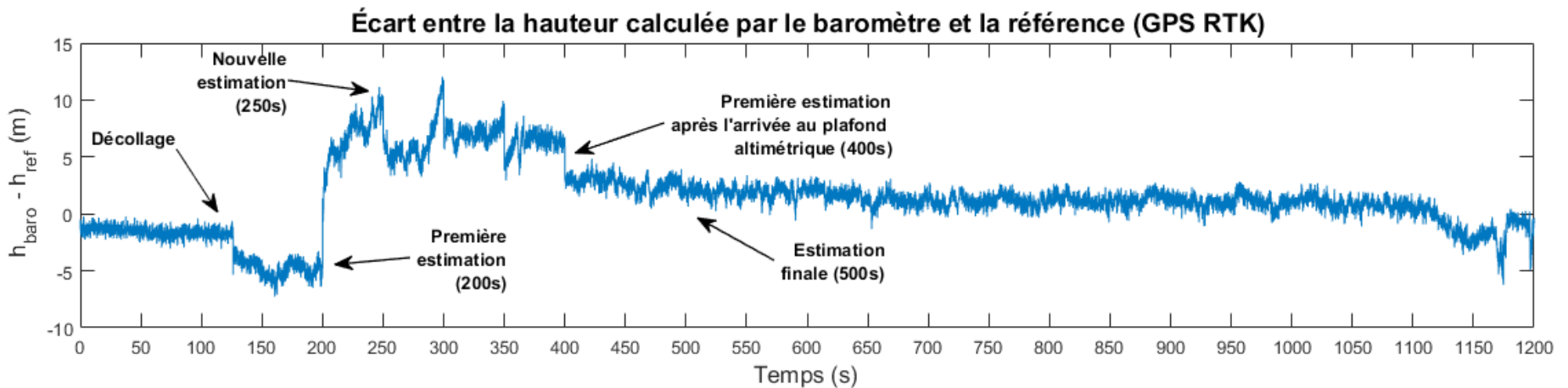
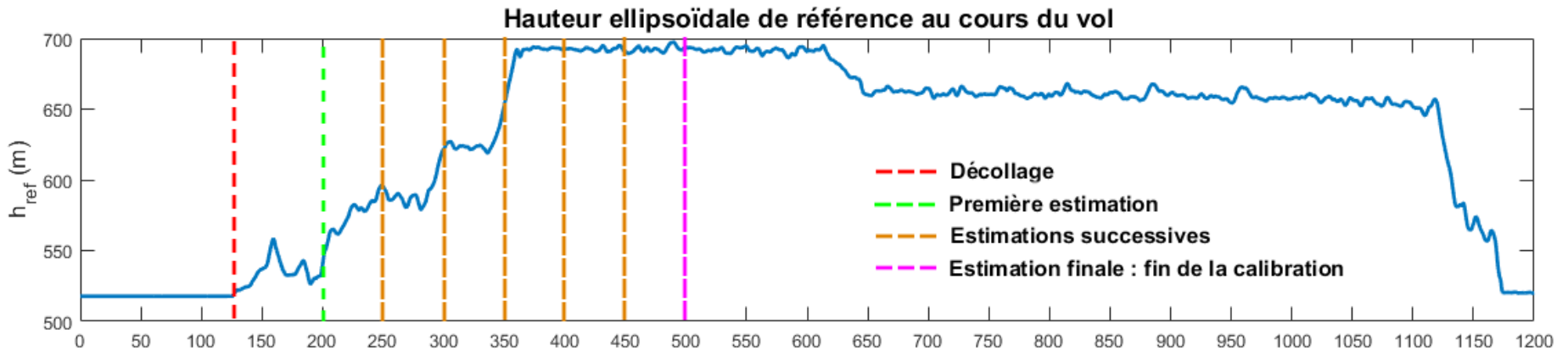
$$H_i - H_0 = -\frac{RK}{g} \log\left(\frac{P_i}{P_0}\right)$$

**we want**

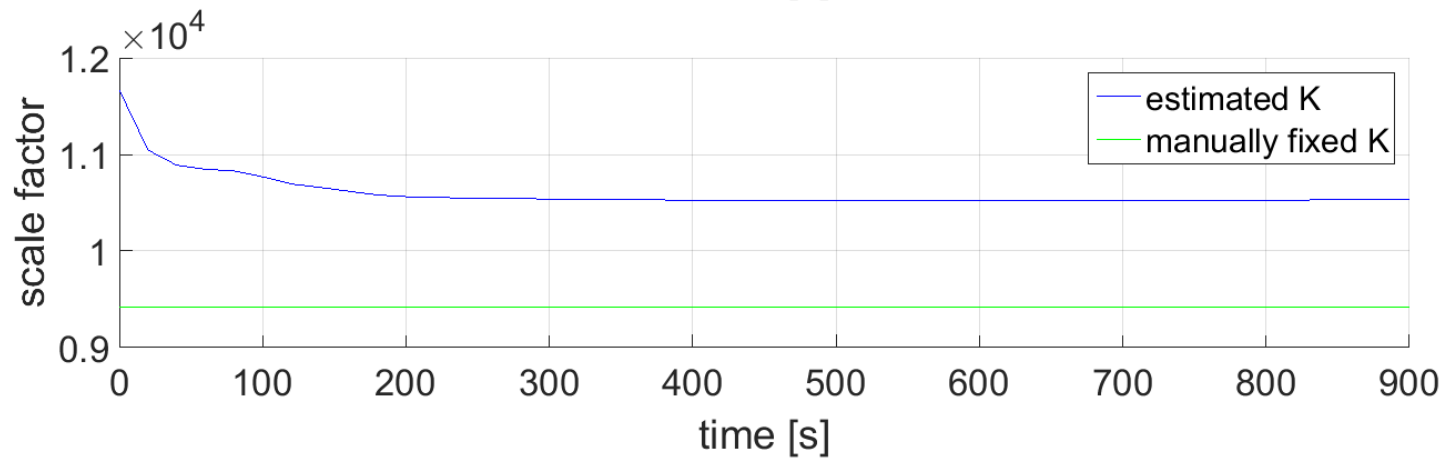
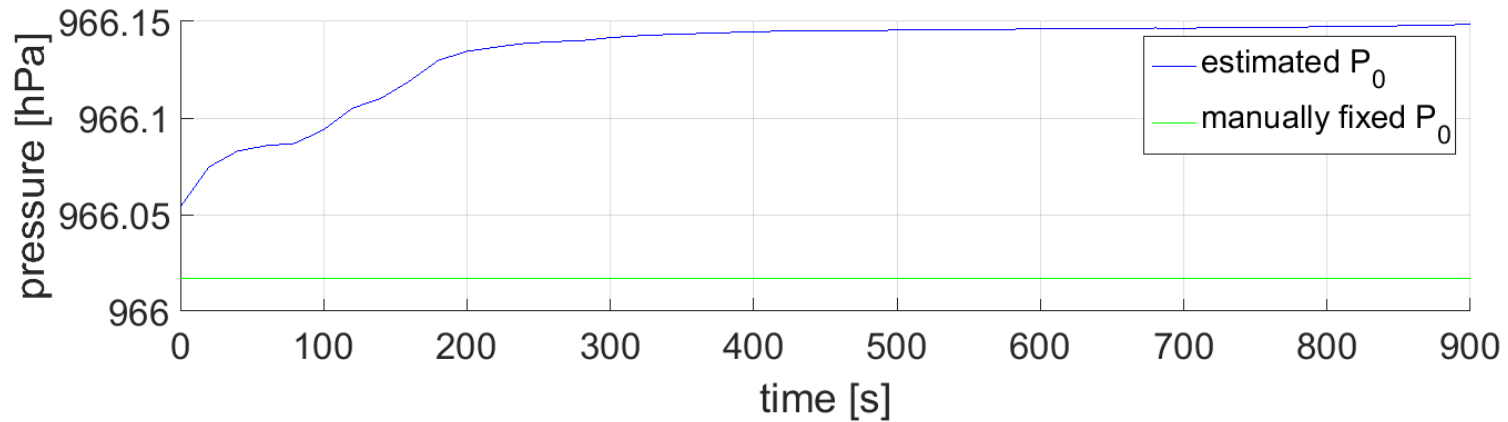
**we measure**

**we estimate**

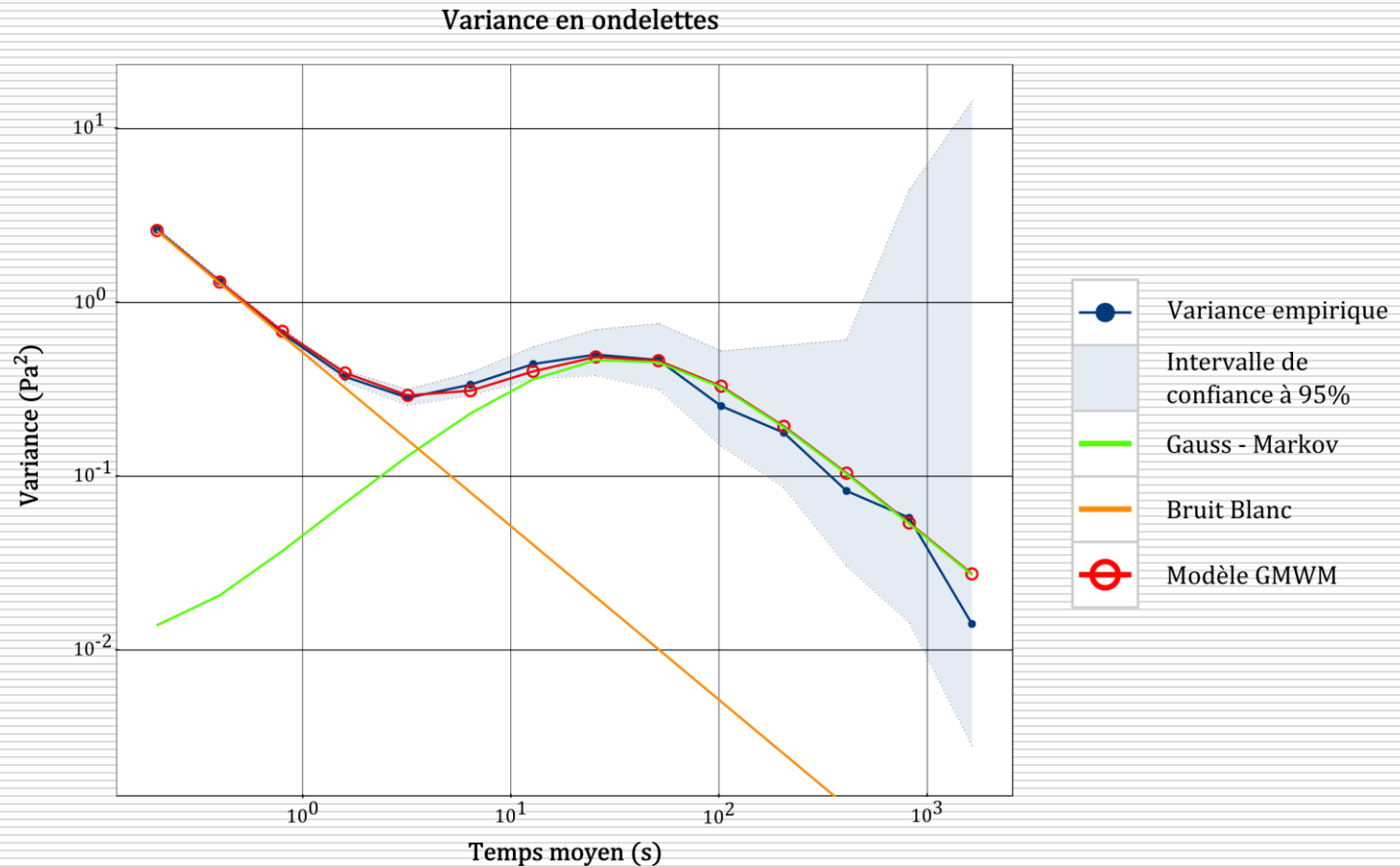
# Least Squares with Code-measurements



# Parameter estimation

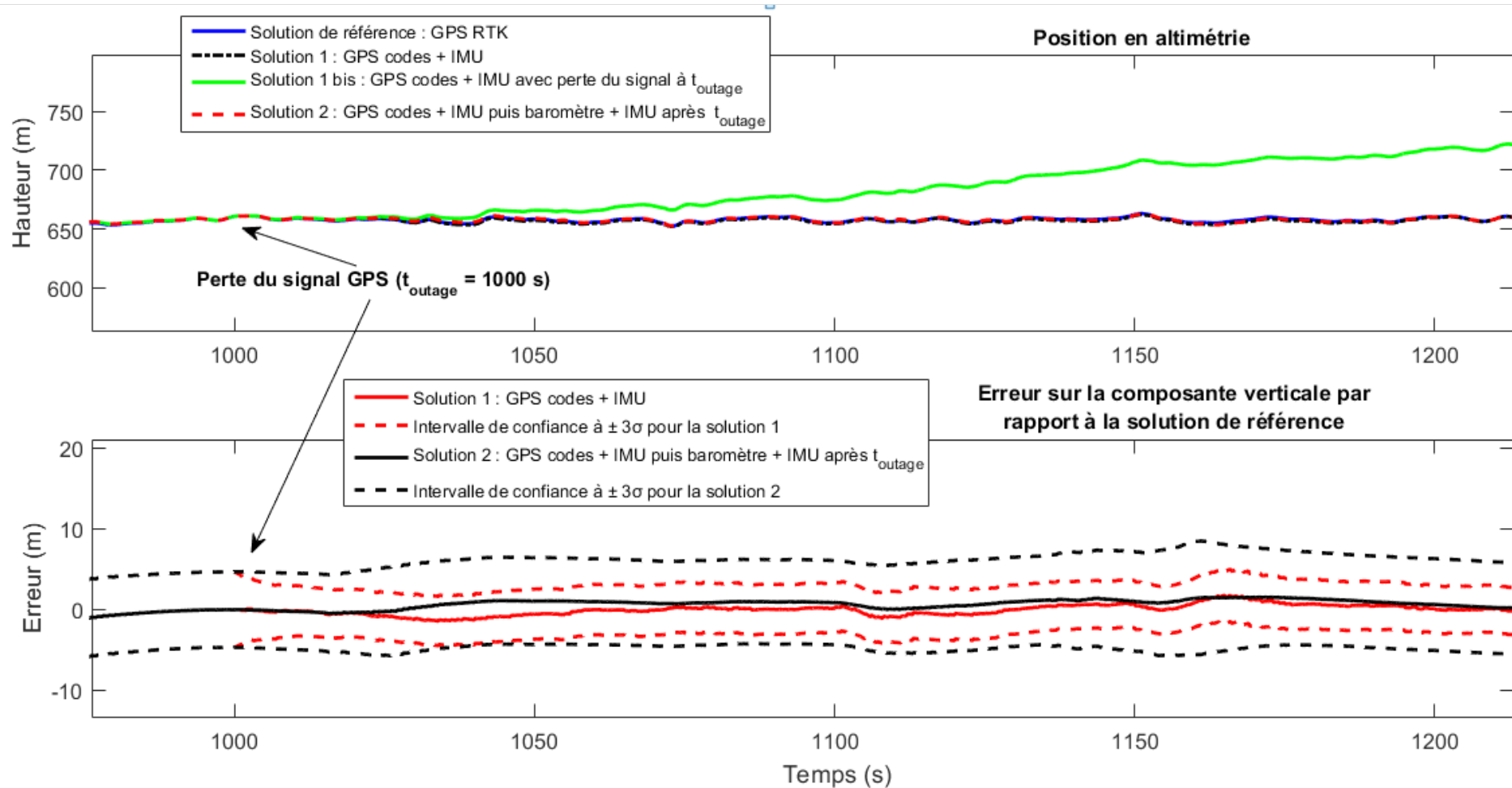


# Stochastic Noise characteristics via GMWM



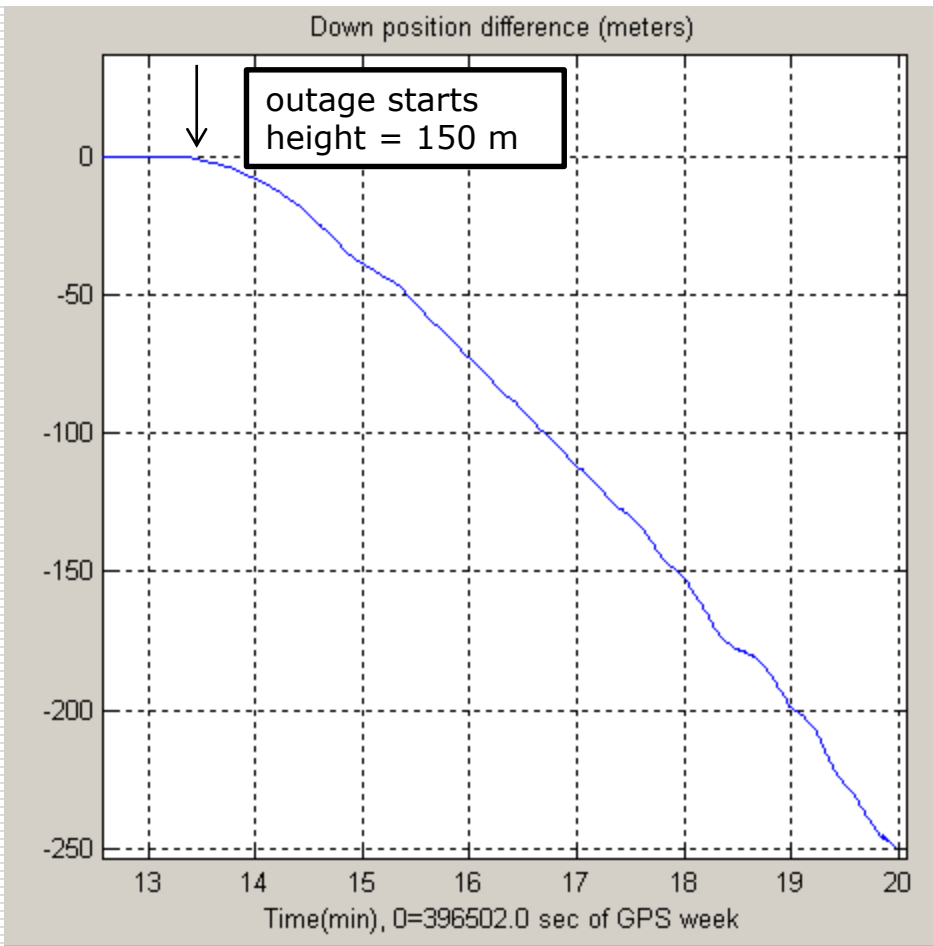


# Example - Impact in height with GPS outage (s)

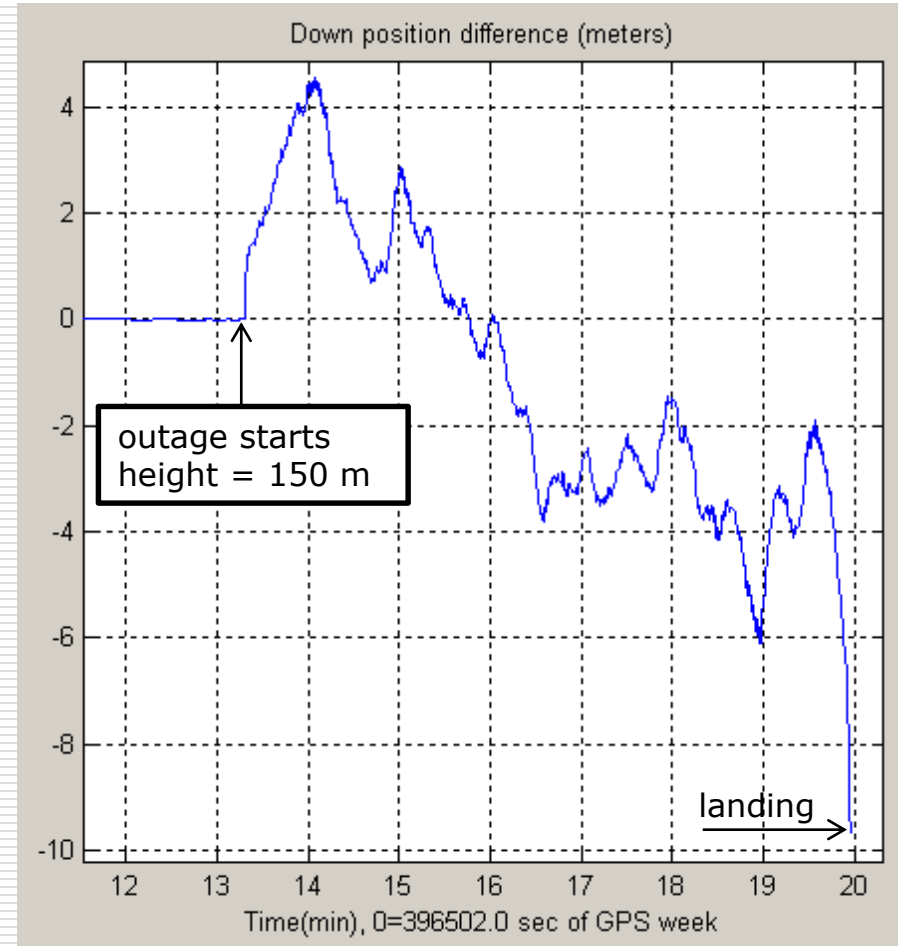


# Example – Impact in height with GPS outage (min)

## Free INS – GPS, SPP



## Baro-aided INS – GPS, SPP



# Processing requirements

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- Correct time-synchronization of data
  - IMU
  - Barometer
  - GNSS
  
- Estimation of scale-factor  $K$  and initial pressure  $P_0$  via GNSS measurements during flight
  
- System calibration with IMU/Barometer/GNSS
  - Noise characteristics
  - Position & attitude
  - Leverarms
  - Custom filter

# Conclusion

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- Increased Redundancy
  
- Better height uncertainty
  - Barometer better than GNSS-Code in height
  
- System calibration is needed
  - Deterministic
  - Stochastic
  
- Bridge certain length of GNSS gap
  - Return-home if problem is detected
  - Height accuracy better than single point positioning < 5 min of GNSS signal outage

# Questions

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# Reference

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- *Luiset, B., Clausen, P., Skaloud J., Développement d'un baromètre – altimètre à bord d'un drone, Projet de Master, 2016, EPFL, ESGT.*